

# **FLIGHT MANUAL**

PART III - Normal Procedures

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## ABOUT THIS MANUAL

#### VERSION: 04 MAY, 2012

WARNING: THIS MANUAL IS DESIGNED FOR MICROSOFT® FSX USE ONLY. DO NOT USE FOR FLIGHT.

The '737 Captain' FLIGHT MANUAL is organized into four Parts: Each Part is provided as a separate Acrobat® PDF document:

Click START > Programs > Captain Sim > 737 Captain >

- Part I User's Manual
   The User's Manual describes the '737 Captain' Sim product as a software title.
- Part II Aircraft Systems
- Part III Normal Procedures this document
- Part IV Flight Crew Training Manual

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FOR GENERAL INFORMATION ON THE '737 CAPTAIN' PRODUCT PLEASE USE <u>WWW.CAPTAINSIM.COM</u> . THIS MANUAL PROVIDES ADDITIONAL INFORMATION ONLY, WHICH IS NOT AVAILABLE ON THE WEB SITE.

# 737 Captain FLIGHT MANUAL

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# **OPERATING LIMITATIONS**

This chapter contains Airplane Flight Manual (AFM) limitations and Boeing recommended operating information. Limitations that are obvious, shown on displays or placards, or incorporated within an operating procedure are not contained in this chapter.

Runway slope	+/- 2%
Maximum Takeoff and Landing Tailwind Component	15 knots
Maximum speeds	Observe Vmo pointer and gear/ flap placards
Turbulent airspeed	280 KIAS/.70M
Mach trim inoperative	max speed .74M
Maximum Operating Altitude	37,000 feet
Maximum Takeoff and Landing Altitude	8,300 feet

Verify that an operational check of the flight deck door access system (as installed) has been accomplished according to approved procedures once each flight day.

On revenue flights, the escape slide retention bar (girt bar) must be installed during taxi, takeoff and landing.

#### WEIGHT LIMITATIONS 737-200 AIRPLANES

CHARACTERISTICS	737-200	737-100	737-200ADV	737-200C/F
TAKEOFF WEIGHT, POUNDS	116,000	111,000	128,100	116,000
ZERO FUEL WEIGHT, POUNDS	95,000	90,000	95,000	95,000
OPERATING EMPTY WEIGHT, POUNDS	59,800	62,000	65,700	65,700

#### AIR SYSTEMS

The maximum cabin differential pressure (relief valves) is 8.65 psi.

#### ANTI-ICE, RAIN

Engine TAI must be on when icing conditions exist or are anticipated, except during climb and cruise below  $-40^{\circ}C$  SAT.

Minimum N1 RPM for operating in icing conditions except for landing: 40% when TAT between 0° and 10°C; 55% when TAT below 0°C; 70% in moderate to severe icing conditions when TAT below -6.5°C.

Window heat inop: max speed 250 KIAS below 10,000 ft.

#### AUTOPILOT

Use of autopilot not authorized for takeoff or landing.

Do not use autopilot roll channel above 30,000 feet with yaw damper inoperative.

Do not use autopilot pitch channel above .81M with hydraulic system A or B depressurized.

#### ELECTRICAL POWER

Max engine driven generator load: 111 amps.

Maximum generator drive oil temperature: 157° C

#### ENGINES

#### ENGINE LIMIT DISPLAY MARKINGS

Maximum and minimum limits are red. Caution limits are amber.

#### **GENERAL ENGINE LIMITATIONS**

Maximum N1 RPM	100.1%
Maximum N2 RPM	100 %
Maximum Acceleration EGT (2 minutes)	580° C
Maximum Takeoff EGT (5 minutes)	580° C
Maximum Continuous EGT	540° C
Maximum Start EGT Ambient Temperature above 15°C (momentary) Ambient Temperature below 15°C	420° C 350° C
Maximum Oil Temperature (continuous) (15 minutes)	120° C 121° C 157° C

#### OIL PRESSURE

Maximum Oil Pressure	55 psi
Minimum Oil Pressure	40 psi

#### **ENGINE IGNITION**

Engine ignition must be on during takeoff and landing.

#### **REVERSE THRUST**

Intentional selection of reverse thrust in flight is prohibited.

#### APU

Maximum start EGT is 760° C. Maximum continuous EGT is 710° C.

With APU bleed + electrical load, maximum altitude is 10,000 ft. With APU bleed, maximum altitude is 17,000 ft.

With APU electrical load, maximum altitude is 35,000 ft. APU bleed valve must be closed when:

- ground air connected and isolation valve open
- engine no. 1 bleed valve open
- isolation valve and engine no. 2 bleed valve open.

APU bleed valve may be open during engine start, but avoid engine power above idle.

#### FLIGHT CONTROLS

Flap extension altitude is 20,000 ft.

In flight, do not extend the SPEED BRAKE lever beyond the FLIGHT DETENT.

Avoid rapid and large alternating control inputs, especially in combination with large changes in pitch, roll, or yaw (e.g. large side slip angles) as they may result in structural failure at any speed, including below VA.

#### FLIGHT MANAGEMENT, NAVIGATION

Do not operate the weather radar in a hangar or within 50 feet of fueling operations or fuel spills.

Do not operate the weather radar within 160 feet of personnel. Warm up radar in STBY position only.

#### FUEL

Do not reset a tripped fuel pump circuit breaker. Maximum fuel temperature is 49° C.

Minimum fuel temperature is fuel freeze point +3° C or -45° C, whichever is higher.

## FUEL LOADING

On the ground, main tanks 1 and 2 must be full if center tank contains more than 1000 lbs.

#### LANDING GEAR

Do not apply brakes until after touchdown.

# NORMAL PROCEDURES: INTRODUCTION

This chapter gives:

- an introduction to the normal procedures philosophy and assumptions
- step by step normal procedures

#### NORMAL PROCEDURES PHILOSOPHY AND ASSUMPTIONS

Normal procedures verify for each phase of flight that:

- the airplane condition is satisfactory
- the flight deck configuration is correct

Normal procedures are done on each flight. Refer to the Supplementary Procedures (SP) chapter for procedures that are done as needed, for example the adverse weather procedures.

Normal procedures are used by a trained flight crew and assume:

- all systems operate normally
- the full use of all automated features.

Normal procedures also assume coordination with the ground crew before:

- hydraulic system pressurization, or
- flight control surface movement, or
- airplane movement

Normal procedures do not include steps for flight deck lighting and crew comfort items.

Normal procedures are done by recall and scan flow. The panel illustration in this section shows the scan flow.

The scan flow sequence may be changed as needed.

#### CREW DUTIES

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Preflight and postflight crew duties are divided between the captain and first officer. Phase of flight duties are divided between the Pilot Flying (PF) and the Pilot Monitoring (PM.)

Each crewmember is responsible for moving the controls and switches in their area of responsibility:

• The phase of flight areas of responsibility for both normal and non-normal procedures are shown in the Area of Responsibility illustrations in this section. Typical panel locations are shown.

• The preflight and postflight areas of responsibility are defined by the "Preflight Procedure - Captain" and "Preflight Procedure - First Officer".

The captain may direct actions outside of the crewmember's area of responsibility.

The general PF phase of flight responsibilities are:

- taxiing
- flight path and airspeed control
- airplane configuration
- navigation

The general PM phase of flight responsibilities are:

- checklist reading
- communications
- tasks asked for by the PF
- monitoring taxiing, flight path, airspeed, airplane configuration, and navigation

PF and PM duties may change during a flight. For example, the captain could be the PF during taxi but be the PM during takeoff through landing.

Normal procedures show who does a step by crew position (C, F/O, PF, or PM):

- in the procedure title, or
- in the far right column, or

in the column heading of a table

The mode control panel is the PF's responsibility. When flying manually, the PF directs the PM to make the changes on the mode control panel.

The captain is the final authority for all tasks directed and done.

## SCAN FLOW AND AREAS OF RESPONSIBILITY

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The scan flow and areas of responsibility diagrams shown below are representative and may not match the configuration of your airplane.

The scan flow diagram provides general guidance on the order of each flight crew member should follow when doing the preflight procedures. Specific guidance on the items to be checked are detailed in the amplified Normal Procedures, Preflight Procedure - Captain and Preflight Procedure - First Officer.

# PREFLIGHT AND POSTFLIGHT SCAN FLOW



# AREAS OF RESPONSIBILITY - CAPTAIN AS PILOT FLYING OR TAXIING



# AREAS OF RESPONSIBILITY - FIRST OFFICER AS PILOT FLYING OR TAXIING



# NORMAL PROCEDURES: AMPLIFIED PROCEDURES

#### PRELIMINARY PREFLIGHT PROCEDURE - CAPTAIN OR FIRST OFFICER

The Preliminary Preflight Procedure assumes that the Electrical Power Up supplementary procedure is complete.

Electronic master switches	ON
Circuit breakers (P6 panel)	Check
Circuit breakers (P18 panel)	Check
Parking brake	As needed

## EXTERIOR INSPECTION

Before each flight the captain, first officer, or maintenance crew must verify that the airplane is satisfactory for flight.

Items at each location may be checked in any sequence.

Use the detailed inspection route below to check that:

- the surfaces and structures are clear, not damaged, not missing parts and there are no fluid leaks
- the tires are not too worn, not damaged, and there is no tread separation
- the gear struts are not fully compressed
- the engine inlets and tailpipes are clear, the access panels are secured, the exterior is not damaged, and the reversers are stowed
- the doors and access panels that are not in use are latched
- the probes, vents, and static ports are clear and not damaged
- the skin area adjacent to the pitot probes and static ports is not wrinkled
- the antennas are not damaged
- the light lenses are clean and not damaged



## LEFT FORWARD FUSELAGE

Probes, sensors, ports, vents, and drains (as applicable)Check	
Doors and access panels (not in use)Latched	
Main deck cargo door (as installed)Check	

## NOSE

Radome.....Check

Conductor straps - Secure

Forward E and E door.....Secure

# NOSE WHEEL WELL

Tires and wheels	Check
Exterior light	Check
Gear strut and doors	Check
View port	Clear and clean
Nose wheel steering assembly	Check

## RIGHT FORWARD FUSELAGE

Probes, sensors, ports, vents, and drains (as applicable)Cheo	ck
Doors and access panels (not in use)Latche	d

# RIGHT WING ROOT, PACK, AND LOWER FUSELAGE

Ram air deflector door	.Extended
Pack and pneumatic access doors	Secure
Probes, sensors, ports, vents, and drains (as applicable)	Check
Exterior lights	Check
Leading edge flaps	Check

# NUMBER 2 ENGINE

Access panelsLatched
Probes, sensors, ports, vents, and drains (as applicable)Check
Fan blades, probes, and spinnerCheck

DO NOT USE FOR FLIGHT	
Thrust reversers	Stowed
Exhaust area and tailcone	Check

## RIGHT WING AND LEADING EDGE

Leading edge slats	Check
Wing Surfaces	Check

## RIGHT WING TIP AND TRAILING EDGE

Position lights	Check
Static discharge wicks	Check
Aileron and trailing edge flaps	Check
Exterior lights	Check

## RIGHT MAIN GEAR

Tires, brakes and wheels Verify that the wheel chocks are in place as needed.	Check
Gear strut, actuators, and doors	Check
Hydraulic linesS	Secure

## RIGHT MAIN WHEEL WELL

View port	Clear and clean
Wheel well	.Check

# RIGHT AFT FUSELAGE

Doors and access panels (not in use)	Latched
Negative pressure relief door	Closed
Outflow valve	Check
Probes, sensors, ports, vents, and drains (as applicable)	Check
APU air inlet	Open

## TAIL

Vertical stabilizer and rudder	Check
Elevator feel probes	Check
Horizontal stabilizer and elevator	Check
Static discharge wicks	Check
APU exhaust outlet	Check

Verify that there is no indication of scorch marks on the outlet.

# LEFT AFT FUSELAGE

Doors and access panels (not in use)Latched	
Probes, sensors, ports, vents, and drains (as applicable)Check	

# LEFT MAIN WHEEL WELL

View portClear an	d clean
Wheel well	.Check

# LEFT MAIN GEAR

Tires, brakes and wheelsC	heck
Verify that the wheel chocks are in place as needed.	
Gear strut, actuators and doorsC	heck
Hydraulic linesS	Secure

# LEFT WING TIP AND TRAILING EDGE

Aileron and trailing edge flapsCh	ieck
Static discharge wicksCh	ieck
Position lightsCh	ieck
Exterior lightsCh	ieck

#### LEFT WING AND LEADING EDGE

Wing SurfacesChe	ck
Leading edge slatsChe	ck

## NUMBER 1 ENGINE

Exhaust area and tailcone	Check
Thrust reversers	Stowed
Fan blades, probes, and spinner	Check
Probes, sensors, ports, vents, and drains (as applicable)	Check
Access panelsL	atched

## LEFT WING ROOT, PACK, AND LOWER FUSELAGE

Leading edge flaps	Check
Probes, sensors, ports, vents, and drains (as applicable)	Check
Exterior lights	Check

## PREFLIGHT PROCEDURE - FIRST OFFICER

The first officer normally does this procedure. The captain may do this procedure if needed.

Flight control panel ..... Check

FLIGHT CONTROL switches - Guards closed

Flight SPOILER switches - Guards closed

YAW DAMPER switch - ON

NAV transfer switches.....NORMAL

Verify that the FUEL VALVE CLOSED lights are illuminated dim.

Verify that the FILTER ICING lights are extinguished.

Fuel HEAT switches - OFF

Verify that the VALVE OPEN lights are extinguished.

CROSSFEED selector - CLOSED

Verify that the VALVE OPEN light is extinguished.

FUEL PUMP switches - OFF

Verify that the center tank fuel pump LOW PRESSURE lights are extinguished.

Verify that the auxiliary tank fuel pump LOW PRESSURE lights (as installed) are extinguished.

Verify that the main tank fuel pump LOW PRESSURE lights are illuminated.

Electrical panel......Set

BATTERY switch - Guard closed

GALLEY power switch - ON

STANDBY POWER switch - Guard closed

Verify that the STANDBY PWR OFF light is extinguished.

Generator drive DISCONNECT switches - Guards closed

Verify that the LOW OIL PRESSURE lights are illuminated.

Verify that the HIGH OIL TEMP lights are extinguished.

DRIVE TEMPERATURE switch - As needed

BUS TRANSFER switch - Guard closed

Verify that the TRANSFER BUS OFF lights are extinguished.

Verify that the BUS OFF lights are extinguished.

Verify that the GEN OFF BUS lights are illuminated.

- Overheat and fire protection panel (Passenger airplanes)
- APU switch (as needed).....START

When the APU GEN OFF BUS light is illuminated:

APU GENERATOR bus switches – ON

Verify that the BUS OFF lights are extinguished.

Verify that the TRANSFER BUS OFF lights are extinguished.

Verify that the LOW OIL QUANTITY light is extinguished.

Verify that the APU LOW OIL PRESSURE light is extinguished.

Verify that the APU HIGH OIL TEMP light is extinguished.

Verify that the APU OVERSPEED light is extinguished.

EQUIPMENT COOLING switch.....NORMAL

Verify that the OFF light is extinguished.

EMERGENCY EXIT LIGHTS switch......Guard closed

Verify that the NOT ARMED light is extinguished.

Passenger signs......Set

NO SMOKING switch - AUTO or ON

Check

Verify that the PACK TRIP OFF lights are extinguished.	
Verify that the WING-BODY OVERHEAT lights are extingu	ished.
Verify that the BLEED TRIP OFF lights are extinguished.	
Cabin pressurization panel	Set
Verify that the AUTO FAIL light is extinguished.	
Verify that the OFF SCHED DESCENT light is extinguished	I.
FAILURE ALTITUDE indicator - Cruise altitude	
LANDING ALTITUDE indicator - Destination field elevation	1
CABIN RATE selector – Index	
CABIN ALTITUDE indicator - 200 feet below destination fi	eld elevation
FLIGHT/GROUND switch - GRD Pressurization mode select	tor – AUTO
Verify that the STANDBY light is extinguished. Verify that the STANDBY light is extinguished.	erify that the MANUAL light is extinguished.
Lighting panel	Set
LANDING light switches - RETRACT and OFF	
RUNWAY TURNOFF light switches - OFF	
TAXI light switch – OFF	
ENGINE START switches	OFF
Lighting panel	Set
POSITION light switch - As needed	
ANTI-COLLISION light switch - OFF	
WING illumination switch - As needed	
WHEEL WELL light switch - As needed	
Flight director panel	Set
Mode selector – OFF	
ALTITUDE HOLD switch – OFF	
PITCH COMMAND control - Full clockwise	
Oxygen	Test and set
STATIC SOURCE SELECTOR switch	NORMAI
Marker beacon lights	
Clock	
Autopilot disengage light	
Verify that the AUTOPILOT disengage light is illu	
Flight instruments Che	
Set the altimeter.	

Verify that the flight instrument indications are correct. Verify that only these flags are shown: TCAS (as installed) disengage light is illuminated. expected RMI flags Hydraulic system B LOW QUANTITY light..... Verify extinguished SYSTEM A HYDRAULIC QUANTITY indicator..... Above RF GROUND PROXIMITY panel..... Check FLAP/GEAR INHIBIT switch - Guard closed Verify that the INOP light is extinguished. Landing gear panel.....Set LANDING GEAR lever - DN Verify that the green landing gear indicator lights are illuminated. Verify that the red landing gear indicator lights are extinguished. TAKEOFF CONFIG light (as installed) ......Verify extinguished CABIN ALTITUDE light (as installed) ......Verify extinguished ANTISKID switches ...... Guards closed Verify that the ANTISKID INOP lights are extinguished. AUTOBRAKE selector..... OFF Verify that the AUTO BRAKE DISARM light is extinguished. EPR reference selectors (on PDCS equipped airplanes) .....Push Engine instruments...... Check Verify that the REVERSER UNLOCKED lights are extinguished. Verify that the START VALVE OPEN lights are extinguished. Verify that the LOW OIL PRESSURE lights are illuminated. Verify that the OIL FILTER BYPASS lights are extinguished. Verify that the primary and secondary engine indications show existing conditions. ENGINE OIL QUANTITY TEST switch..... Push Verify that the oil quantity indicators move toward zero and return to the original position when the switch is released. VHF communications radios ...... Set VHF NAVIGATION radios ....... Audio selector panel ..... Set ADF radios..... Set WEATHER RADAR panel..... Set

Transponder panel..... Set

STABILIZER BRAKE RELEASE knob ......Verify released

# WARNING: Do not put objects between the seat and the aisle stand. Injury can occur when the seat is adjusted.

Seat.....Adjust

Adjust the seat for optimum eye reference.

Verify a positive horizontal (fore and aft) seat lock.

Rudder pedals..... Adjust

Adjust the rudder pedals to allow full rudder pedal and brake pedal movement.

Seat belt and shoulder harness..... Adjust

Do the PREFLIGHT checklist on the captain's command.

## PREFLIGHT PROCEDURE - CAPTAIN

The captain normally does this procedure. The first officer may do this procedure if needed.

Lights Set
Flight director panel Set
Mode selector - OFF
ALTITUDE HOLD switch - OFF
PITCH COMMAND control - Full clockwise
Autopilot panel Set
Autopilot mode selector - MAN
Autopilot system select switch - As needed
Autopilot heading switch - Centered position
Autopilot AILERON engage switch - DISENGAGED
Autopilot ELEVATOR engage switch - DISENGAGED
Autopilot pitch mode selector - OFF
Oxygen Test and set
STATIC SOURCE SELECTOR switch NORMAL
Marker beacon lights Test
Clock Set
Autopilot disengage light Push to test
Verify that the AUTOPILOT disengage light is illuminated.
Flight instruments Check
Set the altimeter

Verify that the flight instrument indications are correct.

V • •	/erify that only these flags are shown: TCAS (as installed) expected RMI flags	
Standby ir	nstruments	Check
V	/erify that the flight instrument indications are corr	ect
V	/erify that no flags are shown.	
STAB OUT	F OF TRIM light	Verify extinguished
SPEED BR	AKE lever	DOWN detent
V	/erify that the SPEED BRAKE ARMED light is extingu	lished.
V	/erify that the SPEED BRAKE DO NOT ARM light is e	extinguished.
Reverse th	hrust levers	Down
Forward tl	hrust levers	Closed
FLAP lever	r	Set
S	Set the flap lever to agree with the flap position.	
Parking br	rake	Set
V	/erify that the parking brake warning light is illumir	ated
	Note: Do not assume that the parking brake will pre be insufficient.	event airplane movement. Accumulator pressure can
Engine sta	art levers	CUTOFF
STABILIZE	ER TRIM cutout switches	NORMAL
VHF comm	nunications radios	Set
VHF NAVI	GATION radios	Set for departure
Audio sele	ector panel	Set
	WARNING: Do not put objects between the sea he seat is adjusted.	at and the aisle stand. Injury can occur when
Seat		Adjust
А	djust the seat for optimum eye reference.	
V	/erify a positive horizontal (fore and aft) seat lock.	
Rudder pe	edals	Adjust
А	Adjust the rudder pedals to allow full rudder pedal a	nd brake pedal movement.
Seat belt	and shoulder harness	Adjust
Call "PREF	FLIGHT CHECKLIST."	

#### BEFORE START PROCEDURE

Start the Before Start Procedure after papers are on board.

Flight deck door..... Closed and locked F/O

Verify that the CAB DOOR UNLOCKED light (as installed) is extinguished.

Verify that the LOCK FAIL light (as installed) is extinguished.

Do the Performance Data Computer System (as installed) Preflight Supplementary Procedure.

PDCS CDU flight mode selector (as installed).....As needed C, F/O

Takeoff data.....Complete C, F/O

Verify the takeoff data to include:

- EPRN1
- V1, VR, and V2
- vi, vit, ullu
- flap setting
- zero fuel weight
- temperature
- altimeter setting
- gross weight
- stabilizer trim setting

Fuel quantity indicators.....Check C, F/O

Verify that the fuel on the dispatch papers and fuel quantity indicators agree.

Verify that the fuel is sufficient for flight.

**Note:** Do not push the QUANTITY TEST switch when the airplane is being refueled. This will cause incorrect indications at the external fueling panel.

С

С

С

C, F/O

Total fuel and VREF indicator.....Set

Zero fuel weight - Set

Flap selector - As needed

Verify VREF on the VREF pointer.

On airplanes without PDCS,

EPR reference selectors......Set

Verify that the EPR reference bugs and digital readouts are correct. On airplanes with PDCS,

EPR reference selectors.....In

Verify that the PDCS reference bugs and digital readouts are correct.

IAS	bugs	.Set (	C, F	=/C	)
-----	------	--------	------	-----	---

Set the speed bugs at V1, VR, V2 + 15, and flaps up maneuvering speed.

Airspeed curso	r controls	Set V2	C, F/O

HSI HEADING selectors......Set

DO NOT USE FO	
HSI course selectors	Set C, F/O
ALTITUDE alert controller	Set C
Taxi and Takeoff briefingsComp	lete C, F/O
The pilot who will do the takeoff does the taxi and takeo	off briefings.
Exterior doorsVerify clo	sed F/O
Flight deck windowsClosed and loci	ked C, F/O
Start clearanceOt	otain C, F/O
Obtain a clearance to pressurize the hydraulic systems.	
Obtain a clearance to start the engines.	
ANTI COLLISION light switch	ON F/O
Trim	Set C
Check each trim for freedom of movement.	
Stabilizer trimUNITS	
Set the trim for takeoff.	
Verify that the trim is in the green band.	
Aileron trim - 0 units	
Rudder trim - 0 units	
Call "BEFORE START CHECKLIST.	С
Do the BEFORE START checklist.	F/O

# PUSHBACK OR TOWING PROCEDURE

The Engine Start procedure may be done during pushback or towing.

Establish communications with ground handling personnel.

CAUTION: Do not hold or turn the nose wheel steering wheel during pushback or towing. This can damage the nose gear or the tow bar.

CAUTION: Do not use airplane brakes to stop the airplane during pushback or towing. This can damage the nose gear or the tow bar.

Set or release the parking brake as directed by ground handling personnel. C

When pushback or towing is complete:

Verify that the tow bar is disconnected	С
Verify that the nose gear steering lockout pin is removed	С
System A HYDRAULIC PUMPS switches - ON	F/O

## ENGINE START PROCEDURE

Air conditioning PACK switchesOFF F/O
Start pressurePSI F/O
The minimum start pressure at sea level is 30 psi. Decrease the minimum start pressure 0.5 psi for each 1,000 feet above sea level.
Start sequenceAnnounce C
Call "STARTENGINE" C
ENGINE START switchGRD F/O
Verify that the N2 RPM increases. C, F/O
Verify that the oil pressure increases and call "OIL PRESSURE RISING" F/O
When N1 rotation is seen and N2 is at 20%, or (if 20% N2 is not possible), at maximum motoring and a minimum of 15% N2:
Engine start leverIDLE C
Monitor fuel flow and EGT indications. C, F/O
At 35-40% N2, verify that the ENGINE START switch moves to OFF. If not, move the ENGINE START switch to OFF. F/O
Verify that the duct pressure increases when the ENGINE START switch moves to OFF. F/O
Verify that the START VALVE OPEN light extinguishes when the ENGINE START switch moves to OFF. F/O
Call "STARTER CUTOUT " F/O

Monitor N1, N2, EGT, fuel flow and oil pressure for normal indications while the engine accelerates to a stable idle. C, F/O

Do the ABORTED ENGINE START checklist for one or more of the following abort start conditions:

- there is no N1 rotation by 20% N2
- there is no oil pressure increase by 30 seconds
- the fuel flow is greater than 1100 pph/500kgph at start
- the EGT does not increase by 20 seconds after the engine start lever is moved to IDLE
- the N1 or N2 does not increase or increases very slowly after the EGT increases
  - the EGT quickly nears or exceeds the start limit

#### BEFORE TAXI PROCEDURE

Fuel HEAT switchesAs needed Before takeoff with tank fuel temperature 0° C or below, set the f Fuel heat must be OFF for takeoff.	F/O Fuel HEAT switches to ON for one cycle.
GENERATOR 1 and 2 switchesON	F/O
PITOT HEAT switchesON	F/O
WING ANTI-ICE switchAs needed	F/O
ENGINE ANTI-ICE switchesAs needed	F/O
Flight recorder REPEAT switchPush	F/O
PACK switchesON	F/O
ISOLATION VALVE switchAUTO	F/O
APU BLEED air switchOFF	F/O
Flight/Ground switchFLT	F/O
APU switchOFF	F/O
ENGINE START switchesLOW IGN	F/O
Engine start leversIDLE detent	С
Verify that the ground equipment is clear.	C, F/O
Call "FLAPS" as needed for takeoff.	С
Flap leverSet takeoff flaps	F/O

Verify that the LE FLAPS EXT green light is illuminated.

Make slow and deliberate inputs, one direction at a time.

Move the control wheel and the control column to full travel in both directions and verify:

- freedom of movement
- that the controls return to center

Hold the nose wheel steering wheel during the rudder check to prevent nose wheel movement.

<b>'737 Captain'</b> FLIGHT MANUAL Part III –Normal Procedures	
DO NOT USE FOR FLIGH	
Move the rudder pedals to full travel in both directions and verify	:
<ul> <li>freedom of movement</li> </ul>	
<ul> <li>that the rudder pedals return to center</li> </ul>	
TransponderAs needed	F/O
Recall Check	C, F/O
Verify that all system annunciator panel lights illuminate and ther	ı extinguish.
Update changes to the taxi briefing, as needed.	C or PF
Call "BEFORE TAXI CHECKLIST "	С
Do the BEFORE TAXI checklist.	F/O

## BEFORE TAKEOFF PROCEDURE

Engine warm up recommendations (there is no need to delay the takeoff for these recommendations):

When the engines have been shut down more than 2 hours:

- run the engine for 5 minutes
- when taxi time is expected to be less than 5 minutes, start the engines as early as feasible
- use a thrust setting normally used for taxi operations.

Pilot Flying		Pilot Monitoring
	Notify the cabin crew to prepare for takeoff. Verify that the cabin is secure.	
The pilot who will do the takeoff updates c	hanges to	the takeoff briefing as needed.
Set the weather radar display as needed.		
Call "BEFORE TAKEOFF CHECKLIST."		Do the BEFORE TAKEOFF checklist.
Pilot Flying		Pilot Monitoring
		When entering the departure runway, use lights as needed.
Verify that the brakes are released. Align the airplane with the runway.		When cleared for takeoff, set the INBOARD LANDING light switches to ON. Set the transponder mode selector to TA/RA (as installed).
Advance the thrust levers to approximately 1.4 EPR (levers in vertical position).		
Allow the engines to stabilize.		
Advance thrust levers to takeoff EPR.		
Verify that the correct takeoff thrust is set.		
Monitor the engine instruments during the takeoff. Call out any abnormal indications.		
	Adjust tal	keoff thrust before 60 knots as needed.
After takeoff thrust is set, the captain's hand must be on the thrust levers until V1.		

Monitor airspeed.	Monitor airspeed and call out any abnormal
	incitations.
Maintain light forward pressure on	
the control column.	
Verify 80 knots and call "CHECK".	Call "80 KNOTS."
Verify V1 speed.	Call "V1"
At VR, rotate toward 15° pitch attitude. Establish a positive rate of climb.	At VR call "ROTATE." Monitor airspeed and vertical speed.
Verify a positive rate of climb on the altimeter and call "GEAR UP "	Verify a positive rate of climb on the altimeter and call "POSITIVE RATE."
Maintain a minimum of V2 + 15 to 25 after the initial climb is established.	Set the landing gear lever to UP.
At thrust reduction height, reduce thrust	Set climb EPR.
to approximately 90% N1 and call "SET CLIMB THRUST."	
Verify that climb thrust is set.	
At acceleration height, call "SET FLAPS SPEED."	Set the flaps up maneuvering speed.
Verify acceleration.	Set the FLAP lever as directed. Monitor flaps
Call "FLAPS " according to the flap retraction schedule.	and slats retraction.
After flap retraction is complete and above	After flap retraction is complete:
minimum altitude for autopilot	Set or verify engine bleeds and air conditioning
engagement:	packs are operating
<ul> <li>engage the autopilot.</li> </ul>	Set the engine start switches as needed
	Set the AUTOBRAKE selector to OFF.
	Set the landing gear lever to OFF after landing gear retraction is complete
Call "AFTER TAKEOFF CHECKLIST."	Do the AFTER TAKEOFF checklist.

# TAKEOFF FLAP RETRACTION SPEED SCHEDULE

Takeoff Flaps	At and Below 117,000 LB	Above 117,000 LB	Select Flaps
25	V2 + 15	V2 + 15	15
	150 170 190	160 180 200	5 1 UP
15 or 10	V2 + 15 170 190	V2 + 15 180 200	5 1 UP
5 or 2	V2 + 15 190	V2 + 15 200	1 UP
1	190	200	UP
Limit bank angle to 15° until reaching V2 + 15.			

# CLIMB AND CRUISE PROCEDURE

Complete the After Takeoff Checklist before starting the Climb and Cruise Procedure.

Pilot Flying	Pilot Monitoring
	At or above 10,000 feet MSL, set the LANDING light switches to OFF.
	Set the passenger signs as needed.
At transition altitude, set and crossche	ck the altimeters to standard.
	During climb, set both center tank fuel pump switches to OFF when both center tank fuel pump LOW PRESSURE lights illuminate.
	When established in a level flight attitude, if the center tank contains usable fuel and the center tank fuel pump switches are OFF, set both center tank fuel pump switches to ON again Set both center tank fuel pump switches to OFF when both center tank fuel pump LOW PRESSURE lights illuminate.
	During the last hour of cruise on ETOPS flights, do a Fuel Crossfeed Valve check.
	Before the top of descent, modify the route as needed for the arrival and approach.

# DESCENT PROCEDURE

Start the Descent Procedure before the airplane descends below the cruise altitude for arrival at destination. Complete the Descent Procedure by 10,000 feet MSL.

Pilot Flying	Pilot Monitoring
	Set both center tank fuel pump switches to OFF when both center tank fuel pump LOW PRESSURE lights illuminate.
	If in level flight for an extended time, with usable fuel in the center tank, and the center tank fuel pump switches OFF, both center tank fuel pump switches may be set to ON again. Set both center tank fuel pump switches to OFF when both center tank fuel pump LOW PRESSURE lights illuminate.
	Verify that pressurization is set to landing altitude.
Review the system annunciator lights.	Recall and review the system annunciator lights.
Set the speed bugs at VREF, VREF + 15	, and flaps up maneuvering speed.
Set radio altimeter minimums as neede	d for the approach.
	Check and set EPR bugs for the GO-AROUND, corrected for the bleed configuration.
Set or verify the navigation radios and course for the approach.	
	Set the AUTOBRAKE selector to the needed brake setting.
Do the approach briefing.	
Call "DESCENT CHECKLIST."	Do the DESCENT checklist.

# APPROACH PROCEDURE

The Approach Procedure is normally started at transition level. Complete the Approach Procedure before:

- the initial approach fix, or
- the start of radar vectors to the final approach course, or
- the start of a visual approach

If a flaps 15 landing is needed because of performance:

GROUND PROXIMITY flap/gear inhibit switch

# FLAP/GEAR INHIBIT F/O

Pilot Flying	Pilot Monitoring
	Set the passenger signs as needed.
	At or above 10,000 feet MSL, set the INBOARD LANDING light switches to ON.
At transition level, set and crossche	ck altimeters.
Update changes to the arrival and a	pproach, as needed.
Update the approach briefing as needed.	
Call "APPROACH CHECKLIST."	Do the APPROACH checklist.

## FLAP EXTENSION SCHEDULE

Current Flap Position	At Speed (knots)	Select Flaps	Command Speed for Se- lected Flaps
Up	210	1	190
1	190	5	170
5	170	15	160
10*	160	15	150/VREF
15	150/VREF	25	140
25	140	30 or 40	(VREF30 or VREF40) + wind additives

\* As needed.

# LANDING PROCEDURE

Pilot Flying	Pilot Monitoring
	Notify the cabin crew to prepare for landing. Verify that the cabin is secure.
Call "FLAPS" according to the flap extension schedule.	Set the flap lever as directed. Monitor flaps and slats extension.
When on localizer intercept heading:	
<ul> <li>verify that the ILS is tun</li> <li>verify that the LOC and (</li> </ul>	
Select AUTO APP.	
Use HDG SEL to intercept the final approach course as needed.	
Verify that the localizer is captured.	
	Call "GLIDE SLOPE ALIVE."
At glide slope alive, call:	Set the landing gear lever to DN.
<ul> <li>"GEAR DOWN"</li> <li>"FLAPS 15"</li> </ul>	Verify that the green landing gear indicator lights are illuminated. Set the flap lever to 15. Set the engine start switches to LOW IGN.
Set the speedbrake lever to ARM.	
Verify that the SPEED BRAKE ARMED light is illuminated.	
At glide slope capture, call "FLAPS " as needed for landing.	Set the flap lever as directed.
	Set the missed approach altitude on the ALTITUDE ALERT controller.
Call "LANDING CHECKLIST."	Do the LANDING checklist.
At the final approach fix or OM, verify the crossing altitude.	
Monitor the approach.	
Disengage the autopilot prior to landir	ng.

# GO-AROUND AND MISSED APPROACH PROCEDURE

Pilot Flying	Pilot Monitoring
At the same time: • push either go-around switch • disengage autopilot • advance the thrust levers to go-	Monitor EPR indication. Set the FLAP lever to 15 and monitor flap retraction.
<ul> <li>around EPR</li> <li>Rotate to go-around attitude</li> <li>call "FLAPS 15"</li> </ul>	Adjust thrust as needed.
Verify: • the rotation to go-around • that the thrust increases	
	Verify that the thrust is sufficient for the go-around or adjust as needed.
Verify a positive rate of climb on the altimeter and call "GEAR UP."	Verify a positive rate of climb on the altimeter and call "POSITIVE RATE."
	Set the landing gear lever to UP.
	Verify that the missed approach altitude is set.
Call "TUNE RADIOS FOR MISSED APPROACH."	Tune the navigation radios as directed.
Verify that the missed approach route	is tracked.
At acceleration height, call "FLAPS" according to the flap retraction schedule.	Position the FLAP lever as directed. Monitor flaps and slats retraction.
Verify that climb thrust is set.	
Verify that the missed approach altitu	de is captured.
	Set the landing gear lever to OFF after landing gear retraction is complete.
	Set the engine start switches as needed.
Call "AFTER TAKEOFF CHECKLIST."	Do the AFTER TAKEOFF checklist.

# LANDING ROLL PROCEDURE

Pilot Flying	Pilot Monitoring
Verify that the thrust levers are closed.	Verify that the SPEED BRAKE lever is UP.
Verify that the SPEED BRAKE lever is UP.	Call "SPEED BRAKES UP."
Without delay, fly the nose wheel smoothly onto the runway.	If the SPEED BRAKE lever is not UP, cal "SPEED BRAKES NOT UP."
	Monitor the rollout progress.
Verify correct autobrake operation.	
CAUTION: start to lower the nose before selecting reverser doors from touching the runway. Without delay, move the reverse thrust levers to the interlocks and hold light pressure until the interlocks	reverse thrust to prevent the
release. Then apply reverse thrust as needed.	
By 60 knots, start movement of the reverse thrust levers to be at the reverse idle detent before taxi speed.	Call "60 KNOTS."
levers to be at the reverse idle detent before taxi	

# AFTER LANDING PROCEDURE

Start the After Landing Procedure when clear of the active runway. Engine cooldown recommendations:

- Run the engines for at least 5 minutesUse a thrust setting no higher than that is normally used for all engine taxi operations.

Pilot Flying	Pilot Monitoring	
The captain moves or verifies that the SPEED BRAKE lever is DOWN.		
	Start the APU, as needed.	
	Set the PITOT STATIC HEAT switches to OFF.	
	Set the Flight/Ground switch to GRD.	
	Set the LANDING and TAXI light switches as needed.	
	Set the ENGINE START switches to OFF.	
	Set the AUTOBRAKE selector to OFF.	
	Set the flap lever to UP.	
Set the weather radar to OFF.		
	Set the transponder mode selector as needed.	

# SHUTDOWN PROCEDURE

Start the Shutdown Procedure after taxi is complete.
Parking brakeSet C
Verify that the parking brake warning light is illuminated.
Electrical power Set F/O
If APU power is needed: Verify that the APU GENERATOR OFF BUS light is illuminated.
APU GENERATOR bus switches - ON Verify that the BUS OFF lights are extinguished.
If external power is needed:
Verify that the GND POWER AVAILABLE light is illuminated.
GROUND POWER switch - ON Verify that the BUS OFF lights are extinguished.
Before engine shutdown, consider engine cooldown recommendations.
Engine start leversCUTOFF C
If towing is needed:
Establish communications with ground handling personnel C
System A HYDRAULIC PUMP switches - OFF
Verify that the system A pump LOW PRESSURE lights are illuminated.
CAUTION: Do not hold or turn the nose wheel steering wheel during pushback or towing. This can damage the nose gear or the tow bar.
CAUTION: Do not use airplane brakes to stop the airplane during pushback or towing. This can damage the nose gear or the tow bar.
Set or release parking brake as directed by ground handling personnel
When towing is complete:
System A HYDRAULIC PUMP switches – ON
FASTEN BELTS switchOFF F/O
ANTI COLLISION light switchOFF F/O
FUEL PUMP switchesOFF F/O
CAUTION: Do not use the center tank fuel pumps with the flight deck unattended.
GALLEY power switchAs needed F/O
WING ANTI-ICE switchOFF F/O
ENGINE ANTI-ICE switches OFF F/O
Hydraulic panelSet F/O

ENGINE HYDRAULIC PUMPS switches - ON

ELECTRIC HYDRAULIC PUMPS switches - OFF	
GASPER FAN switchAs needed	F/O
Air conditioning PACK switchesSet	F/O
One air conditioning PACK switch - ON	
Other air conditioning PACK switch - OFF	
ISOLATION VALVE switchAUTO	F/O
Engine BLEED air switchesON	F/O
APU BLEED air switchON	F/O
Exterior lights switchesAs needed	F/O
Flight director mode selectorOFF	C, F/O
After the wheel chocks are in place:	
Parking brake - Release	С
APU switchAs needed	F/O
Flight deck door Unlock	F/O
Verify that the CAB DOOR UNLOCKED light (as installed)	is illuminated.
Oxygen regulatorsSet	C, F/O
Do the SHUTDOWN checklist.	F/O

# SECURE PROCEDURE

EMERGENCY EXIT LIGHTS switchOFF		F/O
WINDOW HEAT switchesOF	F	F/O
Air conditioning PACK switchesOFF	=	F/O
Call "SECURE CHECKLIST."	С	
Do the SECURE checklist.	F/O	

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#### SPECIAL THANKS TO

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