

## CHAPTER 5

### MECHANICAL/ELECTRICAL INTERFACE

The interface between LV and SC consists of mechanical and electrical interfaces. Through mechanical interface, the SC is mated with the LV mechanically, while the electrical interface functions to electrically connect the LV with SC.

In this chapter, interfaces of LM-2E/ETS and LM-2E/EPKM are focused on.

#### **Part A: Mechanical/Electrical Interface Provided by LM-2E/ETS**

##### **A5.1 LM-2E/ETS Mechanical Interface**

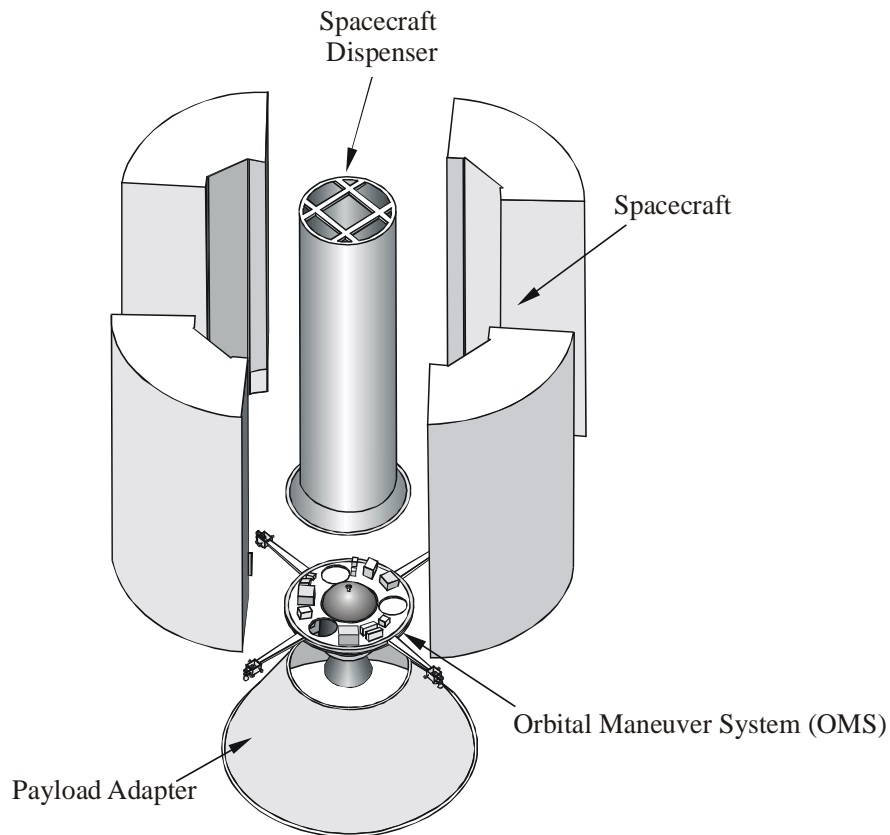
###### **A5.1.1 Summary**

LM-2E/ETS provides two types of mechanical interface: Type A and Type B. Type A mechanical interface is used for connecting SCs laterally, while Type B for connecting SCs from their bottom.

###### **A5.1.2 Type A Mechanical Interface**

The SCs are connected to the dispenser laterally, and the dispenser is bolted on the main structure of OMS that is connected with payload adapter by clampband. See **Figure A5-1**.

*Note:* 1) OMS stands for orbital maneuver system;  
2) ETS consists of OMS and Dispenser;



**Figure A5-1 Type A Mechanical Interface**

- **Type A SC Dispenser**

The SC dispenser functions to fasten and release the satellites. The typical type A SC dispenser is composed of a cylinder and a cone made from frame-skin semi-monocoque structure. The specific design is program dependent.

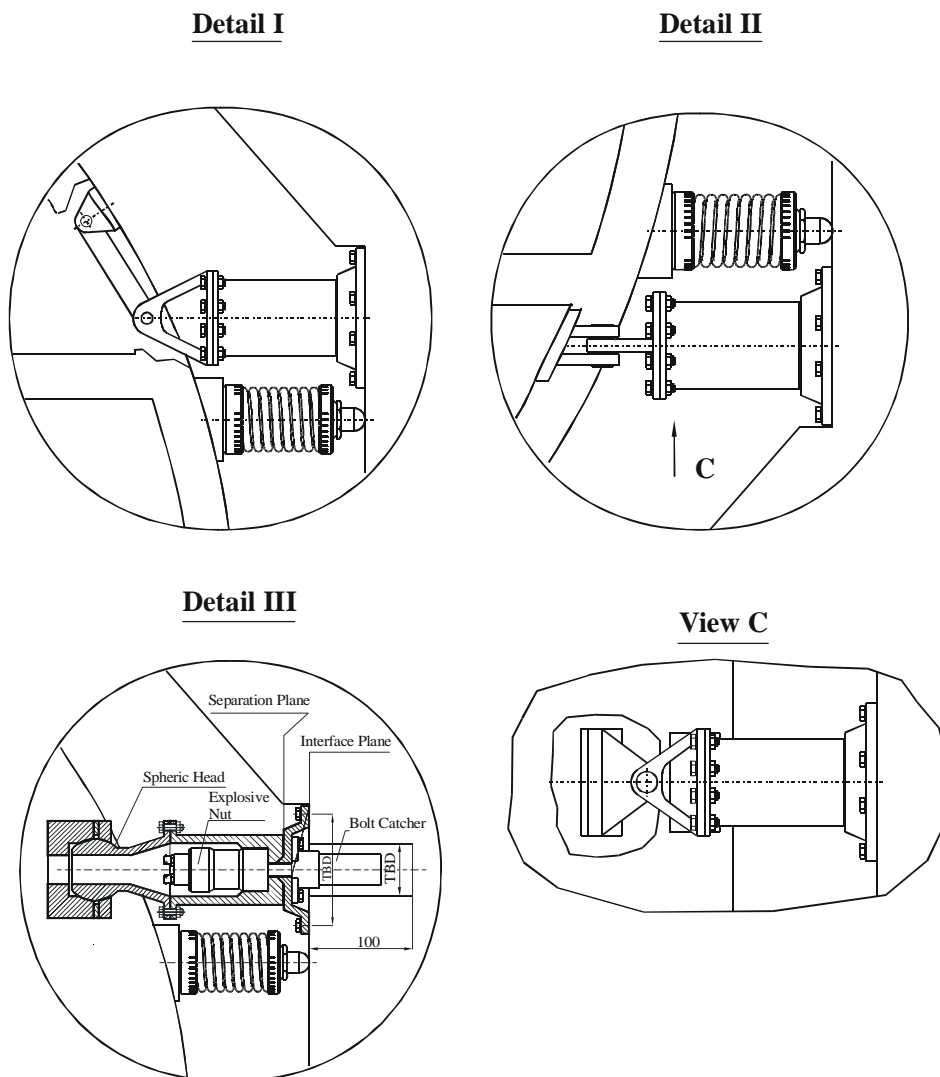
The dispenser is fixed on the main structure of OMS by bolts. The SCs are connected with the dispenser by low-shock explosive nuts and separation springs.

See **Figure A5-2a&b**.



- **Separation Device for Type A Dispenser**

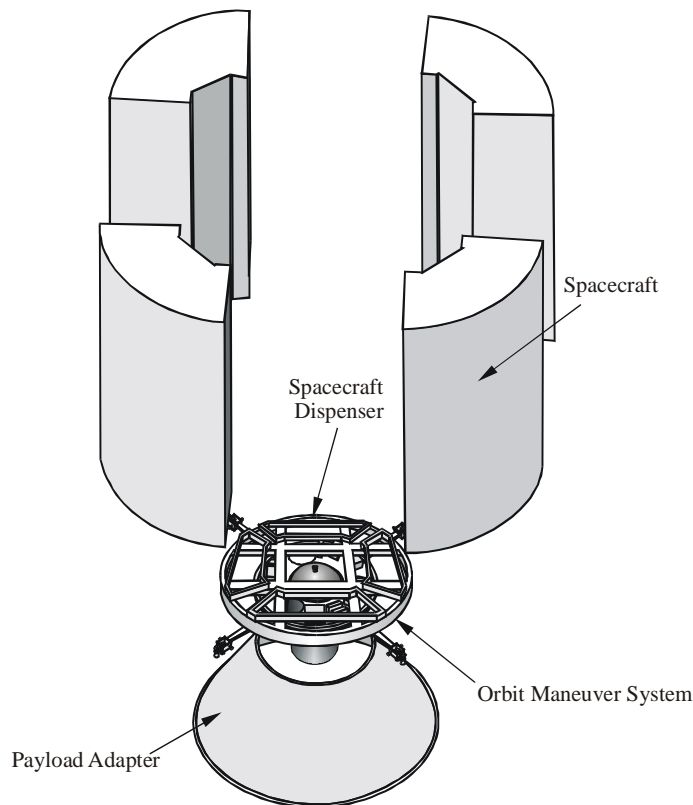
The SC/LV separation device consists of explosive nuts and separation springs as shown in **Figure A5-2b**. The explosive nuts are used for locking and unlocking the SCs. Catchers can collect the separated bolts. The separation springs includes springs, bracket, pushing rod, etc. The device can provide a SC/LV separation velocity according to user's requirements.



**Figure A5-2b Separation Device for Type A SC dispenser**

### A5.1.3 Type B Mechanical Interface

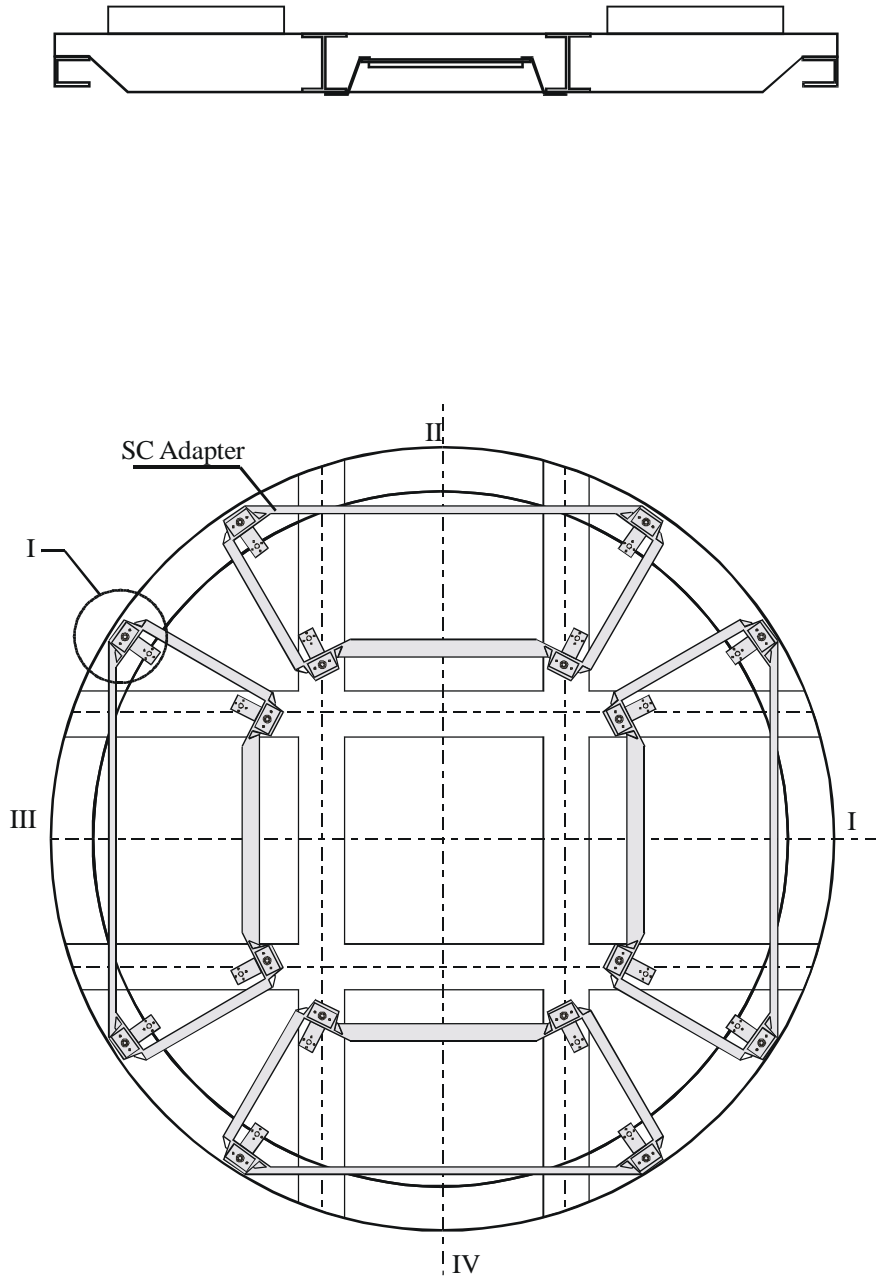
The SCs are connected to the dispenser from their bottom, and the dispenser is fixed on the main structure of OMS, which is connected with payload adapter by clampband. See **Figure A5-3**. There are 4 SC adapters fixed on the main structure of the typical type B dispenser. The SCs are mounted on the SC adapters by low-shock explosive nuts and separation springs.



**Figure A5-3 Configuration of Type B Mechanical Interface**

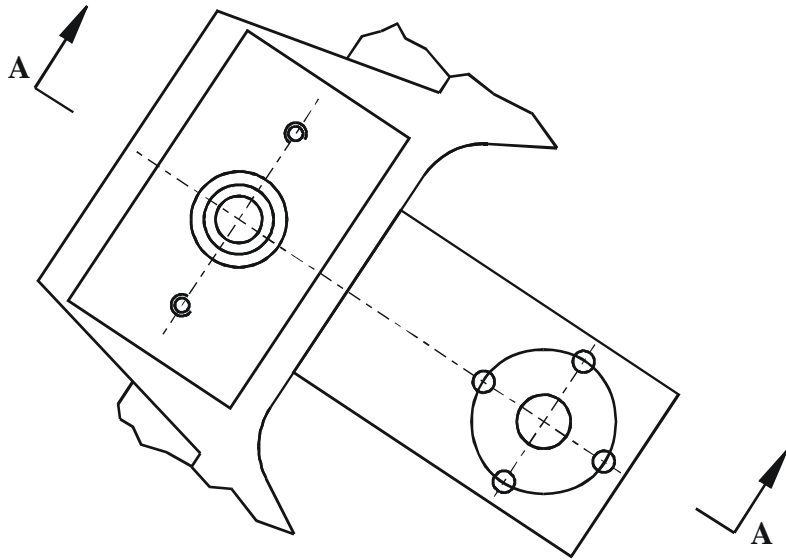
- **Type B Dispenser**

The type B dispenser is a short reverse cone structure with four SC adapters as shown in **Figure A5-4a**. The SC adapter is fastened to the SC adapter at its bottom using explosive nuts. All the separation system except for bolt catcher is attached on the SC adapter.

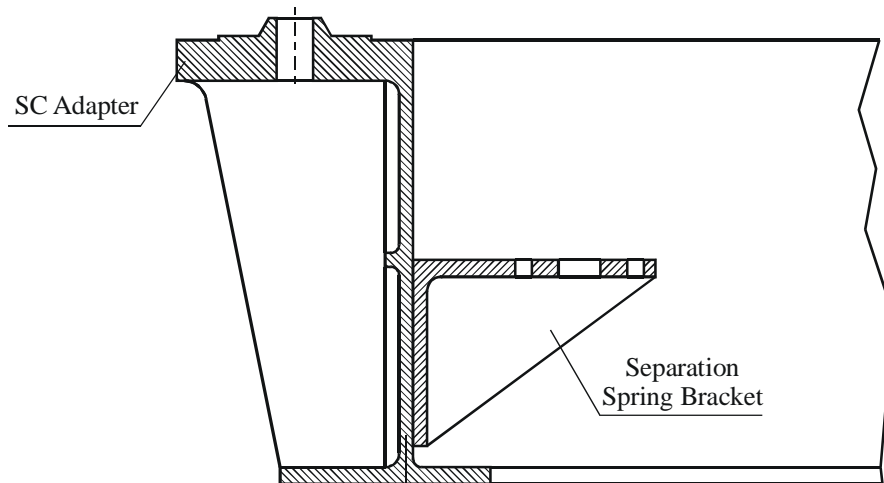


**Figure A5-4a Type B Dispenser (1)**

Detail I



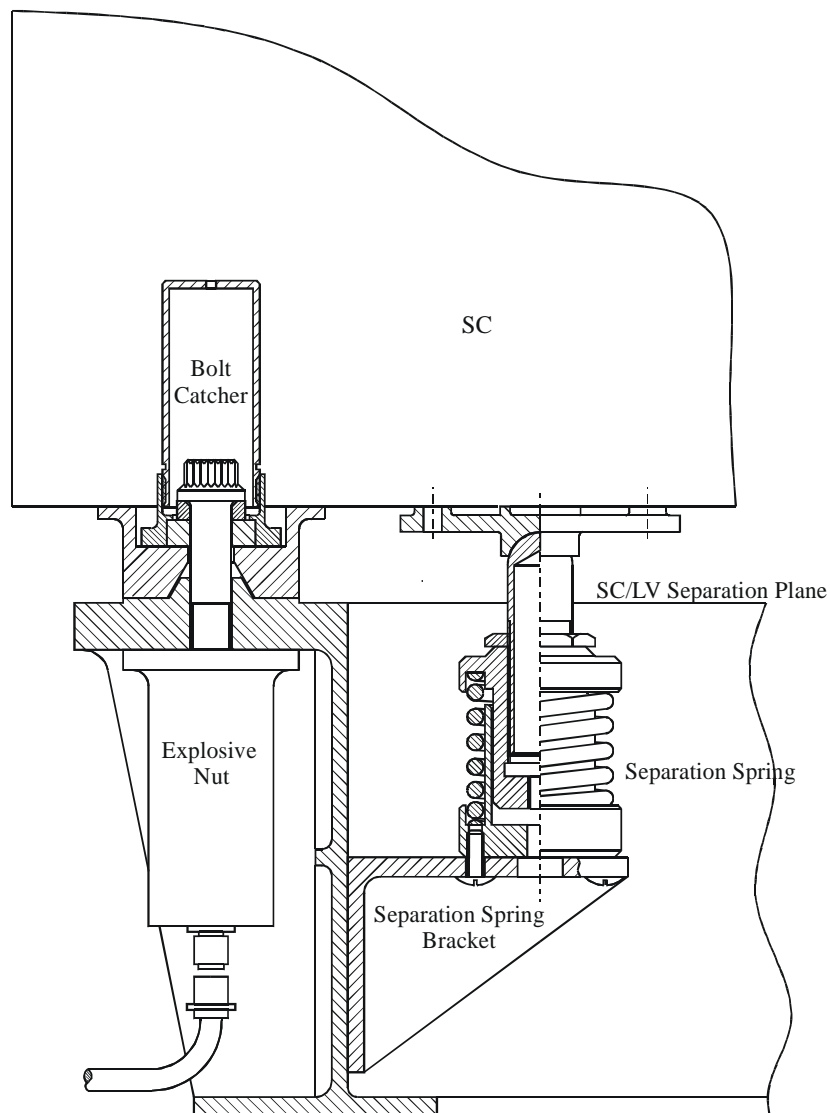
Section A-A



**Figure A5-4b Type B Dispenser (2)**

- **Separation System**

The separation system consists of explosive nuts and separation springs. See **Figure A5-4c**. The explosive nuts are used for locking and unlocking the SCs. The separation springs includes springs, bracket, pushing rod, etc. The catcher can collect the separated bolts. The separation system can provide a SC/LV separation velocity according to user's requirements.



**Figure 5-4c Separation System for Type B Dispenser**



## A5.2 LM-2E/ETS Electrical Interface

The SC is electrically connected with SC's electrical ground support equipment (EGSE) through In-flight-disconnectors (IFDs) and umbilical cables provided by LV side. By using of EGSE and the umbilical cables, SC team can perform wired testing and pre-launch control to the SC, such as SC power-supply, on-board battery charging, wired-monitoring on powering status and other parameters.

### A5.2.1 In-Flight-Disconnectors (IFDs)

- **Quantity**

Typically, there are two IFDs mounted on the dispenser for each SC. The detailed location will be coordinated between SC and LV sides and finally defined in ICD.

- **IFD Supply**

Generally, the IFDs are selected and provided by the user. CALT is responsible to solder IFDs to the umbilical cables. The necessary operation and measurement description shall also be provided. (If the user selects the China-made connectors, CALT will provide the halves installed on the SC side.). The available China-made connectors are YF8-64 (64 pins), FD- 20(20 pins), FD-26(26 pins), FD-50(50 pins), etc.

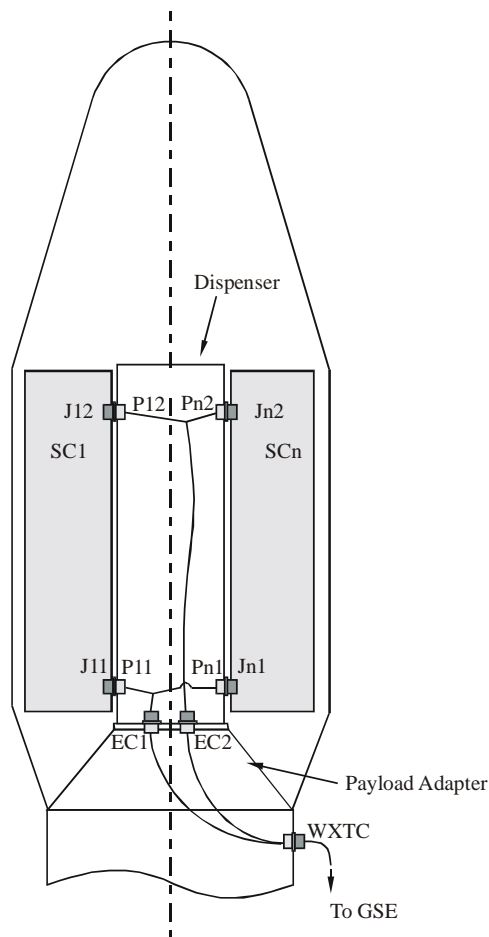
- **Separation signal through IFDs**

There are four break-wires on the two IFDs for each SC, which generate SC/LV separation signals. The SC will receive the SC/LV separation signals once the break-wires circuitry break while SC/LV separates.

In the same way, there are four break-wires on the IFDs. The IFDs will send the SC/Dispenser separation signal to LV once the break-wires circuitry break while SC/Dispenser separates. This separation signal will be sent to LV's telemetry system through EY1 interface. The break-wire's allowable current:  $\leq 100\text{mA}$ , allowable voltage:  $\leq 30\text{V}$ .

### A5.2.2 Umbilical System

The umbilical system consists of onboard-LV Parts and ground parts. Refer to **Figure A5-5** and **Figure A5-6**. The cable from Launch Control Console (LCC) to Umbilical Tower, EB26/EB36, BOX3, BOX4, and Power-supplies 1&2 are the common for different missions. The onboard-LV cable, as well as ground cable from WXTC to ED 13,14&15 and BOX1 & BOX2, will be designed for dedicated SC according to user's needs. In order to assure the quality of the product, the umbilical system will be provided to the user after pre-delivery acceptance test and insulation/conductivity checkouts in the launch site.



**Figure A5-5 LM-2E/ETS Onboard-LV Electrical Interface**

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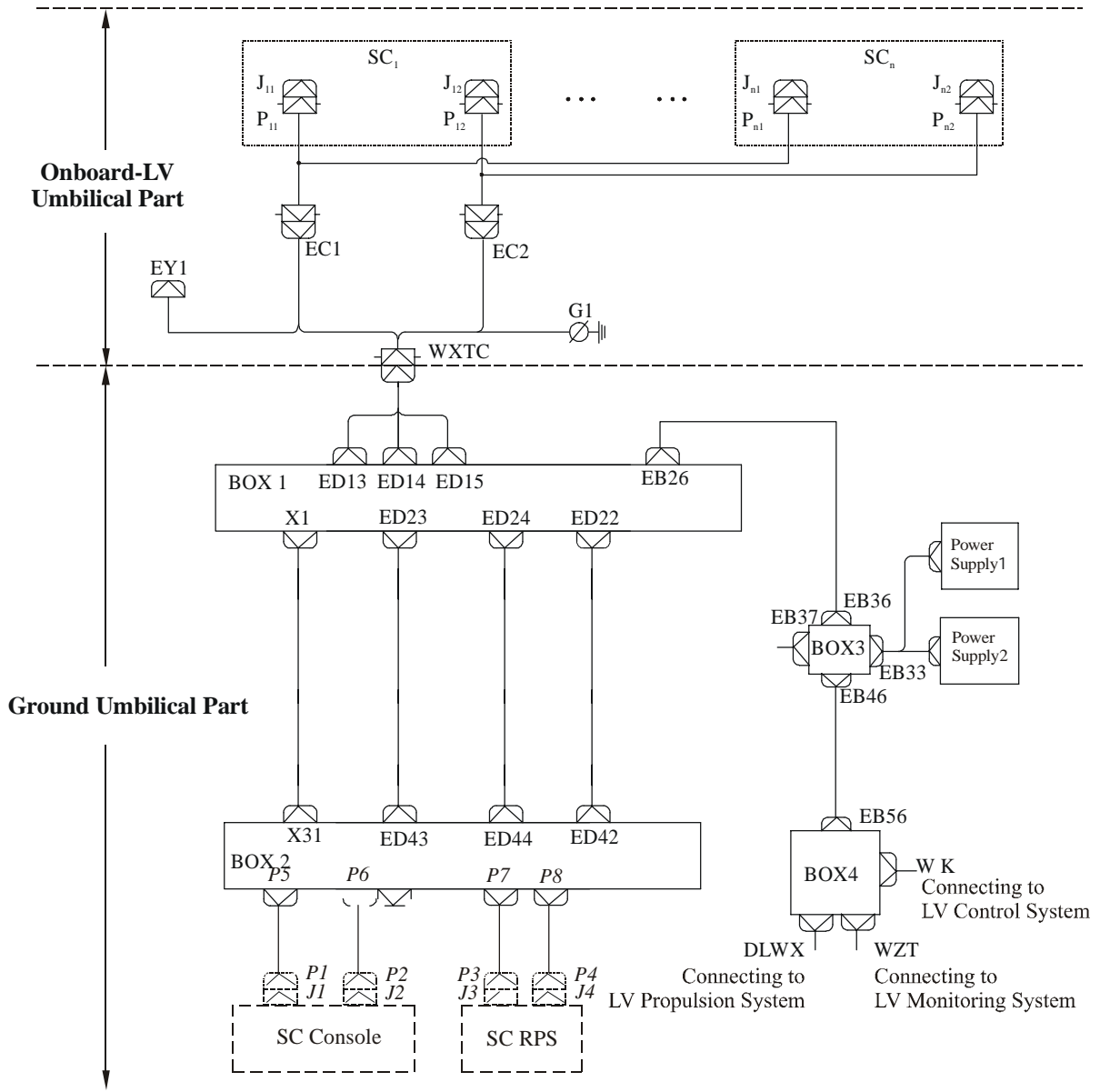


Figure A5-6 Onboard-LV and Ground Umbilical System

### A5.2.2.1 Onboard-LV Umbilical Cable

- **Composition**

The Onboard-LV cable net comprises the cables from the IFDs to WXTC. These umbilical cables will fly with LV.

Whereas:

Code	Description
$P_{11}, P_{12} \dots P_{n1}, P_{n2}$	LV/SC electrical connectors at LV side which is crimp-connected to the cables.
EC1, EC2	Technological interfaces between Dispenser and OMS.
EY1	Interface between umbilical cable and LV TM system, through which the LV/SC separation signal is sent to LV TM system
WXTC	Umbilical cable connector (LV-Ground)
G1	Grounding points to overlap the shielding of wires and the shell of LV

Refer to **Figure A5-6**.

- **Umbilical Cable Design**

SC side shall specify characteristics of the IFDs. The specific contents are pin assignment, usage, maximum voltage, maximum current, one-way maximum resistance etc. CALT will design the umbilical cable according to the specified requirements.

- **Types and Performance of Umbilical Cable**

Generally, ASTVR and ASTVRP wires are adopted for the onboard-LV cable net:  
 ASTVR,  $0.5\text{mm}^2$ , fiber-sheath, PVC insulation;  
 ASTVRP,  $0.5\text{mm}^2$ , fiber-sheath, PVC insulation, shielded.

For both cables, their working voltage is  $\leq 500\text{V}$  and DC resistance is  $38.0\Omega/\text{km}$  ( $20^\circ\text{C}$ ). The single core or cluster is shielded and sheathed.

### A5.2.2.2 Ground Umbilical System

- **Composition**

The ground umbilical cable net consists of umbilical cable connector (WXTC), cables, box adapters, etc. Refer to **Figure A5-6**.

Whereas:

Code	Description
WXTC	<p>WXTC is umbilical cable connector (LV-Ground) whose female half (socket) is installed at the wall of the VEB, while the male half (pin) is attached to the top end of ground cable. The disconnection of WXTC is electrically controlled. (The disconnection is powered by BOX 3 and controlled by BOX 4. In the mean time, forced disconnection is also used as a spare separation method.) Generally, WXTC disconnects at about 8min prior to launch. If the launch was terminated after the disconnection, WXTC could be reconnected within 30min.</p> <p>The SC should switch over to internal power supply and cut off ground power supply at 5 minutes prior to WXTC disconnection. Therefore, during disconnection only a low current monitoring signal (such as 30V, <math>\leq 100\text{mA}</math>) is permitted to pass through the WXTC.</p>
BOX1	<p>BOX 1 is a box adapter for umbilical cable that is located inside the Payload Cable Measurement Room on the umbilical tower. Refer to <b>Chapter 7</b>. (If needed, BOX 1 can provide more interfaces for the connection with SC ground equipment.)</p>
BOX2	<p>BOX 2 is another box adapter for umbilical cable that is located inside the Underground Power-supply Room. Refer to <b>Chapter 7</b>. Other SC ground support equipment (RPS, Console, etc.) are also located inside the Underground Power-supply Room.</p>
BOX3	<p>This is a relay box for the disconnection of the umbilical cable. BOX 3 is located inside the Underground Power-supply Room. Box 3 is powered by 2 DC regulated power supply sets. These two power supply sets are in “working-state” sparing to each other.</p>
BOX4	<p>BOX 4 is located inside the Payload Cable Measurement Room . It is for the control of the pre-launch disconnection of SC umbilical cables.</p>

### ● Interface on Ground

Generally, there are four interfaces on ground, namely, two for SC Console (P1/J1&P2/J2), and the other two for SC power supply (P3/J3&P4/J4). SC side will define the detailed requirement of ground interfaces. Those connectors (P1,P2,P3,P4) to be connected with SC ground equipment should be provided by SC side to LV side for the manufacture of cables.

Location	Code	Specification	Quantity
LV side interfaces	P1	To be defined by SC side	2
	P2		2
	P3		2
	P4		2

If LV side couldn't get the connectors from SC side, this ground interface cable will be provided in cores with pin marks. SC side can also provide this ground cable. The length of this cable is about 5 meters. If so, LV side will provide the connectors to connect with BOX 2.

### ● Type & Performance of Umbilical Cable

#### ✧ Single-Core Shielded Cable

Woven wire net for shielding of cable; Working voltage:  $\leq 60\text{V}$ ; DC resistance ( $20^\circ\text{C}$ ) of each core:  $38.0\Omega/\text{km}$ . Function: common control and signal indicating.

#### ✧ Ordinary Insulation Cable

No shielding for each core, woven tin-plated copper wire for shielding of cable; Working voltage:  $\leq 110\text{V}$ ; DC resistance ( $20^\circ\text{C}$ ) of each core:  $28.0\Omega/\text{km}$ . Function: SC's power supply and battery charging.

#### ✧ Twin-twist Shielded Cable

Each twisted pair is shielded and the whole cable has a woven wire net for shielding. Impedance:  $100\Omega$ . Function: SC data transmission and communication.

Under normal condition, the umbilical cable (both onboard-LV and ground) has a insulation resistance of  $\geq 10\text{M}\Omega$  (including between cores, core and shielding, core and LV shell)

The umbilical system can be dedicatedly designed according to the user's requirements.

### **A5.2.3 Umbilical Cable Disconnect Control**

LV side is responsible for the pre-launch disconnection of umbilical cable through BOX3 and BOX 4.

Inside the underground Power Supply Room, there are two DC regulated power supply which will provide power for the cables. They are all in working condition sparing to each other.

Generally, according to the count-down launch procedure, only after LV side has received the confirmation that SC has turned to internal power and SC is normal, could the order of umbilical cable disconnection be sent out.

### **A5.2.4 Anti-lightning, Shielding and Grounding**

In order to assure the safety of the operations of both LV and SC, some measures have been taken for anti-lightning, shielding and grounding.

- (1) The cable has two shielding layers, the outer shielding is for anti-lightning while the inner shielding is for anti-interference.
- (2) For the cables from WXTC to BOX 2, the outer shielding (anti-lightning) has a grounding point every 20m. These grounding measures can assure the lightning and other inductance to be discharged immediately. The grounding locations are either on the swing rods or the cable's supporting brackets.
- (3) The inner shield has a single grounding. The inner shields of the on-board cables are connected to BOX 2 through WXTC. BOX 2 has a grounding pole.
- (4) The inner and outer shields are insulated with each other inside the cables.

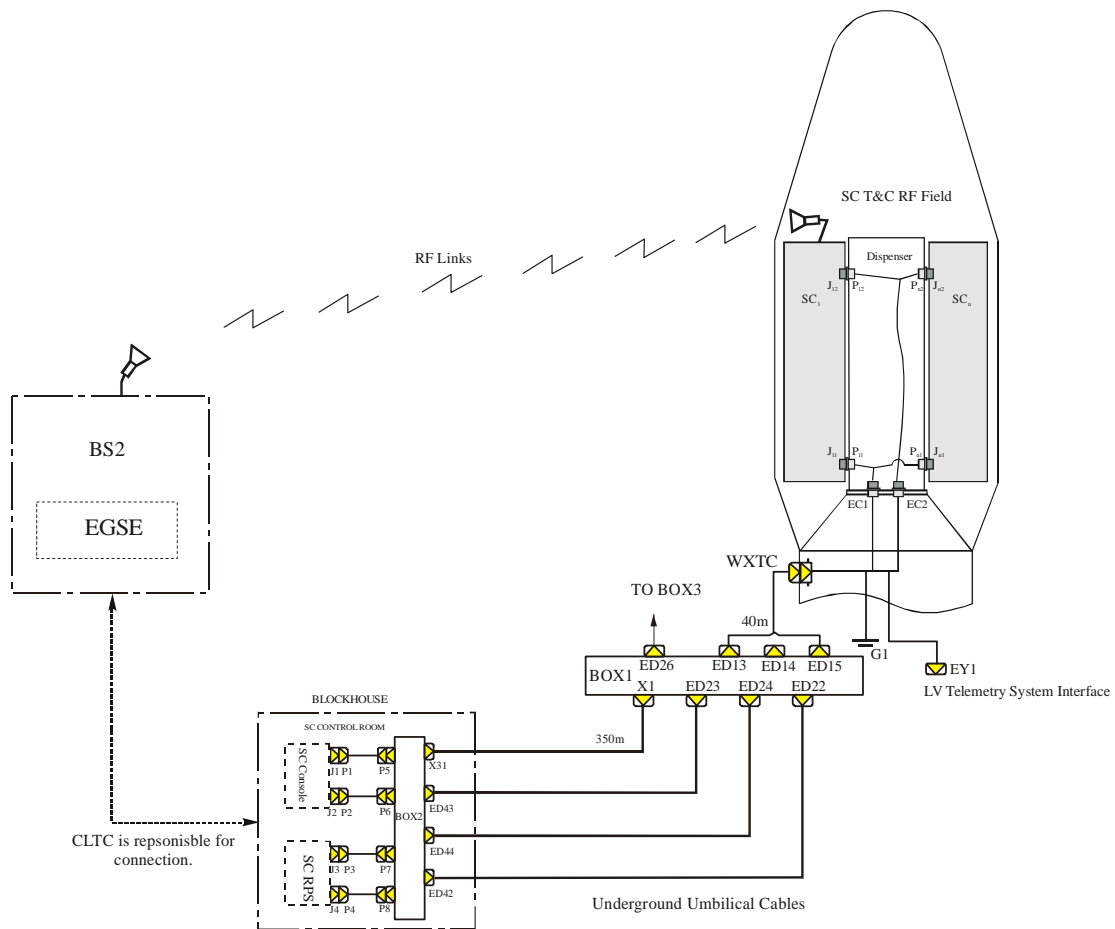
<b>SPECIAL STATEMENT</b>
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Any signal possibly dangerous to the flight can not be sent to the SCs during the whole flight till LV/SC separation. Only LV/SC separation can be used as the initial reference for all SC operations. After LV/SC separation, SC side can control SC through microswitches and remote commands.

### A5.3 RF Links

#### A5.3.1 RF Path

JSLC can provide RF link from technical center to the umbilical tower. Refer to **Figure A5-7**.



**Figure A5-7 RF Links in JSLC**



### A5.3.2 Characteristics of RF Link

#### (1) Frequency

C Band:           Up-link: 5925~6425 MHz  
                       Down-link: 3700~4200 MHz

Ku Band:           TBD

#### (2) Signal Level

C Band: See following table

Ku Band: TBD

Frequency	SC Antenna		EGSE	
	EIRP	PFD	Input	Output
Telemetry	37dBm		-70dBm	
Command		-85dBW/m <sup>2</sup>		30dBm

### SPECIAL STATEMENT

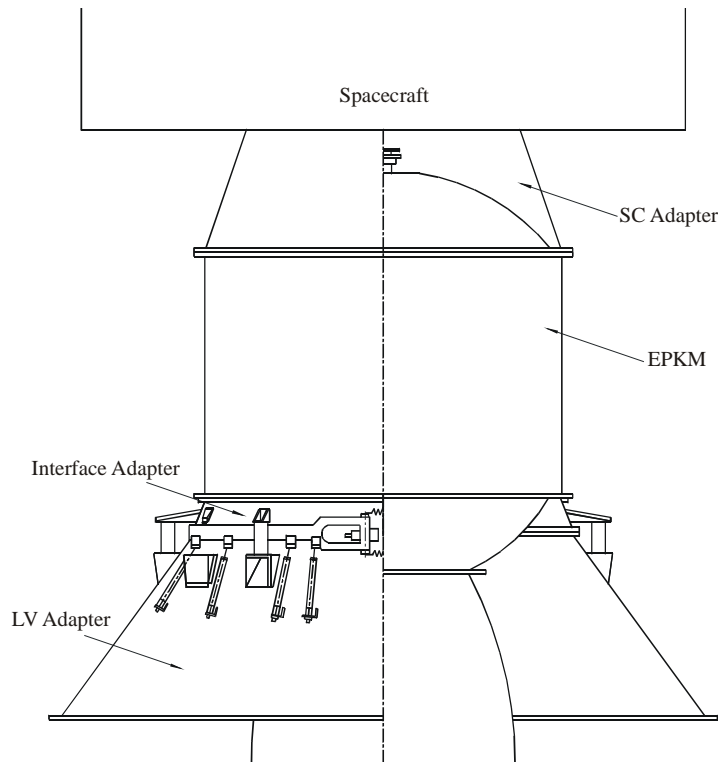
A mission dedicated RF working plan will be worked out. Anyway, the SC RF equipment should be turned off during the whole flight phase of LV until all SCs are separated from the LV hardware.

## Part B: Mechanical/Electrical Interface Provided by LM-2E/EPKM

### B5.1 LM-2E/EPKM Mechanical Interface

#### B5.1.1 Summary

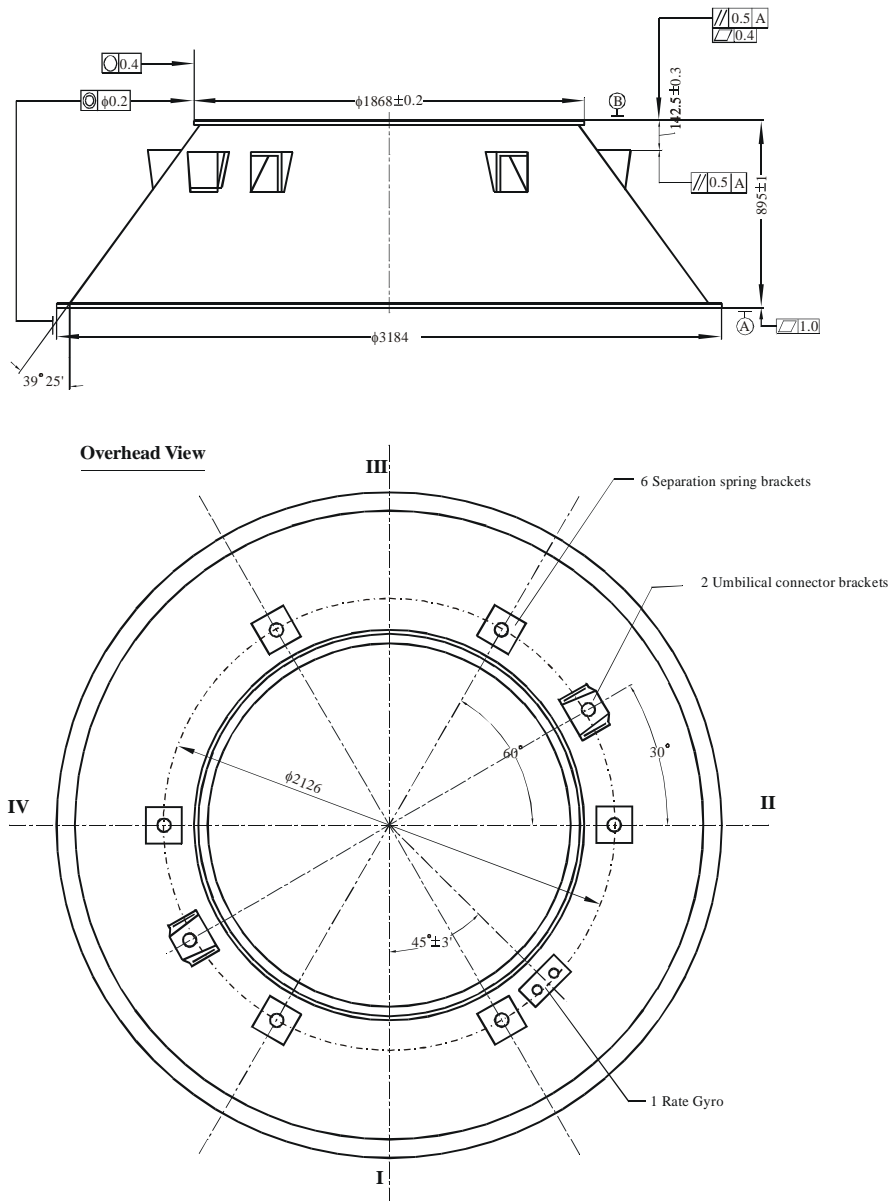
As shown in **Figure B5-1**, the SC adapter is connected with the SC on the top, and bolted with EPKM on the bottom. EPKM is bolted with the interface adapter, which is connected with LV adapter by clampband. When the clampband is released, the EPKM/SC stack, together with interface adapter, separates from LV adapter. The SC adapter connects with EPKM by 100 bolts. In general, SC will control the EPKM flight as well as EPKM/SC separation. CALT is willing to satisfy other requirements.



**Figure B5-1 LM-2E/EPKM Mechanical Interface Overview**

**B5.1.2 LV Adapter**

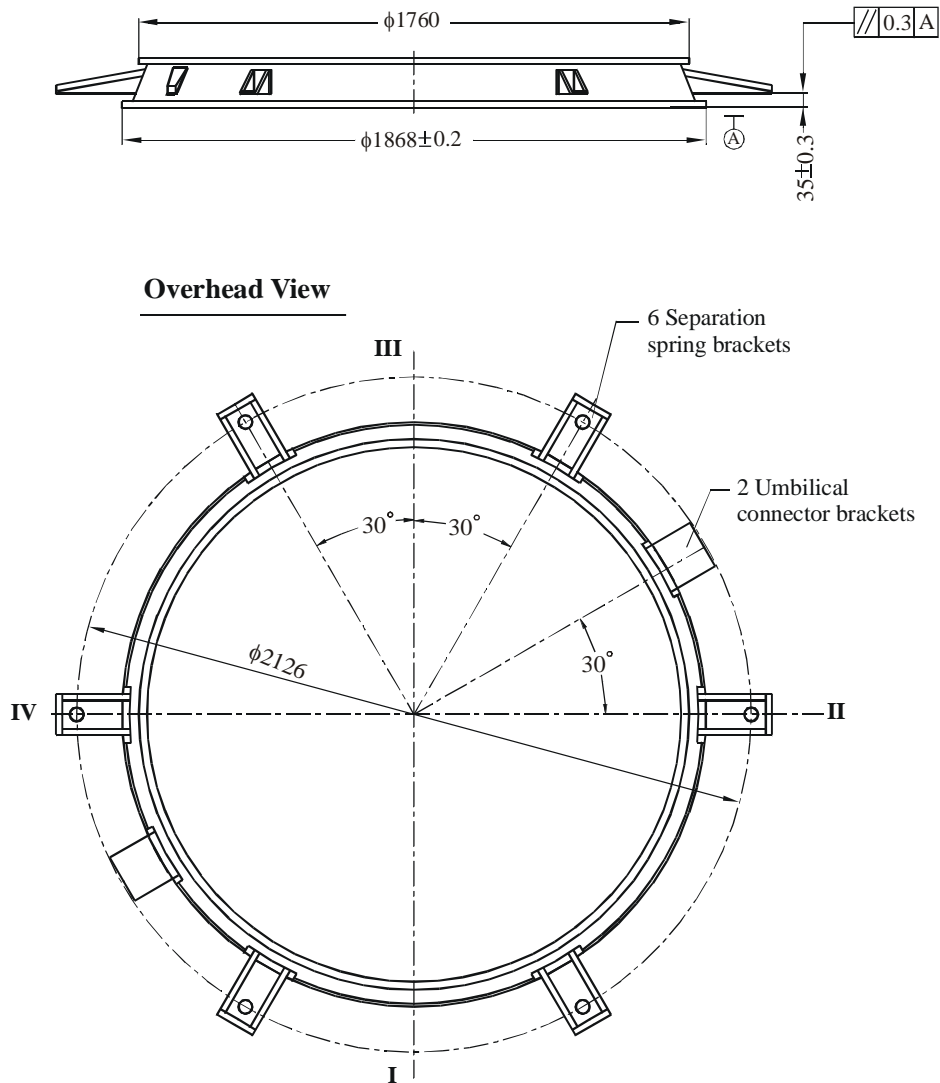
LV adapter is an 895mm-high truncated cone, whose top ring diameter is 1868mm and bottom ring diameter is 3184mm. Refer to **Figure B5-2**.



**Figure B5-2 LV Adapter**

### B5.1.3 Interface Adapter

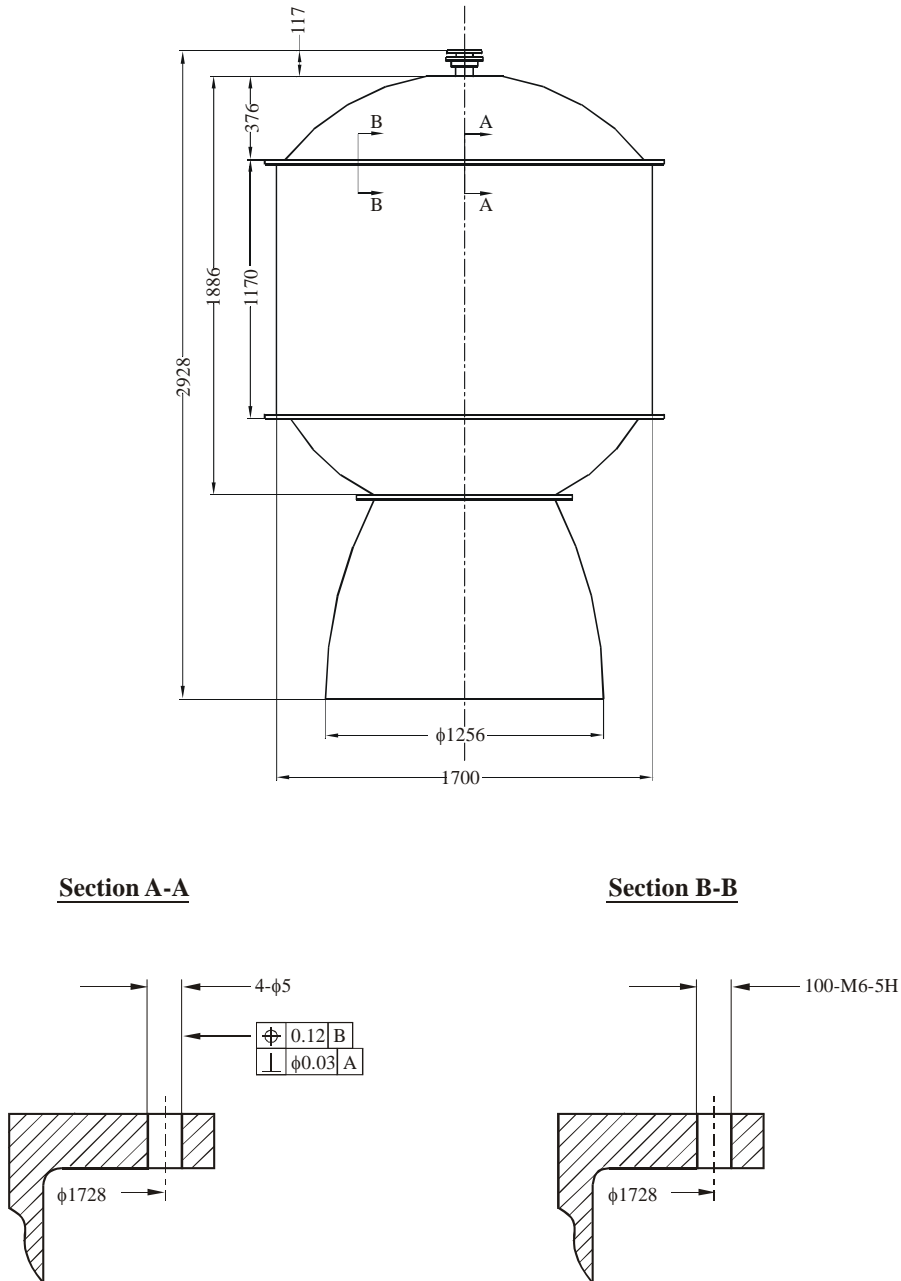
LV adapter is a truncated cone, whose top ring diameter is 1868mm and bottom ring diameter is 3184mm. Refer to **Figure B5-3**.



**Figure B5-3 Interface Adapter**

### B5.1.4 EPKM/SC Interface

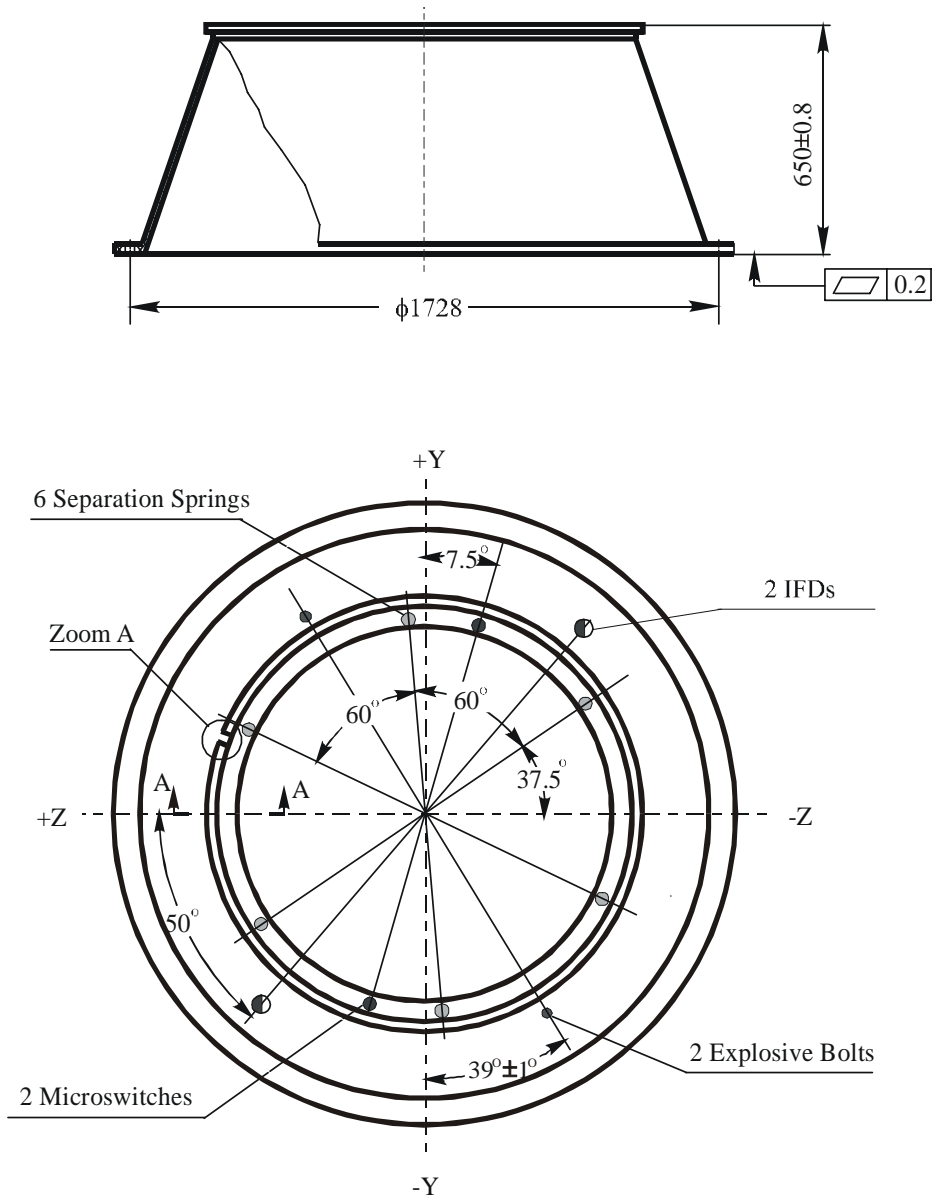
The top ring of EPKM is connected with SC adapter with 100 bolts as shown in **Figure B5-4**.



**Figure B5-4 EPKM/SC Interface**

**B5.1.5 SC Adapter**

CALT can provide 1194 adapter as shown in **Figure B5-5a&b.**



**Figure B5-5a SC Adapter (1)**

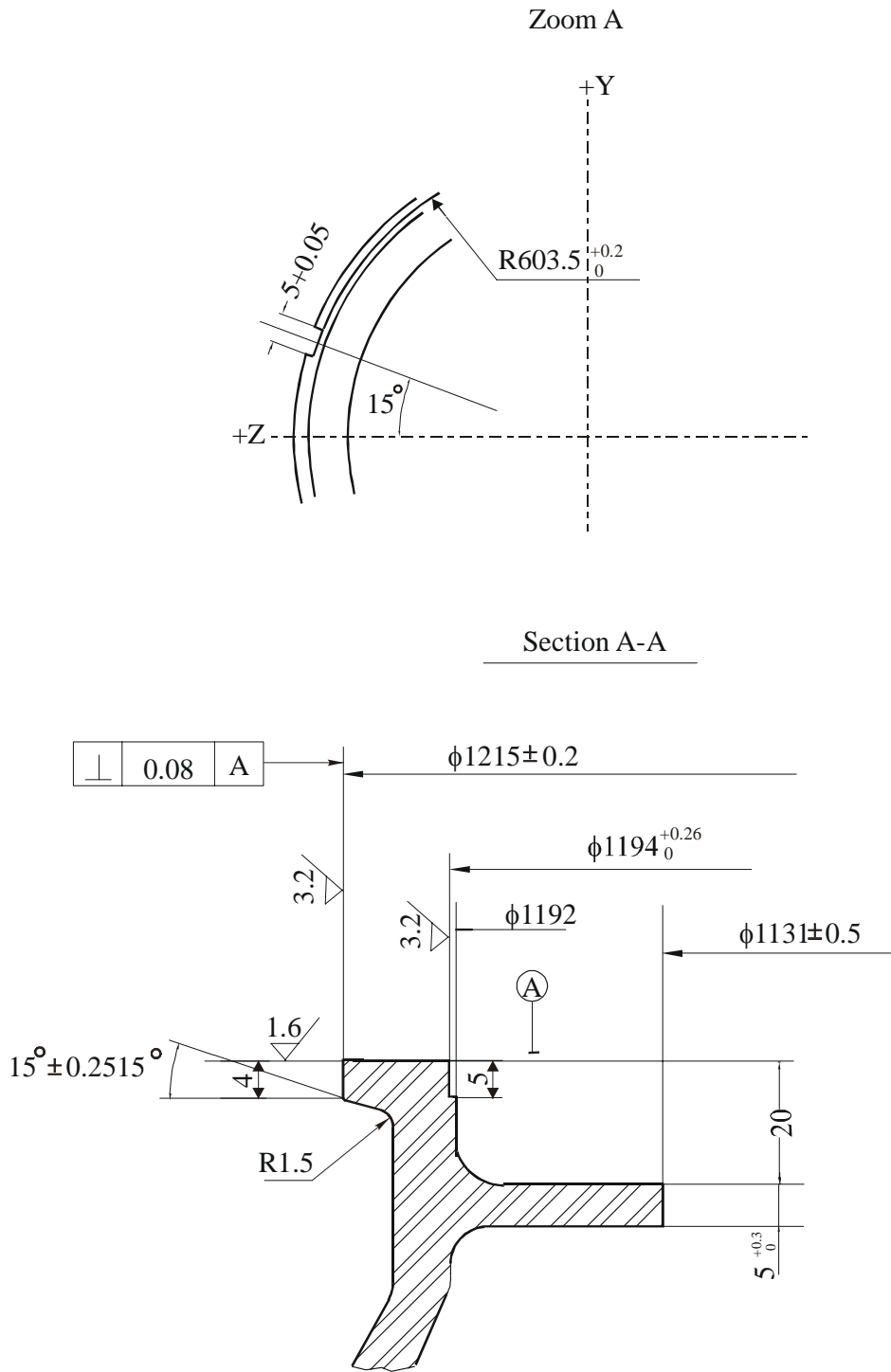
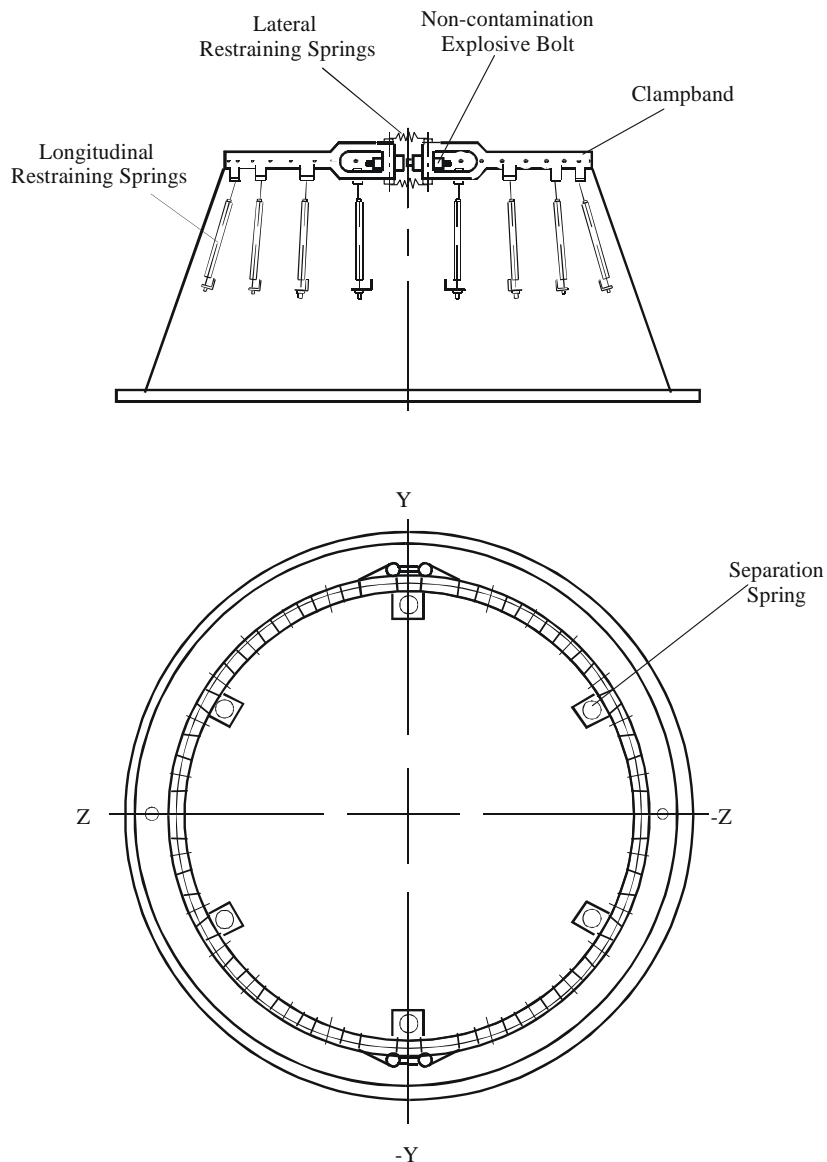


Figure B5-5b SC Adapter (2)

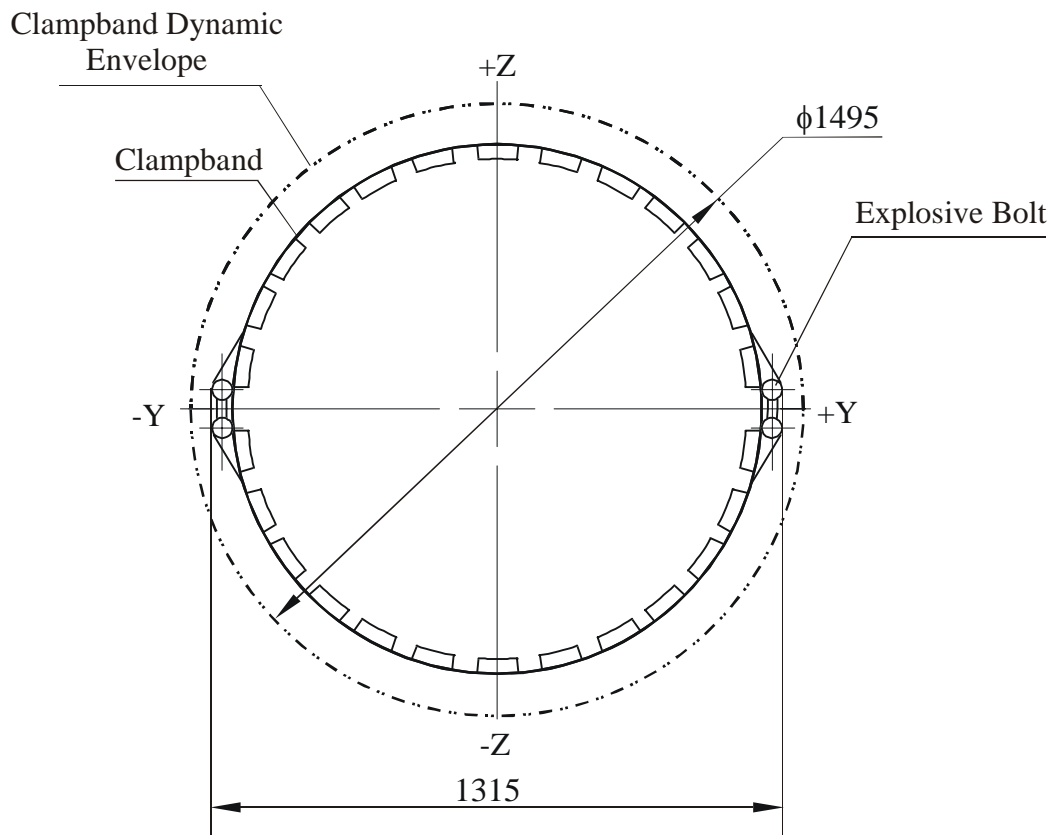
### B5.1.6 SC/EPKM Separation System

CALT can provide SC/EPKM separation system. The SC/EPKM separation system consists of clampband system and separation springs. The clampband system is used for locking and unlocking SC adapter and the SC. The separation springs are mounted on the SC adapter, which provides relative velocity between the SC and EPKM. Refer to **Figure B5-6a,b,c,d&e**.

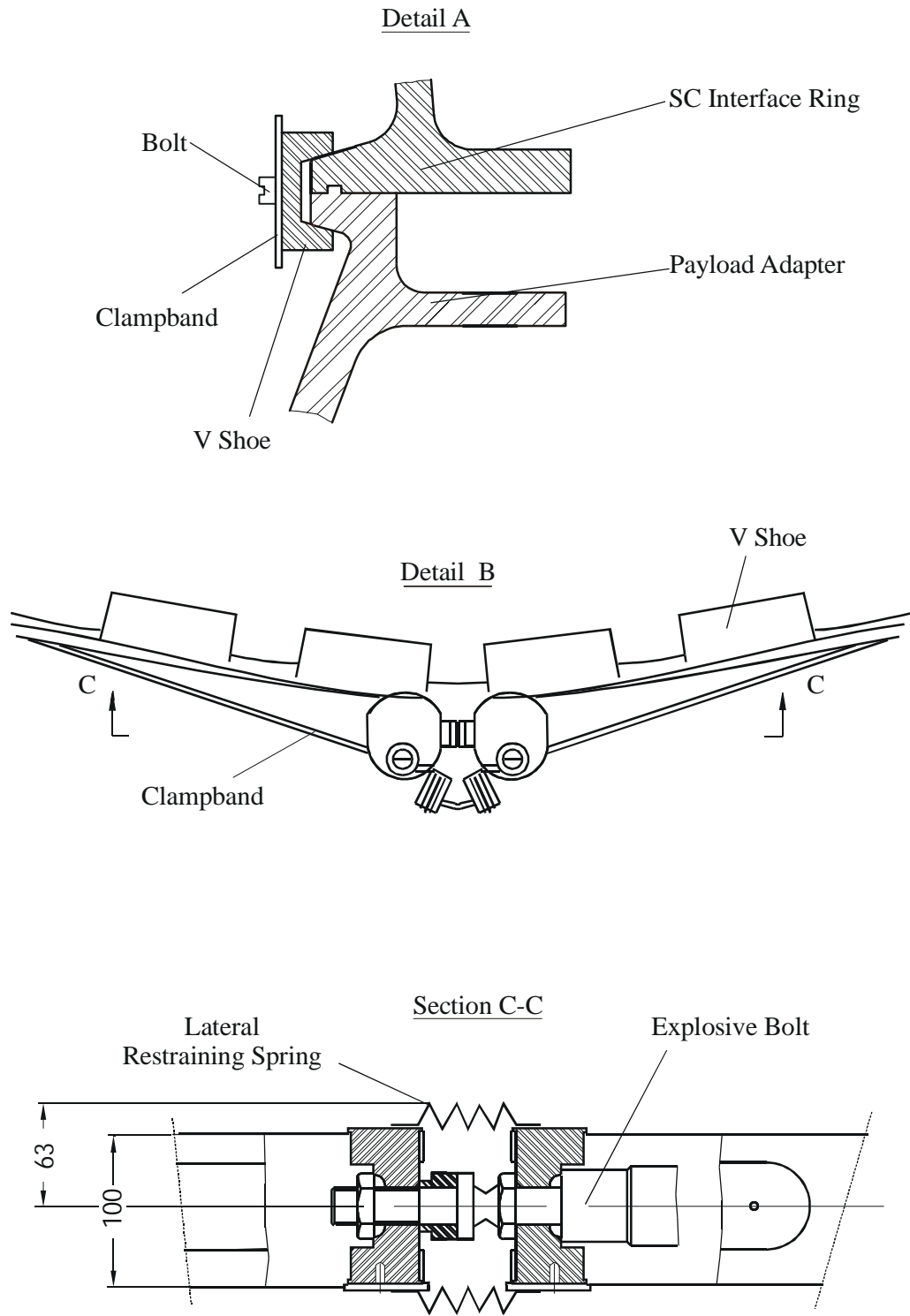


**Figure B5-6a SC/EPKM Separation System**

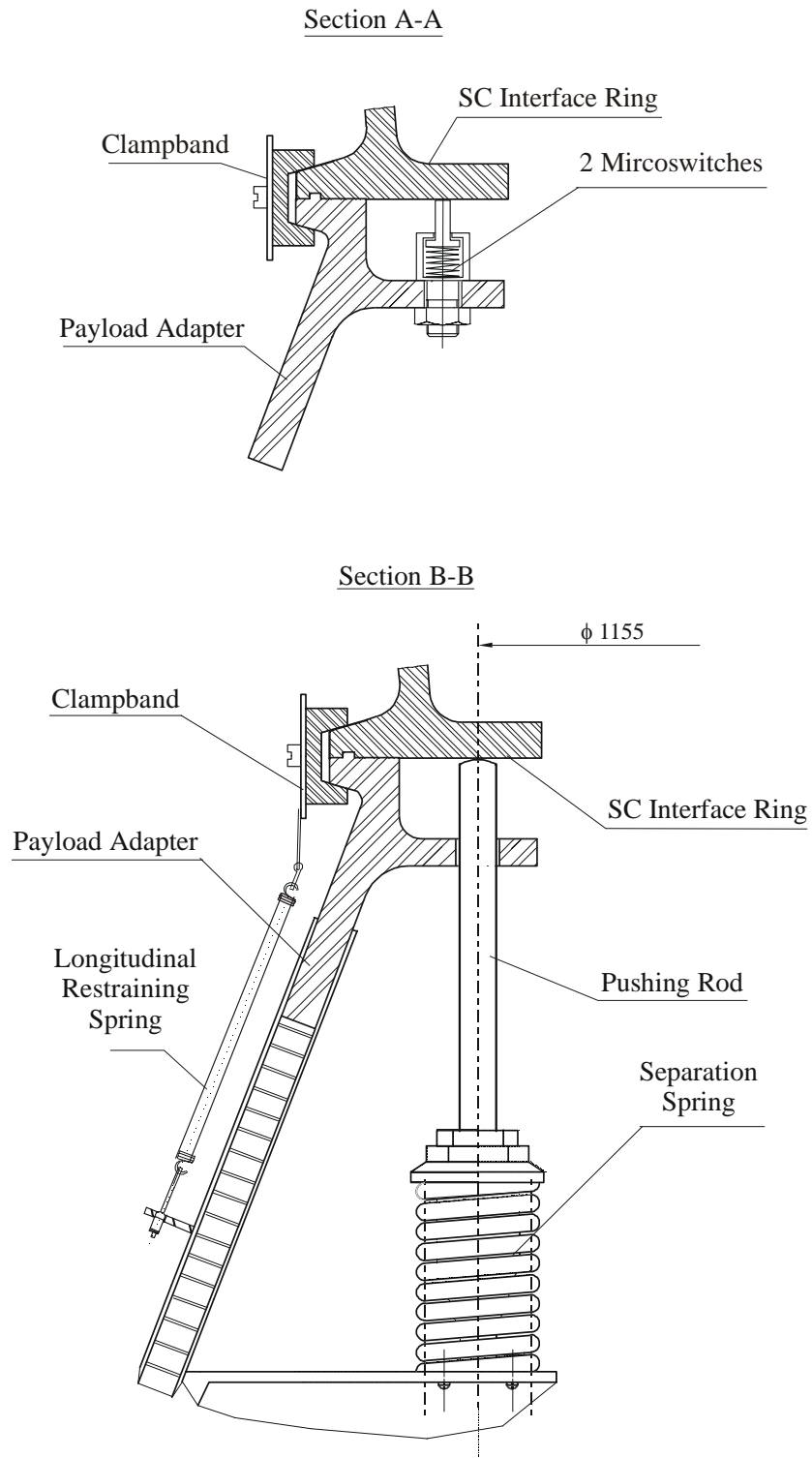




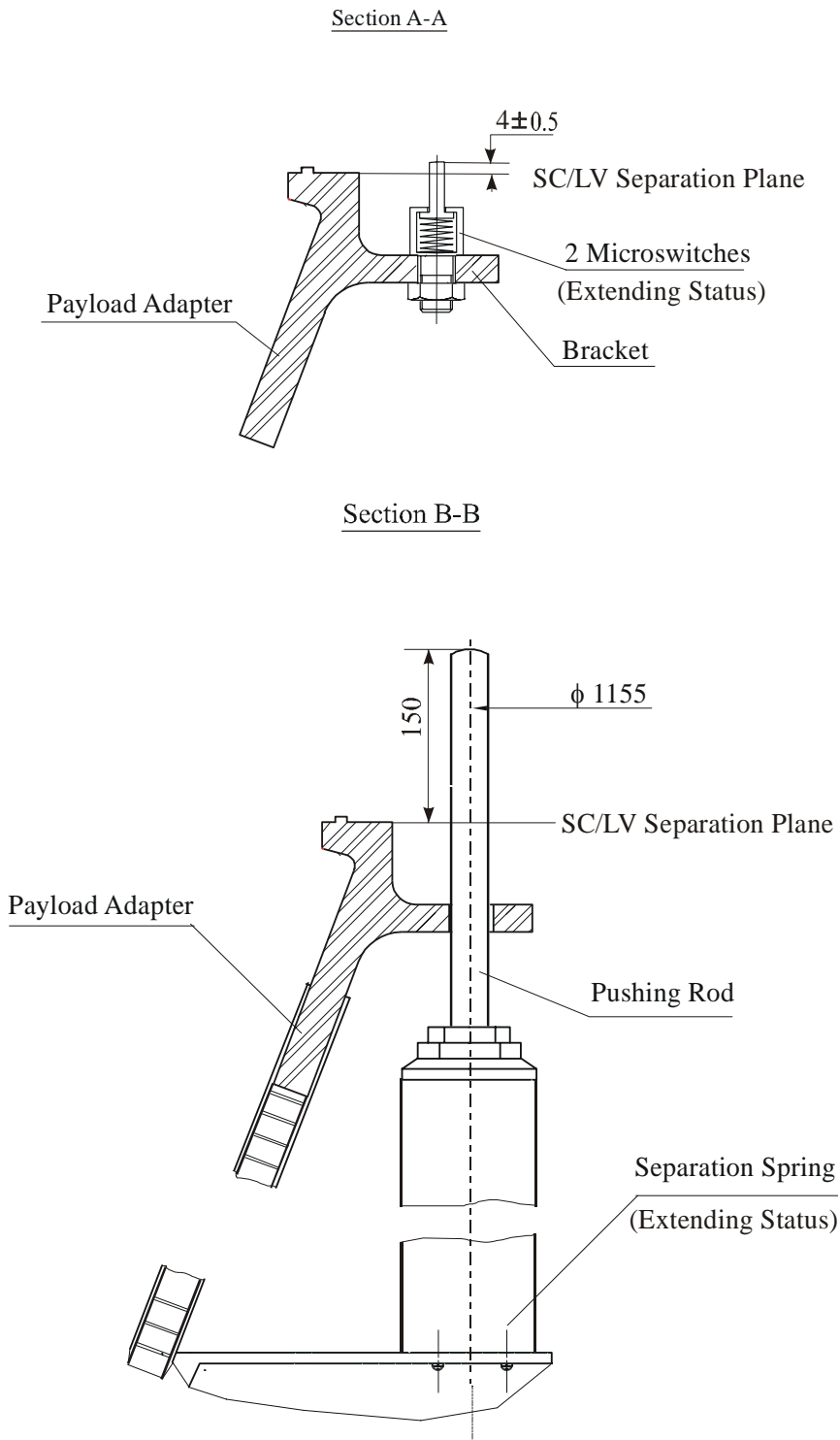
**Figure B5-6b Clampband Dynamic Envelope**



**Figure B5-6c Clampband in Detail**



**Figure B5-6d SC/EPKM Separation Spring**



**Figure B5-6e SC/EPKM Separation Spring (Extending Status)**

## B5.2 LM-2E/EPKM Electrical Interface

The SC is electrically connected with SC's electrical ground support equipment (EGSE) through SC/LV In-flight-disconnectors (IFDs) and umbilical cables provided by LV side. In general, SC will control SC/EPKM separation, the SC adapter will separate from SC together with EPKM. The actual electrical interface between LV and SC is defined at the interface between LV and EPKM, i.e. the top of the interface ring. By using of EGSE and the umbilical cables, SC team can perform wired testing and pre-launch control to the SC, such as SC power-supply, on-board battery charging, wired-monitoring on powering status and other parameters.

### B5.2.1 In-Flight-Disconnectors (IFDs)

- **Quantity**

Typically, there are two IFDs symmetrically mounted outside the top ring of the interface adapter. The detailed location will be coordinated between SC and LV sides and finally defined in ICD.

- **IFD Supply**

Generally, the IFDs are selected and provided by the user. CALT is responsible to solder IFDs to the umbilical cables. The necessary operation and measurement description shall also be provided. (If the user selects the China-made connectors, CALT will provide the halves installed at the SC side.)The available China-made connectors are YF8-64 (64 pins), FD- 20(20 pins), FD-26(26 pins), FD-50(50 pins), etc.

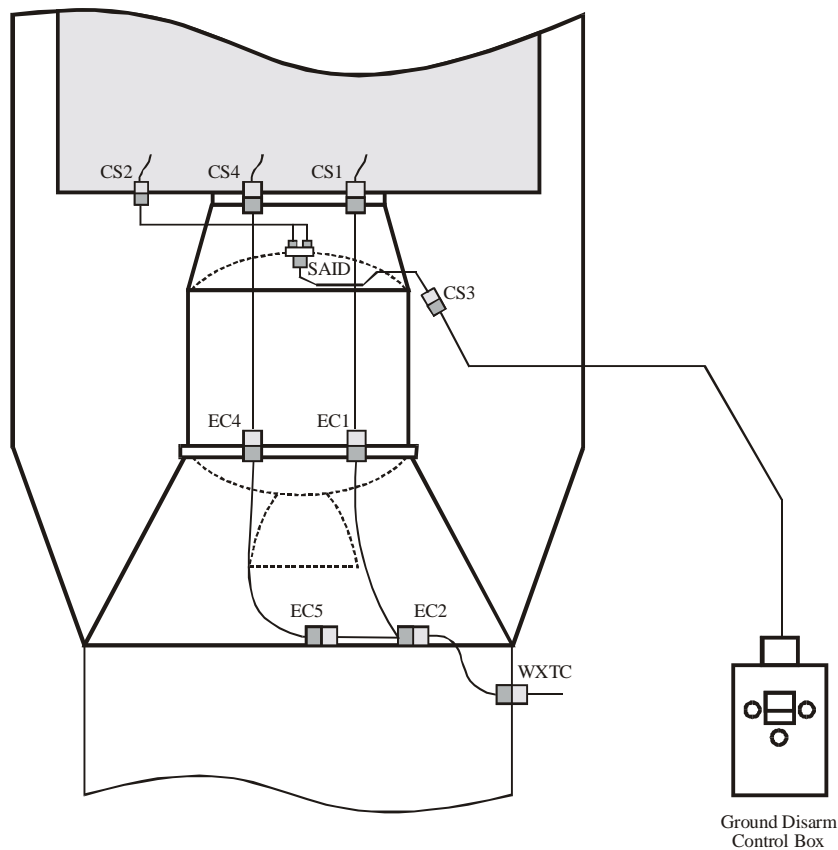
- **Separation Signal through IFDs**

There are four break-wires on the two IFDs for each SC, which generate EPKM/LV separation signals. The SC will receive the EPKM/LV separation signals once the break-wires circuitry break while EPKM/LV separates.

In the same way, there are two break-wires on the IFDs J1 & J2. The IFDs will send the EPKM/LV separation signal to LV once the break-wires circuitry break while EPKM/LV separates. This separation signal will be sent to LV's telemetry system through EY1 interface. The break-wire's allowable current:  $\leq 100\text{mA}$ , allowable voltage:  $\leq 30\text{V}$ .

### B5.2.2 Umbilical System

The umbilical system consists of onboard-LV Parts and ground parts. Refer to **Figure B5-5**, **Figure B5-6** and **Figure B5-7**. The 350m-cable from Launch Control Console (LCC) to Umbilical Tower, EB26/EB36, BOX3, BOX4, and Power-supply 1&2 are the common to different missions. The onboard-LV cable, as well as ground cable from WXTC to ED 13,14&15 and BOX1 & BOX2, will be designed for dedicated SC according to User's needs. In order to assure the quality of the product, the umbilical system will be provided to the User after pre-delivery acceptance test and insulation/conductivity checkouts in the launch site.



**Figure B5-5 LM-2E/EPKM Electrical Interface**

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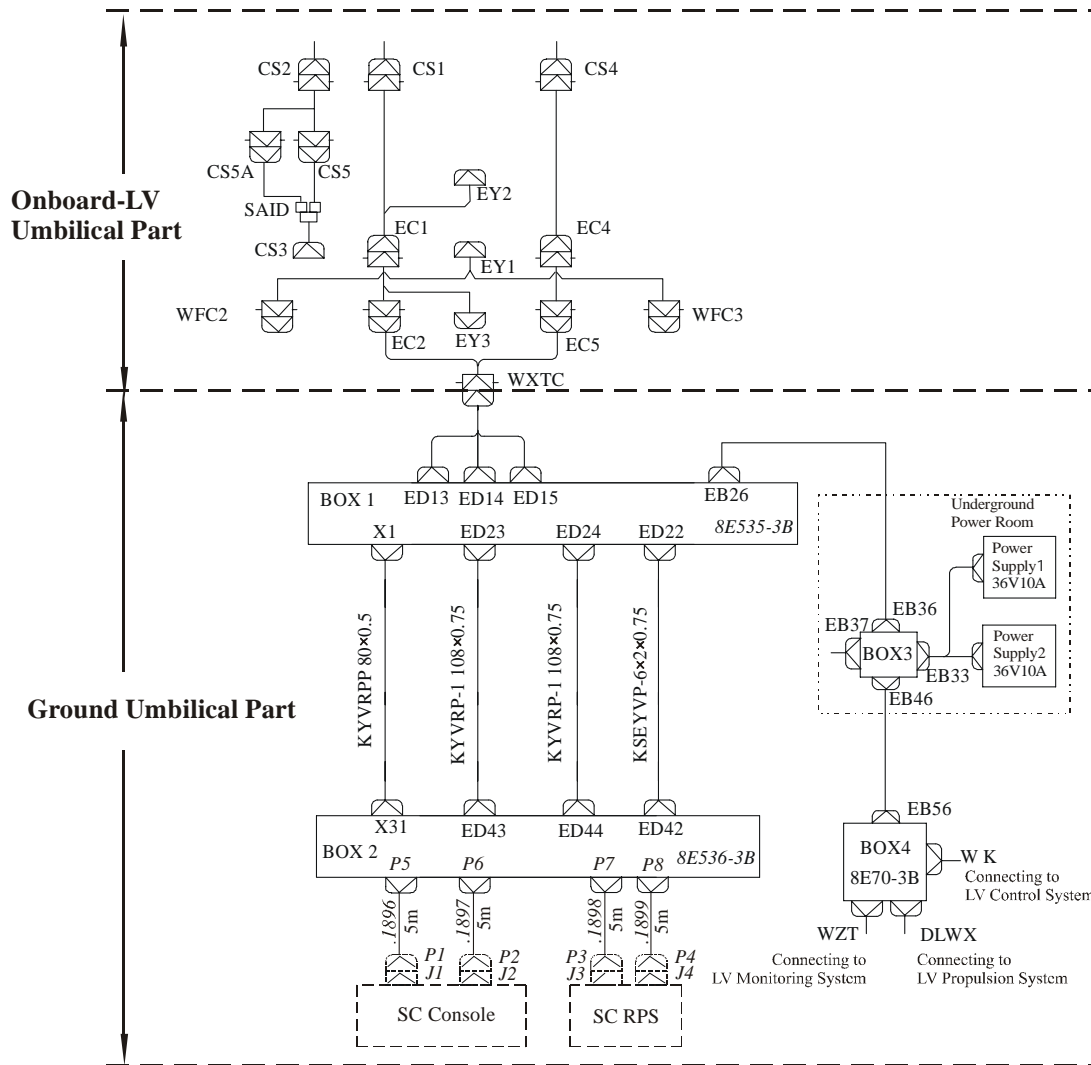


Figure B5-6 Umbilical Cable for SC

### B5.2.2.1 Onboard-LV Umbilical Cable

- Composition

The Onboard-LV cable net comprises the cables from the IFDs (P1, P2) to WXTC. These umbilical cables will fly with LV.

Whereas:

Code	Description
P1、 P2	LV/SC electrical connectors at LV side which is crimp-connected to the cables.
EC1、 EC4	Technological interfaces between SC adapter and LV
EY1	Interface between umbilical cable and LV TM system, through which the EPKM/LV separation signal is sent to LV TM system
WXTC	Umbilical cable connector (LV-Ground)
G1	Grounding points to overlap the shielding of wires and the shell of LV

Refer to **Figure B5-6**.

- **Umbilical Cable Design**

SC side shall specify characteristics of the IFDs. The specific contents are pin assignment, usage, maximum voltage, maximum current, one-way maximum resistance etc. CALT will design the umbilical cable according to the specified requirements.

- **Types and performance of Umbilical Cable**

Generally, ASTVR and ASTVRP wires are adopted for the onboard-LV cable net:  
 ASTVR, 0.5mm<sup>2</sup>, fiber-sheath, PVC insulation;  
 ASTVRP, 0.5mm<sup>2</sup>, fiber-sheath, PVC insulation, shielded.

For both cables, their working voltage is  $\leq 500V$  and DC resistance is 38.0Ω/km (20°C). The single core or cluster is shielded and sheathed.



### B5.2.2.2 Ground Umbilical Cable Net

- **Composition**

The ground umbilical cable net consists of umbilical cable connector (WXTC), cables, box adapters, etc. Refer to **Figure B5-6** and **Figure B5-7**.

Whereas:

Code	Description
WXTC	WXTC is umbilical cable connector (LV-Ground) whose female half (socket) is installed at the wall of the VEB, while the male half (pin) is attached to the top end of ground cable. The disconnection of WXTC is electrically controlled. (The disconnection is powered by BOX 3 and controlled by BOX 4. In the mean time, forced disconnection is also used as a spare separation method.) Generally, WXTC disconnects at about 8min prior to launch. If the launch was terminated after the disconnection, WXTC could be reconnected within 30min. The SC should switch over to internal power supply and cut off ground power supply at 5 minutes prior to WXTC disconnection. Therefore, during disconnection only a low current monitoring signal (such as 30V, $\leq 100\text{mA}$ ) is permitted to pass through the WXTC.
BOX1	BOX 1 is a box adapter for umbilical cable that is located inside the SC Cable Measurement Room on Floor 8.5 of the umbilical tower. (If needed, BOX 1 can provide more interfaces for the connection with SC ground equipment.)
BOX2	BOX 2 is another box adapter for umbilical cable that is located inside the SC Blockhouse on ground. Other SC ground support equipment (RPS, Console, etc.) are also located inside the Blockhouse.
BOX3	This is a relay box for the disconnection of the umbilical cable. BOX 3 is located inside the under-ground Power-Supply Room. Box 3 is powered by 2 DC regulated power supply sets. These two power supply sets are in "working-state" sparing to each other.
BOX4	BOX 4 is located inside Blockhouse. It is for the control of the pre-launch disconnection of SC umbilical cables.

### ● Interface on Ground

Generally, there are four interfaces on ground, namely, two for SC Console (P1/J1&P2/J2), and the other two for SC power supply (P3/J3&P4/J4). SC side will define the detailed requirement of ground interfaces. Those connectors (P1,P2,P3,P4) to be connected with SC ground equipment should be provided by SC side to LV side for the manufacture of cables.

Location	Code	Specification	Quantity
LV side interfaces	P1	To be defined by SC side	2
	P2		2
	P3		2
	P4		2

If LV side couldn't get the connectors from SC side, this ground interface cable will be provided in cores with pin marks.

SC side can also provide this ground cable. The length of this cable is about 5 meters. If so, LV side will provide the connectors (as Y11P-61) to connect with BOX 2.

### ● Type & Performance

The type and performance of the umbilical cables are listed in **Figure B5-6**.

#### ✧ Single-Core Shielded Cable

KYVRPP 80×0.5, Copper core, PV insulation, copper film plating on PV for shielding of each core, PVC sheath, woven wire net for shielding of cable; 80 cores/cable, 0.5mm<sup>2</sup>/core; Working voltage: ≤60V; DC resistance (20°C) of each core: 38.0Ω/km.

#### ✧ Ordinary Insulation Cable

KYVRP-1 108×0.75, copper core with PV insulation, PVC sheath, woven wire for shielding, flexible; 108 cores/cable, 0.75mm<sup>2</sup>/core; No shielding for each core, woven tin-plated copper wire for shielding of cable; Working voltage: ≤110V; DC resistance (20°C) of each core: 28.0Ω/km.

#### ✧ Twin-twist Shielded Cable

KSEYVP 6×2×0.75, 6 pairs of twin-twisted cores, 0.75mm<sup>2</sup>/core. Each twisted pair is shielded and the whole cable has a woven wire net for shielding. Impedance: 100Ω.

Twin-twist shielded cable (KSEYVP) are generally used for SC data transmission and communication. Single-core shielded cable (KYVRPP) is often used for common control and signal indicating. KYVRP-1 cable is adopted for SC's power supply on ground and multi-cores are paralleled to meet the SC's single-loop resistance requirement.

Under normal condition, the umbilical cable (both on-board and ground) has a insulation resistance of ≥10MΩ (including between cores, core and shielding, core and LV shell)

### **B5.2.3 Umbilical Cable Disconnect Control**

LV side is responsible for the pre-launch disconnection of umbilical cable through BOX3 and BOX 4.

Inside the underground Power Supply Room, there are two 36V/10A DC regulated power supply which will provide power for the cables. They are all in working condition sparing to each other.

Generally, according to the count-down launch procedure, only after LV side has received the confirmation that SC has turned to internal power and SC is normal, could the order of umbilical cable disconnection be sent out.

### **B5.2.4 Anti-lightning, Shielding and Grounding**

In order to assure the safety of the operations of both LV and SC, some measures have been taken for anti-lightning, shielding and grounding.

- ✧ The cable has two shielding layers, the outer shielding is for anti-lightning while the inner shielding is for anti-interference.
- ✧ For the cables from WXTC to BOX 2, the outer shielding (anti-lightning) has a grounding point every 20m. These grounding measures can assure the lightning and other inductance to be discharged immediately. The grounding locations are either on the swing rods or the cable's supporting brackets.
- ✧ The inner shield has a single grounding. The inner shields of the on-board cables are connected to BOX 2 through WXTC. BOX 2 has a grounding pole.

- ✧ The inner and outer shields are insulated with each other inside the cables.

### **B5.2.5 Continuity of SC “Earth-Potential”**

The SC should have a reference point of earth-potential and this benchmark should be near to the EPKM/LV separation plane. Generally, the resistance between all other metal parts of SC (shell, structures, etc.) and this benchmark should be less than  $10\text{m}\Omega$  under a current of  $10\text{mA}$ .

There is also a reference-point of earth-potential at the bottom of the adapter. The resistance between LV reference point at the adapter and SC reference should be less than  $10\text{m}\Omega$  with a current of  $10\text{mA}$ . In order to keep the continuity of earth-potential and meet this requirement, the bottom of SC to be mated with adapter should not be treated chemically or treated through any other methodology affecting its electrical conductivity.

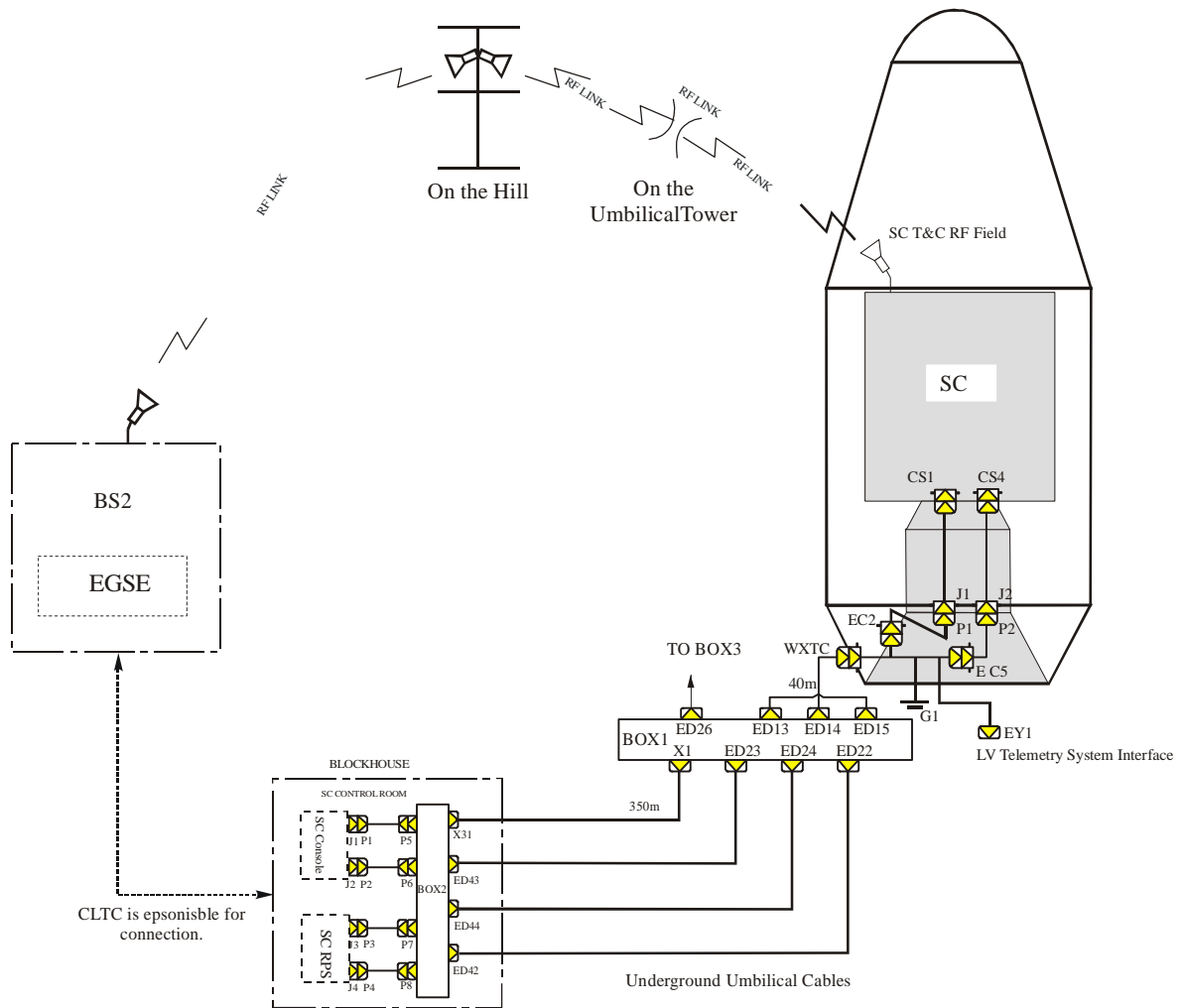
<b>SPECIAL STATEMENT</b>
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Any signal possibly dangerous to the flight can not be sent to the SC during the whole flight till EPKM/LV separation. Only EPKM/LV separation can be used as the initial reference for all SC operations. After EPKM/LV separation, SC side can control SC through microswitches and remote commands.

### B5.3 RF Links

#### B5.3.1 RF Relay Path

XSLC can provide RF link from EGSE to SC either in BS or on the umbilical tower. Refer to **Figure B5-7**.



**Figure B5-7 RF Links**

**B5.3.2 Characteristics of RF Link**

## (3) Frequency

C Band:           Up-link: 5925~6425 MHz  
                       Down-link: 3700~4200 MHz

Ku Band:           TBD

## (4) Signal Level

C Band: See following table  
 Ku Band: TBD

Frequency	SC Antenna		EGSE	
	EIRP	PFD	Input	Output
Telemetry	37dBm		-70dBm	
Command		-85dBW/m <sup>2</sup>		30dBm

**SPECIAL STATEMENT**

A mission dedicated RF working plan will be worked out. Anyway, the SC RF equipment should be turned off during the whole flight phase of LV until all SCs are separated from the LV hardware.