

DA42 TWINSTAR

QUICK REFERENCE HANDBOOK

REV 1.2.1

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WARNING!
This QRH is to be used for training purposes only within SATC/SFA. In case there is a discrepancy between the AFM/POH or this QRH, the AFM/POH will overrule the procedures, limitations, performance or systems described in this QRH.

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PPT

Revised to be more accurate to match reality.

MAN

Cruise speed >5000' = 145KTAS.

1 engine out approach speed & setting.

Introduction of Flap retraction speed schedule.

Steep turn speed @120KIAS. Procedure adapted.

HACL V-list items revised (height, note about seat adjustments removed)

Introduction of Approach to stall procedure.

Clean Stall recovery speeds.

NORM PROC

Clarification of callouts (config changes, Loc & GS alive)

Standardisation of cockpit flows items to match checklist & flows. Added procedures to test De-ice system, Electrical trim & Autopilot.

Overview of flows: completely revised to match Norm Proc.

Normal Checklist: revised to match Norm Proc & Diamond checklist V14.1.

EMER PROC

Definition of Non normal work method.

Clarification of Emergency checklists layout (Warnings, Cautions, Recall items).

Checklists layout adapted.

Oil Temp emergency checklist corrected according to POH.

L/R Engine fire/ failure Emergency checklist replaced by Engine failure/ fire and shutdown Emergency checklist.

Cabin Smoke and Fire above 10000' : Only applicable to aircraft equipped with oxygen system.

PAT

Speeds, Settings & Body attitudes updated for visual patterns

Introduction of stabilized approach concept & configuration changes.

LDG Checklist performed earlier on instrument approaches.

W&B PERF

BEW and moments updated.

Loadsheet: Usable fuel 50USG max.

Introduction of CRUISING TAS graph.

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PITCH POWER TABLE

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PITCH POWER

QRH DA42
REV 1.2 — 1.1

SPEED	CONF	ALT/ENGINES	POWER	BODY ATTITUDE (°)
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TAKEOFF				
VR 75 KIAS	FLAPS UP	2 engines	FULL	+9

CLIMB				
90 KIAS	CLEAN	2 engines	100%	+9
105 KIAS	CLEAN	2 engines	90%	+6
80 KIAS	CLEAN	2 engines	100%	+13
82 KIAS	CLEAN	1 engines	100%	+6

LEVEL				
65 KIAS	CLEAN	2 engines	40%	+10
105 KIAS	CLEAN	2 engines	50%	+4
120 KIAS	CLEAN	2 engines	60%	+1
<5.000FT 140 KIAS	CLEAN	2 engines	70%	+0
>5.000FT 145 KTAS	CLEAN	2 engines	70%-75%	+0
105 KIAS	CLEAN	1 engine	90%	+4
110 KIAS	APP FLAPS / GD	2 engines	65%	-1

DESCENT 500FPM				
130 KIAS	CLEAN	2 engines	50%	-2
120 KIAS	CLEAN	2 engines	40%	-1
105 KIAS	CLEAN	1 engine	75%	+1

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PITCH POWER

QRH DA42
REV 1.2 — 1.2

SPEED	CONF	ALT/ENGINES	POWER	BODY ATTITUDE (°)
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TRAFFIC PATTERN—2 ENGINES

105 KIAS	CLEAN	DOWNWIND	50%	+4
95 KIAS	APP FLAPS/GU	DOWNWIND	50%	-1
90 KIAS	APP FLAPS/GD	BASELEG	50%	-2
85 KIAS	LDG FLAPS/GD	FINAL	50%	-4

FLAPLESS PATTERN—2 ENGINES

105 KIAS	CLEAN	DOWNWIND	50%	+4
100 KIAS	CLEAN/GU	DOWNWIND	45%	+5
95 KIAS	CLEAN/GD	BASELEG	35%	-2
90 KIAS	CLEAN/GD	FINAL	35%	-1

TRAFFIC PATTERN—1 ENGINE

105 KIAS	CLEAN	DOWNWIND	90%	+4
95 KIAS	CLEAN/GD	BASELEG	70%	-2
95 KIAS	APP FLAPS/GD	FINAL	70%	-3
85 KIAS	LDG FLAPS/GD	SHORT FINAL	80%	-4

IFR APPROACH

100 KIAS	APP FLAPS/GD	2 engines	55%	-4
100 KIAS	APP FLAPS/GD	1 engine	80%	-4

MANEUVERS

PERF	W & B	LIM	PAT	EMER	NORM	MAN	PPT	TOC
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SPEEDS		
NORMAL CLIMB	90 KIAS	< 1.000 FT AGL
NORMAL CLIMB	105 KIAS	> 1.000 FT AGL
BEST RATE OF CLIMB	79 KIAS CLEAN	VY
BEST ANGLE OF CLIMB	79 KIAS CLEAN	VX
TO EXPEDITE CLIMB	85 KIAS	FULL POWER
CLIMB 1 ENGINE	82 KIAS	PITCH FOR THE BLUE LINE
NORMAL CRUISE	140 KIAS	<5 .000 FT MSL
NORMAL CRUISE	145 KTAS	> 5.000 FT MSL
CRUISE 1 ENGINE	105 KIAS	90—100%
VFR MANEUVERS	105 KIAS	50%
STEEP TURNS	120 KIAS	60%
MAX SPEED LIGHT TURBULENCE	155 KIAS (Vno)	
MAX SPEED ROUGH AIR	126–120 KIAS (Va)	
TO EXPEDITE DESCENT	(Gear Down) Maintain speed	1.000 FPM

SPEEDS				
IFR HOLDING 2 ENG	120 KIAS	60% POWER	CLEAN	FLAPS UP
IFR COURSE REVERSAL 2 ENG	120 KIAS	60% POWER	CLEAN	FLAPS UP
IFR APPROACH SPEED 2 ENG	100 KIAS	55% POWER	GEAR DOWN	FLAPS APP
IFR APPROACH SPEED 1 ENG	100 KIAS	80% POWER	GEAR DOWN	FLAPS APP

FLAPS SPEED SCHEDULE

RETRACTION		
FLAPS	AT SPEED	SELECT FLAPS
LDG	> 75 KIAS	APP
APP	> 85 KIAS	UP

SLOW FLIGHT

- DECELERATION

SPEED 105 KIAS
 POWER REDUCE TO 20%
 ALTITUDE..... MAINTAIN
 BAGRADUALLY INCREASE TO + 10°
 SPEED 65 KIAS
 POWER 40%

Trim

- ACCELERATION

POWER FULL
 ALTITUDE..... MAINTAIN
 BA.....GRADUALLY DECREASE TO +4°
 SPEED 105 KIAS
 POWER 50%

Trim

STEEP TURNS

SPEED 120 KIAS
 BANK ANGLE NORMAL ROLL RATE TO 45°
 BA INCREASE by 2° (BA +3°)
 POWER INCREASE +5%
 RUDDER AS REQUIRED TO CENTER SIDE SLIP

Do not trim

HEADING 15 ° BEFORE INITIAL HEADING
 BANK NORMAL ROLL RATE TO 0°
 BA.....DECREASE TO +1°
 POWER 60%

Control the roll to repeat exercise in opposite direction.

RUDDER AS REQUIRED TO CENTER SIDE SLIP

STALLS

- **BEFORE (EACH) STALL EXERCISE—VITAL ACTIONS**
 - H HEIGHT..... ABOVE 2300 FT AGL
 - A AREAOUT OF CONTROLLED AIRSPACE
 - NOT ABOVE TOWNS OR AIRFIELDS
 - C COCKPIT..... NO LARGE LOOSE ARTICLES
 - SEAT BELTS SECURED
 - ENGINE INSTRUMENTS IN LIMITS
 - L LOOKOUT..... 180° OR 2x90° CLEARING TURN(S)

FULL STALL RECOVERY TECHNIQUE

VALID FOR ALL STALL EXERCISES

- **AS THE NOSE STARTS TO DROP**
 - 1—BODY ATTITUDE..... SLIGHTLY BELOW THE HORIZON
 - 2—POWER FULL
- **IF A WING SHOULD DROP**
 - RUDDER OPPOSITE TO PREVENT YAW
 - SPEEDCHECK > 75 KIAS
 - AILERONSUSE TO LEVEL THE WINGS
- **WHEN FLYING SPEED HAS BEEN REGAINED**
 - 3—RECOVERY.....SEE BELOW ACCORDING TO THE TYPE OF STALL

APPROACH TO STALL RECOVERY TECHNIQUE

VALID FOR ALL STALL EXERCISES

- **AT THE 1st INDICATION OF THE STALL (BUFFET OR STALL WARNING)**
 - 1—BODY ATTITUDE..... MAINTAIN
 - 2—POWER FULL
- **WHEN SPEED INCREASING**
 - BODY ATTITUDE..... DECREASE TO MAINTAIN ALTITUDE
 - FLAPSRETRACT ON SCHEDULE

STALL (CLEAN)

0° BANK—64 KIAS

IDLE—FORWARD CG—MAX WEIGHT

- ENTRY

SPEED 105 KIAS
 POWERREDUCE TO IDLE
 ALTITUDE..... MAINTAIN
 TRIM DO NOT TRIM BELOW 75 KIAS

- RECOVERY

POWERFULL
 SPEED ACCELERATING THROUGH 75 KIAS
 BA.....ROTATE TO +10°
 SPEEDMAINTAIN 90 KIAS
 ALTITUDE..... CLIMB TO INITIAL ALTITUDE
 HEADINGTURN BACK TO INITIAL HEADING

LEVEL OFF AT INITIAL ALTITUDE

BA.....GRADUALLY DECREASE TO +4°
 SPEED 105 KIAS
 POWER 50% RPM

TAKEOFF STALL (POWER—ON / CLEAN)

NO AFM DATA ON EXACT STALL SPEED AVAILABLE

EXPECT ACTUAL STALL SPEED TO BE BELOW 50 KIAS (0° BANK)

- ENTRY

SPEED 105 KIAS

POWER 25%

WHEN PASSING THROUGH 75 KIAS:

BA RAISE TO +20°

POWER FULL POWER or 65% (UP TO INSTRUCTOR)

TRIM DO NOT TRIM BELOW 75 KIAS

- RECOVERY

POWER FULL

SPEED ACCELERATING THROUGH 75 KIAS

BA ROTATE TO +10°

SPEED MAINTAIN 90 KIAS

ALTITUDE BACK TO INITIAL ALTITUDE

HEADING TURN BACK TO INITIAL HEADING

LEVEL OFF AT INITIAL ALTITUDE

SPEED 105 KIAS

POWER 50%

Trim

LANDING STALL (POWER—OFF / FLAPS LDG)

0° BANK—57 KIAS

IDLE—FORWARD CG—MAX WEIGHT

- **ENTRY**

SPEED 105 KIAS
POWER 50%

USE TRAFFIC PATTERN SEQUENCE TO ARRIVE AT FINAL CONFIGURATION (LDG FLAPS—85KIAS)

POWER 40%
SPEED 85 KIAS
FLIGHT PATH ESTABLISH A STABILIZED DESCENT
TRIM TRIM FOR 85 KIAS
POWER IDLE TO SIMULATE A FLARE TO LAND

- **RECOVERY**

POWER FULL
FLAPS APPROACH
SPEED 75 KIAS
BA ROTATE TO +5°
ALTIMETER & VSI POSITIVE CLIMB

“POSITIVE CLIMB”

GEAR UP
SPEED 85 KIAS
FLAPS UP
SPEED MAINTAIN 90 KIAS
ALTITUDE CLIMB TO INITIAL ALTITUDE
HEADING TURN BACK TO INITIAL HEADING

LEVEL OFF AT INITIAL ALTITUDE

BA GRADUALLY DECREASE TO +4°
SPEED 105 KIAS
POWER 50%

Trim

FLAPLESS PATTERN
SEE ALSO CHAPTER 'PATTERNS'

- SPEEDS
ALL TARGET PATTERN SPEEDS ADD 5 KNOTS

- KEYPOINTS
 - BODY ATTITUDES WILL BE HIGHER THAN NORMAL (TRIM!!)
 - DOWNWIND EXTEND SLIGHTLY (1/4 MILE)
 - SLOPE FLY A NORMAL 3° SLOPE TO RWY
 - FLARE APPLY REDUCED FLARE—DON'T FLOAT
 - GO—AROUND “GO—AROUND, FLAPS UP”

GO—AROUND
“GO—AROUND, FLAPS!”

POWER FULL
 BA ROTATE TO +9°
 FLAPS APP (EXCEPT IF FLAPLESS APPROACH)
 ALTIMETER & VSI POSITIVE CLIMB

“POSITIVE CLIMB”

GEAR UP
 SPEED CHECK ABOVE 85 KIAS
 FLAPS UP
 SPEED MAINTAIN 90 KIAS
 ATC INFORM

“AFTER TAKEOFF CHECKLIST”

EFFECTS OF ASYMMETRICAL POWER

- EACH DEMO ONCE

- STRAIGHT AND LEVEL FLIGHT
 - SPEED..... 105 KIAS
 - LEFT OR RIGHT THROTTLEREDUCE TO IDLE
 - AIRPLANE REACTION & PERFORMANCE OBSERVE

- CLIMB
 - SPEED..... 90 KIAS
 - LEFT OR RIGHT THROTTLEREDUCE TO IDLE
 - AIRPLANE REACTION & PERFORMANCE OBSERVE

- DESCENT
 - SPEED..... 120 KIAS
 - LEFT OR RIGHT THROTTLEREDUCE TO IDLE
 - AIRPLANE REACTION & PERFORMANCE OBSERVE

- LEVEL TURN
 - SPEED..... 105 KIAS
 - OUTBOARD THROTTLEREDUCE TO IDLE
 - AIRPLANE REACTION & PERFORMANCE OBSERVE
 - INBOARD THROTTLEREDUCE TO IDLE
 - AIRPLANE REACTION & PERFORMANCE OBSERVE

- CLIMBING TURN
 - SPEED..... 90 KIAS
 - OUTBOARD THROTTLEREDUCE TO IDLE
 - AIRPLANE REACTION & PERFORMANCE OBSERVE

 - SPEED..... 90 KIAS
 - INBOARD THROTTLEREDUCE TO IDLE
 - AIRPLANE REACTION & PERFORMANCE OBSERVE

- DESCENDING TURN

SPEED 120 KIAS
 OUTBOARD THROTTLE REDUCE TO IDLE
 AIRPLANE REACTION & PERFORMANCE OBSERVE

SPEED 120 KIAS
 INBOARD THROTTLE..... REDUCE TO IDLE
 AIRPLANE REACTION & PERFORMANCE OBSERVE

- ONE ENGINE FLIGHT

BA +4°
 POWER GOOD ENGINE 100%
 SIMULATED FEATHER POWER (20% = ABOVE GEAR HORN)
 SET THROTTLE FOR 105 KIAS CLEAN (90—100%)

DEMONSTRATE: LEVEL FLIGHT-TURN-CLIMB-DESCENT

EMPHASIZE:

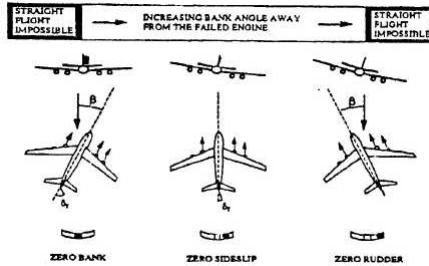
THROTTLE AND FOOT TOGETHER
 THROTTLE TRAVEL DOUBLES
 THE SIDE SLIP INDICATOR SHOULD BE +1/2 WIDTH OUT ON THE SIDE OF THE
 GOOD ENGINE
 BANK ANGLE 3-5° TOWARDS THE GOOD ENGINE

- Demonstrate:

1—Zero bank (bad)

2—Zero rudder (bad)

3—Zero sideslip (good)



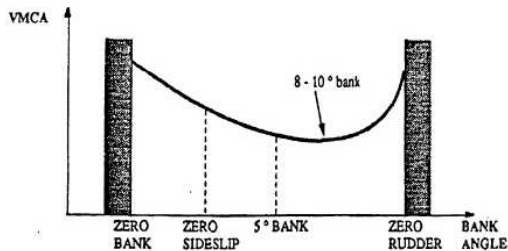
- Demonstrate:

1—Optimum performance = max performance climb at blue line speed

- Show that max climb rate is obtained with 2 to 3 degrees of bank into the good engine

2—Optimum control = Vmca demo

- Show that lowest VMCA is obtained with 8–10 degrees of bank into good engine



VMCA DEMONSTRATION

- BEFORE (EACH) DEMONSTRATION—VITAL ACTIONS
 H HEIGHT..... ABOVE 4000 FT AGL
 A AREAOUT OF CONTROLLED AIRSPACE
 NOT ABOVE TOWNS OR AIRFIELDS
 C COCKPIT.....NO LARGE LOOSE ARTICLES
 SEATS AND SEAT BELTS SECURES
 ENGINE INSTRUMENTS IN LIMITS
 L LOOKOUT..... 180° OR 2x90° CLEARING TURN(S)

- SEQUENCE
 SPEED 105 KIAS
 POWER FULL GOOD ENGINE-IDLE DEAD ENGINE
 BANK..... 5° INTO THE GOOD ENGINE
 BASLOWLY INCREASE (SPEED DECR 1KT/ SEC)
 RUDDER PRESSURE... GRADUALLY INCREASE TO MAINTAIN HEADING

WHEN ROLL OFF, STALL HORN OR LOSS OF DIRECTIONAL CONTROL OCCURS (RUDDER MAXIMUM):

POWERREDUCE ON GOOD ENGINE TO KEEP CONTROL
 RUDDER PRESSURE..... RELAX
 BA DECREASE

WHEN DIRECTIONAL CONTROL IS REGAINED:

POWER FULL GOOD ENGINE-IDLE DEAD ENGINE
 SPEED 82 KIAS
 BANK..... MAX 5° INTO THE GOOD ENGINE
 TURN TO ORIGINAL HEADING

NOTE AIRPLANE PERFORMANCE

REAL ENGINE SHUTDOWN & RESTART IN FLIGHT

- STAY WITHIN 10 NM OF A SUITABLE AIRPORT FOR THE DA42
- REMAIN AT OR ABOVE 4.000FT AGL
- PERFORM THE ‘ENGINE FAILURE/ FIRE AND SHUTDOWN CHECKLIST’ IN READ & DO
- WITH 1 PROPELLER FEATHERED: STRAIGHT AND LEVEL TURNS (BCAA REQUIREMENT!)
- PERFORM THE ‘ENGINE RESTART CHECKLIST’ (UNFEATHERING PROCEDURE) IN READ & DO
- RESTART BELOW 6.000 FT AGL—SPEED 80-120 KIAS.
- DO NOT SWITCH OFF THE ALTERNATORS AND FUEL SELECTORS (SIMULATE THESE ACTIONS WITHOUT ACTUALLY DOING THEM)

TOUCH & GO

- DURING THE TOUCH AND GO ROLL:
 FLAPS UP (BY THE INSTRUCTOR)
 - VERIFY WITH INSTRUCTOR PRIOR TO RETRACTING FLAPS
 “TAKEOFF, YOU HAVE CONTROL”

GO—AROUND 1 ENGINE

“GO—AROUND, FLAPS UP !”

POWER FULL

FLAPS UP

GROUND CONTACT UNLIKELY

GEAR UP

SPEED BLUE LINE (APPROX. +6°)

“AFTER TAKEOFF CHECKLIST”

SHORT FIELD TAKEOFF

APPROACH FLAPS

- TAKEOFF
 - BRAKES HOLD
 - POWER FULL
 - ENGINE INSTRUMENTS CHECK
 - BRAKES RELEASE
 - SPEED CHECK INCREASING—ROTATE AT 73 KIAS

- CLIMB
 - ALTIMETER & VSI POSITIVE CLIMB
 - “POSITIVE CLIMB”**
 - GEAR UP
 - BA +10°
 - SPEED MAINTAIN 79 KIAS

 - WHEN CLEAR OF OBSTACLES (100 FT ABOVE OBSTACLE)

 - BA LOWER TO +5°
 - SPEED ACCELERATE TO 85 KIAS
 - FLAPS UP
 - SPEED CLIMB AT 90 KIAS
 - “AFTER TAKEOFF CHECKLIST”**

SHORT FIELD LANDING

- FINAL TURN COMPLETED
 - FLAPS LDG
 - SPEED 80 KIAS

 - WHEN CLEAR OF OBSTACLES

 - BA AIM FOR THRESHOLD
 - POWER ADJUST FOR 80 KIAS
 - “SHORT FINAL CHECKLIST”**

- TOUCHDOWN
 - BRAKES APPLY (GRADUALLY)

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NORMAL PROC

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INTRODUCTION

BEFORE EVERY FLIGHT DEPARTING FROM HOME BASE

- 1 / COMPANY INFORMATION CHECK ON WEBSITE
(BULLETINS—WARNINGS—LATEST DOCUMENTS)
- 2 / WEATHER / NOTAMS / TFR'SCHECK
USE WEBSITE OR 1-800-WX-BRIEF BY PHONE
- 3 / ATIS OBTAIN THROUGH RADIO AT DISPATCH
- 4 / FLIGHT RELEASE FORMPRINT & COMPLETE
- 5 / AIRCRAFT TECHNICAL LOG.....CHECK SQUAWKS
- 6 / REQUIRED EQUIPMENT.....CHECK
- 7 / AIRCRAFT POUCH.....OBTAIN FROM DISPATCH
CHECK CONTENT (KEYS—FUEL CARD—LOGBOOK)
- 8 / I-M-S-A-F-E CHECKLISTCHECK
- 9 / MISSION BRIEFINGCHECK

NOTES

► A black triangle (sideways) printed in front of a checklist or flow item, indicates optional equipment and must be read as “if installed...”.

PROCEDURES FOR CHECKLISTS AND FLOWS

- Student in the left seat—Instructor in the right seat. Student will perform all flying duties, and scans and actions (flows), and all checklists, except the ones shown in the table as ‘PM’ – ‘Pilot Monitoring’. The reader will announce the start of any checklist by calling out its title, and will read all items out loud (with sufficient pause between the items for verification and/or answer), and will announce the checklist completed by announcing its title + ...“COMPLETED”.
- ‘PM’ – ‘Pilot Monitoring’ will be the instructor. For single-pilot operations, all items listed in the table below under ‘PM’ will have to be performed by the ‘PF’. All students must be proficient in operating the DA-42 in a single-pilot environment.
- All configuration changes must be announced aloud by the PF. E.g. “**Gear Up/Down, Flaps Up/ Approach/ Land**”.
- During IFR approach procedures the following items must be called out:
 - “Localiser Alive”
 - “Glideslope Alive”
 - “Needle Alive” (VOR/NDB/GPS approach)
 - “Outer Marker_____ft”

FLOW AND CHECKLIST HANDLING

Phase of Flight	Flow		Checklist	
	Performed by	Loud or Silent	Performed by	Loud or Silent
Before Start	PF	Loud	PM	Loud
After Start	PF	Loud	PM	Loud
Before Takeoff	PF	Loud	PM	Loud
Line Up	PM	Silent	N/A	N/A
After Takeoff	PF	Silent	PF (by heart)	Loud
Descent / Approach	PF	Silent	PF (by heart)	Loud
Landing	PF	Silent	PF (by heart)	Loud
After Landing	PM	Silent	PM	Silent
Shutdown	PF	Loud	PM	Loud

TOC

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MAN

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W & B

PERF

APPROACHING THE AIRCRAFT

- Through flight = same day, same airplane, same crew.
- Through flight walk around = fuel, oil and tires only.
- All cockpit flows must be read **out loud**.

NEARBY OBSTACLES CHECK
FLAP POSITION..... NOTE

The flap position should be noted before boarding the airplane.

SURFACES CLEAR

Check that all movable surfaces are clear and can be moved without hitting obstacles.

ANTENNAS CHECK

Check presence and condition of following antennas: COM, NAV, transponder, ELT.

PRELIMINARY INSPECTION

Be careful not to use the top of the instrument panel as a support when entering or exiting the aircraft. Use the dedicated handle behind the glareshield for support.

Verify that the canopy is clean and undamaged. Check for cracks and major scratches. Check locking mechanism. To avoid scratching the paint or damaging the wing, do not sit or put any equipment on the wing. Do not use the top of the instrument panel as a shelf for equipment at any time. Make sure not to touch the canopy with hands, equipment or charts.

IGNITION KEYCHECK OFF & KEY REMOVED
FRONT CANOPY & REAR DOOR CHECK
FIRE EXTINGUISHER CHECK SECURE/ CONDITION

UPPER SWITCH PANEL ALL SWITCHES OFF
ECU SWITCHES AUTO
ALTERNATOR SWITCHES ON

LOWER SWITCH PANEL ALL SWITCHES OFF

Check pitot heat off, both engine masters off, start key pulled out, electric master off and avionics master off.

GEAR SELECTOR DOWN

To prevent inadvertent gear retraction when the battery master switch is turned on and the squat switch located on the left main gear fails.

PARKING BRAKESET

To set the parking brake: set parking brake on, then pump the brake pedals to build up pressure in the brake cylinder.

ELECTRIC MASTER SWITCH ON

When switching on the electric master switch, the electrically driven hydraulic gear pump may activate for 5 to 20 seconds in order to restore system pressure. Should the pump operate continuously or intermittently, this indicates a malfunction in the landing gear system.

FUEL GAUGES CHECK QUANTITY
 FLAPS LDG
 ELECTRIC MASTER SWITCH OFF
 TRIMS NEUTRAL

Check full control travel of rudder trim and elevator trim, then, in preparation for the walk around, set in neutral position.

SUNSCREENS, PITOT COVER, STALL WARNING COVER, TIE DOWN, CHOCKS REMOVED

Put sunscreens (if available) in their dedicated bag. Put pitot cover in the flight gear bag. If the aircraft's own tie-down ropes and wheel clocks were used, put in the flight gear bag.

TOW BAR PROPERLY STOWED

REQUIRED EQUIPMENT ON BOARD

The following equipment must be on board:

- Aircraft pouch containing: aircraft flight time log book, emergency sick sacs, fuel card & keys.
- Bag containing sunscreens for the windows (if available) .
- Flight gear bag containing: fuel tester, flip-up training glasses, flash light, wheel chocks, pitot cover, 3 tie-down ropes, first aid kit.
- Fire extinguisher.
- Rescue hammer.
- EASA/FAA approved airplane flight manual (POH).
- JAA/FAA required documents: certificate of airworthiness, airplane registration.
- Alternate fuel measuring device
- G1000 User manual

FRONT BAGGAGE DOORSSECURE

The door must be locked with the key before the flight to prevent opening in flight.

WALK AROUND

Walkaround is clockwise around the aircraft.

LEFT WING

STEP CHECK
NACELLE UNDERSIDE FUEL COOLER AIR IN & OUTLETCHECK/CLEAR

WING FLAPS CHECK
FLAP AND LINKAGE CHECK
FLAP HINGES AND SAFETY PIN CHECK

AILERON CHECK
AILERON AND LINKAGE CHECK
FOREIGN OBJECTS IN AILERON PADDLE..... CHECK
STATIC DISCHARGERS CHECK
WING TIP..... CHECK
POSITION LIGHT, STROBE LIGHT (ACL)..... CHECK
TIE-DOWNCHECK/CLEAR
MAIN FUEL TANK FILLERCHECK/CLOSE
PITOT PROBE..... CHECK

Check pitot probe is clean, orifices clear, cover removed and no deformation.

STALL WARNING DEVICE..... CHECK
TANK AIR OUTLET..... CHECK
WING SURFACE..... CHECK
TANK DRAIN/TANK AIR INLETCHECK/DRAIN

Check for water and sediment (drain until no water or sediment comes out).

OPENINGS ON LOWER SURFACE..... CHECK

Check for foreign objects and for traces of fuel (if tank is full, fuel may spill over through the tank vent).

LEFT ENGINE NACELLE

3 AIR INLETS/2 AIR OUTLETS CLEAR
ENGINE OIL LEVELCHECK

Check oil level through inspection hole in upper cowling. Normal oil quantity is 5.5 quarts. Minimum is 4.8 quarts, maximum is 6.3 quarts. Do not refill when oil quantity is above 5.5 quarts. When oil quantity is at 5 quarts, add 1 full oil can (1quart). Maximum oil consumption is 0.1 quart/hour.

GEARBOX OIL LEVELCHECK

Check oil level through inspection hole in lower cowling.

COWLINGCHECK

PROPELLER & SPINNERCHECK

Check no detrimental nicks, cracks or dents in propeller blades, and no traces of oil (leak in the CSU). Check blades no excessive play in CSU.

WARNING
Even in the OFF position, an engine may fire. Always treat a propeller as potentially dangerous. Do not take position underneath a propeller at any time. Never push/pull the propeller to move the aircraft.
Never move the propeller by hand.

EXHAUSTCHECK

WARNING
The exhaust can cause burns when hot.

VENTING PIPECHECK FOR BLOCKAGE

GASCOLATOR/AIR INLETDRAIN / CLEAR

Check for water and sediment (drain until no water or sediment comes out). Do not pour the drained fuel back into the tank.

NACELLE UNDERSIDE..... CHECK

Check for excessive contamination particularly by oil, fuel and other fluids.

- ▶ AUXILIARY TANK FILLERCHECK/CLOSE
- ▶ AUXILIARY TANK VENT OUTLET..... CHECK
- ▶ AUXILIARY TANK DRAINCHECK/DRAIN

Check for water and sediment (drain until no water or sediment comes out).

LEFT MAIN LANDING GEAR

LANDING GEAR STRUT & LOCK..... CHECK

Check visually and verify strut height is sufficient. Typical visible length of bare piston is at least 4cm/1.6".

DOWN & UPLOCK SWITCHES (3X)..... CHECK
WEAR, TREAD DEPTH OF TIRE..... CHECK

Tires should be taken out of service when they have one or more flat spots. Generally, a single flat spot or skid burn does not expose the carcass body and the tire may remain in service, unless severe unbalance is reported by the crew. Small cuts are acceptable, if they do not protrude into the tire carcass. Cuts in the side wall are not acceptable. Shallow chevron-shaped cuts across the tread of a tire pose no problem; they are caused by landing on a grooved concrete runway. As long as the tread does not wear down into the body plies of the carcass, the basic strength of the tire is not affected. To provide traction during wet runway operation, operators should replace their tires when the tread reaches $1 / 32$ inch = 0.79 mm.

WHEEL & BRAKES/HYDRAULIC BRAKE LINE..... CHECK

When checking the brakes: verify that there is even wear on the disc, no scratches, no grease, and no leaks near the brake line.

- SLIP MARKSCHECK
- LANDING GEAR DOORSCHECK
- CHOCKS REMOVE

FRONT FUSELAGE AND NOSE LANDING GEAR

- FRONT CANOPY, LEFT & RIGHT SIDECHECK
- LEFT AND RIGHT FRONT BAGGAGE DOOR CHECK/CLOSE & LOCK
- NOSE LANDING GEAR STRUT, LOCK & CENTERING DEVICECHECK

Check visually and verify strut height is sufficient. Typical visible length of bare piston is at least 5cm / 5.9".

- DOWN & UPLOCK SWITCHESCHECK
- WEAR, TREAD & DEPTH OF TIRECHECK

Tires should be taken out of service when they have one or more flat spots. Generally, a single flat spot or skid burn does not expose the carcass body and the tire may remain in service, unless severe unbalance is reported by the crew. Small cuts are acceptable, if they do not protrude into the tire carcass. Cuts in the side wall are not acceptable. Shallow chevron-shaped cuts across the tread of a tire pose no problem; they are caused by landing on a grooved concrete runway. As long as the tread does not wear down into the body plies of the carcass, the basic strength of the tire is not affected. To provide traction during wet runway operation, operators should replace their tires when the tread reaches $1 / 32$ inch = 0.79 mm.

- SLIP MARKSCHECK
- CHOCKS REMOVE
- LANDING GEAR DOOR & LINKAGECHECK
- EPU CONNECTOR.....CHECK
- TOW BAR REMOVE

RIGHT MAIN LANDING GEAR

LANDING GEAR STRUT & LOCK..... CHECK

Check visually and verify strut height is sufficient. Typical visible length of bare piston is at least 4cm/1.6".

DOWN & UPLOCK SWITCHES (3X)..... CHECK
WEAR, TREAD DEPTH OF TIRE..... CHECK

Tires should be taken out of service when they have one or more flat spots. Generally, a single flat spot or skid burn does not expose the carcass body and the tire may remain in service, unless severe unbalance is reported by the crew. Small cuts are acceptable, if they do not protrude into the tire carcass. Cuts in the side wall are not acceptable. Shallow chevron-shaped cuts across the tread of a tire pose no problem; they are caused by landing on a grooved concrete runway. As long as the tread does not wear down into the body plies of the carcass, the basic strength of the tire is not affected. To provide traction during wet runway operation, operators should replace their tires when the tread reaches $1 / 32$ inch = 0.79 mm.

WHEEL & BRAKES/HYDRAULIC BRAKE LINE..... CHECK

When checking the brakes: verify that there is even wear on the disc, no scratches, no grease, and no leaks near the brake line.

SLIP MARKS..... CHECK
LANDING GEAR DOORS CHECK
CHOCKS..... REMOVE

RIGHT ENGINE NACELLE

3 AIR INLETS/2 AIR OUTLETS..... CLEAR
ENGINE OIL LEVELCHECK/CLEAR

Check oil level through inspection hole in upper cowling. Normal oil quantity is 5.5 quarts. Minimum is 4.8 quarts, maximum is 6.3 quarts. Do not refill when oil quantity is above 5.5 quart. When oil quantity is at 5 quarts, add 1 full oil can (1 quart). Maximum oil consumption is 0.1 quart/hour.

GEARBOX OIL LEVEL.....CHECK

Check oil level through inspection hole in lower cowling.

COWLING.....CHECK

PROPELLER & SPINNERCHECK

Check no detrimental nicks, cracks or dents in propeller blades, and no traces of oil (leak in the CSU). Check blades no excessive play in CSU.

WARNING
Even in the OFF position, an engine may fire. Always treat a propeller as potentially dangerous. Do not take position underneath a propeller at any time. Never push/pull the propeller to move the aircraft. Never move the propeller by hand.

EXHAUSTCHECK

WARNING
The exhaust can cause burns when hot.

VENTING PIPECHECK FOR BLOCKAGE

GASCOLATOR/AIR INLET.....DRAIN / CLEAR

Check for water and sediment (drain until no water or sediment comes out). Do not pour the drained fuel back into the tank.

NACELLE UNDERSIDECHECK

Check for excessive contamination particularly by oil, fuel and other fluids.

- ▶ AUXILIARY TANK FILLER..... CHECK/CLOSE
- ▶ AUXILIARY TANK VENT OUTLETCHECK
- ▶ AUXILIARY TANK DRAIN..... CHECK/DRAIN

Check for water and sediment (drain until no water or sediment comes out).

RIGHT WING

CABIN VENT AIR INLETCHECK/CLEAR
TANK DRAIN/TANK AIR INLETCHECK/DRAIN

Check for water and sediment (drain until no water or sediment comes out).

TANK AIR OUTLET CHECK
OPENINGS ON LOWER SURFACE CHECK

Check for foreign objects and for traces of fuel (if tank is full, fuel may spill over through the tank vent).

ENTIRE WING SURFACE CHECK
MAIN FUEL TANK FILLERCHECK/CLOSE
WING TIP..... CHECK
STATIC DISCHARGERS CHECK
POSITION LIGHT, STROBE LIGHT (ACL)..... CHECK
TIE-DOWNCHECK/CLEAR
AILERON & LINKAGE..... CHECK
FOREIGN OBJECTS IN AILERON PADDLE..... CHECK
FLAP AND LINKAGE CHECK
FLAP HINGES & SAFETY PIN CHECK
NACELLE UNDERSIDE FUEL COOLER AIR IN- & OUTLET ..CHECK/CLEAR
STEP CHECK

FUSELAGE, RIGHT SIDE, UNDERSIDE

AFT CANOPY CHECK
FUSELAGE SKIN CHECK
ANTENNAS UPPER SIDE..... CHECK
STATIC PORT CHECK

EMPANNAGE

- STABILIZERS & CONTROL SURFACES, ELEVATOR TIPS.....CHECK
- HINGESCHECK
- ELEVATOR TRIM TAB.....CHECK
- RUDDER TRIM TAB.....CHECK
- TIE-DOWN CHECK/CLEAR
- TAIL SKID AND LOWER FIN.....CHECK
- STATIC DISCHARGERS.....CHECK

FUSELAGE, LEFT SIDE, UNDERSIDE

- STATIC PORTCHECK
- FUSELAGE SKINCHECK
- FUSELAGE UNDERSIDE.....CHECK

Check for contamination with hydraulic fluid.

- ANTENNAS UNDERSIDECHECK
- AFT CANOPYCHECK

MISCELLANEOUS

BATTERY MASTER SWITCH.....ON

For night flight only:

INTERIOR LIGHTING ON AND CHECK
 EXTERIOR LIGHTING ON AND CHECK
 ALL LIGHTING SWITCHESOFF

For IMC flight only:

PITOT HEATON
 PITOT HEAT ANNUNCIATOR..... EXTINGUISHED
 PITOT TUBE.....CHECK WARM
 PITOT HEATOFF

For flight in known icing conditions only:

DEICE FLUID..... CHECK QUANTITY

The minimum de-icing fluid quantity for dispatch is 22 liters (5.8 US gal) or indication of ¾ full on G1000.

This minimum allows at least 90 minutes of ice protection with NORM selected. The pilot must ensure adequate fluid quantity before each flight.

Maximum system operating times with full de-icing fluid:

NORM mode2 hrs. 30 min.
 HIGH mode..... 1 hr
 MAX mode.....30 min

Do not operate the main pumps with an empty de-icing fluid tank. If priming is required use the windshield de-icing.

ICE LIGHT..... ON
 ICE LIGHTS CHECK
 ICE LIGHT.....OFF

BATTERY MASTER SWITCH.....OFF

COCKPIT PREPARATION

Before entering the cockpit check Towbar removed, Nose baggage doors closed and locked and Rear door closed and secured.

RUDDER PEDALSADJUSTED & LOCKED

The rudder pedals may only be adjusted on the ground. The pedals are unlocked by pulling the black handle which is located behind the rear attachment. Forward adjustment: while keeping the handle pulled, push the pedals forward with your feet to the desired position. Release the handle and allow the pedals to lock in place. Rearward adjustment: using the unlocking handle, pull the pedals back to the desired position. Release the handle and allow the pedals to lock in place. Verify that the pedals are locked by applying pressure to try to move them.

The instructor should advise the student of the correct position. The rudder pedals must be adjusted as far as possible with full deflection of the rudder still possible. This will enable more force to be applied on the pedal from the lower back and leg, and more accurate inputs. If required a cushion can be put behind the back,

FLIGHT CONTROLSPROPER OPERATION

Check the flight controls for proper operation: **“Stick left, left aileron up, right aileron down, stick right, right aileron up, left aileron down. Stick aft, elevator up, stick forward, elevator down. Left pedal, right pedal.”**

Aileron and elevator deflections can be visually checked from the pilot’s seat. Do not push the rudder pedals too hard against system resistance.

SEATBELTS.....LOCKED

Use of all available seat belts and/or harnesses per seat is mandatory. Seat belts of empty seats will be fastened and crossed over the seat bottom to prevent control interference or passenger injury during flight in turbulent air. Fasten belts before closing canopy. Do not unlock seat belts during flight or taxi at any time.

REAR DOOR CLOSED & SECURED
FRONT CANOPY POSITION 1 OR 2

When operating the canopy, pilots/operators must ensure that there are no obstructions between the canopy and the mating frame, for example seat belts, clothing, etc. When operating the locking handle, DO NOT apply undue force. A slight downward pressure on the canopy may be required to ease the handle operation.

FRONT BAGGAGE DOORS CLOSED

Visually check that the front baggage doors are closed.

AVIONICS MASTER SWITCH OFF
ELECTRIC MASTER SWITCH ON
ENGINE MASTER SWITCHES OFF

When switching on the electric master switch, the electrically driven hydraulic gear pump may activate for 5 to 20 seconds in order to restore system pressure. Should the pump operate continuously or intermittently, this indicates a malfunction in the landing gear system.

IGNITION KEY READY & OFF
PITOT HEAT OFF
ALTERNATE STATIC OFF
ALTERNATORS CHECK ON
ECU SWAP CHECK AUTOMATIC

INSTRUMENT PANEL

DIMMER CONTROL SET

Use the rotary buttons to adjust instrument lighting and flood light.
Day flight: OFF. Night flight: as required (avoid setting the lights too bright).

LIGHT SWITCHES OFF
EMERGENCY SWITCH CHECK OFF & GUARDED

STANDBY INSTRUMENTS CHECKED
MAGNETIC COMPASS CHECK

Check magnetic compass for normal reading, no bubbles in the fluid and no leaks. Deviation chart present.

DEVIATION CARD CHECK
ELT ARMED

► DE-ICING SYSTEM CHECK

ANNUN-TEST ON

The ANNUN-TEST mode activates the DEICE LVL LO caution immediately if the de-ice fluid quantity is low (below 10 liters) and the DEIC PRES LO caution after 120 seconds.

DEICE PRES HI VERIFY NOT ILLUMINATED
DE-ICE HIGH
DEIC PRES LO EXTINGUISHES (AFTER 20S)

If the ambient temperature is above 10 °C (50 °F) the system pressure may not get high enough to disable the DEIC PRES LOW warning.

DE-ICE OFF
ANNUN-TEST OFF

CIRCUIT BREAKERS IN

If a circuit breaker needs resetting, check why it has tripped or been pulled.

G1000 ACKNOWLEDGE

Wait until power-up completed. Press ENT on MFD to acknowledge. Note: the engine instruments are not available on the MFD until acknowledged. Push the DISPLAY BACKUP button on the audio panel to activate the G1000 reversionary (backup) mode. Push the SYSTEM softkey on the MFD to activate the engine/system page.

FUEL TEMPERATURE CHECK

Limitation:

Diesel Fuel: Engine start not below -5°C, Take Off not below +5°C.

JET A1 fuel: Engine start and Take Off not below -30°C.

Make sure which fuel grade is being used (see Section 7.9.5 of the AFM). If it is not possible to determine the fuel grade, the diesel fuel temperature limitations must be observed.

FUEL QUANTITY CHECK

FUEL CALCULATOR..... SET

TOTAL TIME IN SERVICE NOTE

Write down the G1000 total time in service.

G1000 BACKUP OR REV. MODE

Set Engine page on PFD and System page on MFD.

GEAR WARNING/FIRE DETECTOR CHECK

The gear warning system is combined with the engine fire detection system, push the gear warning/fire detector test button and verify you hear the aural alert and observe the warning on the PFD screen. If the aural alert or the warning does not appear, maintenance action is required.

MANUAL GEAR EXTENSION HANDLE CHECK PUSHED IN

ALTERNATE AIR..... CHECK CLOSED

CENTER CONSOLE

FLAPS.....CHECK LDG

RUDDER TRIMCHECK & SET

PARKING BRAKE SET

DEFROSTER SWITCHOFF

HEATER SWITCHOFF

VARIABLE ELEVATOR BACKSTOP CHECK

Pull the stick fully aft. While holding the stick at the backstop, check the flaps in the landing position (LDG) and set both power levers MAX. The stick must move slightly forward during power lever forward movement. Set power levers to IDLE. The stick must regain full movement during power lever rearward movement. Select flaps UP. For details: see Chapter 7 of the AFM.

WARNING

The proper function of the variable elevator backstop is required for the safety of the flight, as else handling qualities during power-on stalls are degraded significantly.

FLAPS	UP
FUEL SELECTORS	ON
ELEVATOR TRIM	SET FOR TAKEOFF
THROTTLE FRICTION	ADJUST
AVIONICS MASTER SWITCH	ON
ATIS	COPY
VFR OR IFR CLEARANCE	COPY
AVIONICS MASTER SWITCH	OFF
BRIEFING	PERFORM

Briefing must contain:

- Pilot flying
- Type of takeoff (normal or short field) and % load.
- Vr and Vclimb
- Engine failure procedure
- VFR and IFR departure procedure (routing)
- Initial climb instructions +safety altitude (VFR or IFR) (altitude)

Example:

“I fly normal takeoff, 100%, flaps up, Vr 75, Vclimb 90.
Engine failure below 85 knots, abort on the remaining runway and advise ATC.
Engine failure above 85 knots and above obstacles, fly the blue line, brief the escape route (ER).
E.g. VMC: visual pattern or IMC: ER to MSA then return for IFR approach.

VFR departure procedure. E.g. right crosswind departure, climb to 3500ft, MSA 4500ft” or,
IFR departure procedure. E.g. at 2000’ turn R hdg 120, intercept IWA R-030.
Questions?

“BEFORE START CHECKLIST”

ENGINE START

Starting with the external power source: perform procedure in read –and do with DA42 AFM (Abnormal Operating Procedures 4B.7).

CAUTION

If diesel fuel or a blend of diesel/Jet A-1 is used, or if the fuel grade is unknown, the engine must not be started if the fuel temperature indication prior to operation is below -5°C (+23°F).

STROBE LIGHTS (DAY) OR POSITION LIGHTS (NIGHT) ON
ENGINE MASTER SWITCH ON

It is recommended to start the LH engine (pilot side) first. If required by operational reasons, the RH engine can also be started first.

ENGINE GLOW CHECK ON

Verify the L/R engine glow annunciation is displayed on the PFD.

ENGINE INDICATIONS CHECK

Check that the engine indications are being powered (Red X disappear).

ENGINE GLOW CHECK OFF
PROPELLER AREA..... CLEAR

Check the area around the airplane. Open the window momentarily and shout “Props clear”.

BRAKES HOLD
IGNITION KEY START

Turn the start key left or right as required to start the appropriate engine, and release when the engine has started. Do not overheat the starter motor. Do not operate the starter motor for more than 10 seconds. After 1 start cycle, apply a cooling period of 20 seconds. After 6 consecutive start cycles, apply a cooling period of 30 minutes.

STARTER ANNUNCIATION OFF

If the ‘L/R STARTER’ annunciation does not extinguish after the engine has started and the START KEY has been released, set the ENGINE MASTER OFF.

RPMCHECK
OIL PRESSURECHECK

If the oil pressure has not moved from the red range within 3 seconds after starting, set the ENGINE MASTER OFF. When starting a cold engine, the oil pressure can be as high as 6.5 bar for a maximum of 20 seconds.

ANNUNCIATIONS/ENGINE/SYSTEM PAGE.....CHECK

Check volts and AMP load. Set idle power and verify the RPM is 900 +/- 20. Leave power at idle for 2 minutes. Thereafter, set 1400 RPM until oil temperature above 50°C and coolant temperature above 60°C.

REPEAT FOR OTHER ENGINE

CAUTION

With one engine running, before switching on the engine master for the second engine, check engine glow for the running engine is extinguished. Failure to do so, may result in engine glow being operated on 2 engines simultaneously, which will overload the electrical system and will subsequently require a maintenance intervention.

AVIONICS MASTER SWITCH..... ON

“AFTER START CHECKLIST”

BEFORE TAXI

PITOT HEAT CHECK

Switch pitot heat ON. Verify the amber PITOT HT OFF and STALL HT OFF annunciations disappear from the PFD. Note an increase in AMP output. Switch pitot heat OFF.

G1000 SETUP COMPLETE

I – F – R – B

- Initialize: Aux Page 4
- Check units (e.g. hPa or mbar)
- Check database: WGS 84
- Check CDI Selected AUTO / ILS CDI capture MANUAL
- Flightplan: go to FPL
- Check Dep AD correct (if not enter Dep AD)
- Load DP (softkey)
- Enter waypoints
- Load Apr (press softkey then enter AD of arrival)
- Radios :
- Tune : fcy
- Identify: if possible on ground
- Set: Nav 1: HSI /Nav 2: BRG/ ADF: BRG/ DME: NAV1 /2.

- Bugs (speed bugs, QNH 3x, Altitude bugs 2x)

Use the VFR and IFR (real or simulated) clearance (each item chronologically) as a guideline to set radios and nav aids. COM1: ground frequency. COM1 will be the only radio used for ATC communications during the whole flight. COM2 will be the radio dedicated to ATIS and company communications during the whole flight. NAV1: first VOR required by clearance, set first radial on HSI. NAV2: second VOR required by clearance, set BRG 2 (optional). NAV1 and NAV2 will be used as follows: at all times VOR (or LOC) in use on NAV1, next VOR on NAV2, DME and ADF: as required by clearance.

At least 1 frequency must be set to enable a quick return to the departure airport in case of an emergency.

AUTOPILOTTEST

1. Engage AP (feel resistance on stick)
2. Set hdg bug on top
3. Select HDG/ALT
4. Turn HDG bug (left/right) - monitor stick moving (to the left/right)
5. Press CWS button and check free movement of stick
6. Release CWS button, stick goes to the left again
7. Disengage AP by using red disconnect button

ELECTRIC TRIMTEST

1. Push both trim split buttons FWD, monitor elev trim moving FWD,
2. Press red button on stick, check elev trim movement stops,
3. Repeat steps 1 & 2 in AFT direction,
4. Move left split trim button FWD and AFT, check elev trim not moving,
5. Repeat step 4 with right split trim button,

ECU..... TEST

The following test sequence will be executed for both engines simultaneously. The whole test procedure must be completed without any error. If any error occurs, do not fly, even if the engine appears to run smoothly after the test procedure.

ECU test procedure:

Set POWER LEVER IDLE. Press and hold the ECU TEST BUTTON(S) and observe following sequence:

- ECU A/B FAIL LIGHTS..... ON
- PROPELLER RPM..... INCREASE
- ECU A/B FAIL LIGHTS..... OFF
- ECU B FAIL LIGHT ON
- PROPELLER RPM..... DECREASE/INCREASE
- ECU B FAIL LIGHT OFF
- ECU A FAIL LIGHT ON
- PROPELLER RPM..... DECREASE/INCREASE
- ECU A FAIL LIGHT OFF
- PROPELLER RPM..... DECREASE TO IDLE

Test sequence completed. Release the ECU TEST BUTTON(S).

CAUTION
If the 'L/R ECU A/B FAIL' do not illuminate and extinguish during the test sequence, there is a malfunction in the engine control system.

ECU SWAP CHECK

Set ECU swap to ECU B. A light shake of the engine may occur, but check that other than this, the engine keeps running without a change. Set ECU swap back to AUTOMATIC.

CAUTION
Running the engine with the ECU swap on ECU B, other than for this test or in an emergency is prohibited. The engines control system redundancy is only given with the ECU swap set to AUTOMATIC.

“BEFORE TAXI CHECKLIST”

TAXI

TAXI LIGHT ON
 FUEL SELECTORS CROSSFEED

Select both FUEL SELECTOR to CROSSFEED for 30 seconds.

AREA FREE

Check the area around the airplane. Call out “Left is free, free right?”
 Instructor will reply “Right is free”. In case of solo flights, the call
 out will be “Left is free, right is free” after having verified both sides.
 Then release the parking brake.

BRAKES CHECK

Perform brake check immediately after the aircraft starts rolling.
 Apply light and even pressure on both pedals. It is not necessary to
 bring the aircraft to a complete stop. As soon as it becomes appar-
 ent that normal brake pressure is available, release the brakes
 again. Ask the pilot in the right seat to perform a brake check as
 follows “Brakes check, you have controls, check your brakes”. The
 right seat pilot will reply “I have controls” and perform the brake
 check, then he will announce “Brakes, checked you have controls”.
 Pilot flying (left seat pilot) will reply “I have controls” and continue
 taxi.

Taxi turns can be made using rudder pedal motion only. Brakes are
 only needed to reduce the taxi speed or when maneuvering in tight
 spaces. On the ground and in flight, the pilot’s feet should be in an
 almost horizontal position—heels on the floor, toes on the lower
 part of the rudder pedals—sliding the feet up on the rudder pedals
 only when required to apply brakes.

Differential power (more power on the engine on the outside of the
 turn, less on the inside engine) may be used during turns on the
 ground.

FLIGHT INSTRUMENTS CHECK

In a left turn: "Turning left (=> check turn coordinator going left), skidding right (=> check side slip going right), HDG and compass decreasing (=> check numbers on DG and magnetic compass decreasing), horizon level (=> check attitude indicator level)".

In a right turn: "Turning right, skidding left, HDG and compass increasing, horizon level".

The checks need to be done only once, in a left turn or a right turn, outside congested area.

FUEL SELECTORS ON

BEFORE TAKEOFF

CONTROLS FREE

CIRCUIT BREAKERS IN

FLAPS AS REQUIRED

Normal takeoff FLAPS UP (CRUISE), short field takeoff FLAPS T/O.

G1000 REV. MODE

Set Engine page on PFD and System page on MFD.

TRIMS SET

FUEL SELECTORS ON

SHORT BRIEFING PERFORM

The short briefing will include:

- Type of takeoff (normal or short or soft)
- Initial heading and altitude
- The instructor answers "Confirmed" or amends as necessary.

Example VFR:

"Normal takeoff, heading 120°, 3.500FT, confirmed? – "Confirmed".

FRONT CANOPY CLOSED & LOCKED

DOOR WARNING CHECK OFF

“BEFORE TAKEOFF CHECKLIST”

LINE-UP

When ATC clearance (if required) received and acknowledged to line up, FIRST release the parking brake and start to roll to the runway, THEN perform the following actions (while taxiing). Fast taxi turns immediately prior to takeoff should be avoided to prevent unporting fuel feed lines. On request of PF **“Line-up items”**, the starred * items may be performed by the PM.

APPROACHFREE

Visually clear the final approach area before entering the runway and call out: **“Approach free”**.

- *LANDING LIGHT ON
- *TAXI LIGHT OFF
- *STROBE LIGHTS ON
- ALTIMETERCHECK (3X)

Read threshold elevation from Jeppesen or other airport chart and call out: **“Threshold elevation ____feet”**. This may be done while taxiing out of the run-up area. Altimeter reading should not differ more than 60 feet from threshold elevation at sea level and 80 feet at 5000 feet pressure altitude.

COMPASS SYSTEM..... RUNWAY HEADING

This is a runway + compass system check. Read runway heading from chart, takeoff data (QFU) and call out: **“Runway heading ____degrees”**. Check HSI and magnetic compass. Maximum difference allowed between published runway heading and any compass system is 6°. If a large discrepancy exists, do not take off, but investigate the cause first (wrong runway, compass system malfunction).

*PITOT HEATAS REQUIRED

Switch pitot heat ON. Verify the amber PITOT HT OFF and STALL HT OFF annunciations disappear from the PFD.

TAKE-OFF

Before applying engine power for takeoff, top the timer and call out: **“Takeoff, I have control”**. The instructor will confirm: **“You have control”**.

When the aircraft is lined up, perform the available takeoff power check.

BRAKES HOLD
POWER LEVERS FULLY FORWARD
PARAMETERS CHECK

Check:

- ANNUNCIATIONS CHECK
- RPM STABILIZES 2240 TO 2300 RPM
- LOAD STABILIZES 90 TO 100%

Call out: **“Power checked”**.

Note:

1. Under high temperature/high altitude conditions, load indications below 90% are possible. If the engine does not stabilize at the target RPM of 2240 to 2300 RPM, do not continue takeoff.
2. An engine is warm for takeoff when the throttle can be opened fully without engine faltering (backfiring or skipping) and without a reduction in engine oil pressure. It is important to check takeoff power early in the takeoff run or power application. Any sign of rough engine operation or sluggish acceleration is good cause for discontinuing the takeoff.
3. Takeoffs are made with full throttle.

When speed tape comes alive. Call out: **“Speed checked”**

At rotation speed: **“75 knots rotate”**

AFTER TAKEOFF

“POSITIVE CLIMB”

GEAR UP

Apply brakes to stop the spinning of the main wheels before entering the wheel well, then select lever up. Retract the gear when no more runway is available for an emergency re-landing.

FLAPS UP

After gear is up and at a safe altitude and speed.

Perform following actions at 1.000 FT AGL.

ENGINE INSTRUMENTSCHECK
LANDING LIGHTAS REQUIRED
TRANSPONDER CODE & ALT
MAP PAGE..... SWITCH

When out of the traffic pattern and congested area, switch off the landing lights. Do not switch off the landing lights when remaining in the traffic pattern or transiting controlled airspace.

“AFTER TAKEOFF CHECKLIST”

CRUISE

Cruise checklist should be performed every 15 minutes.

FUEL STATUSCHECK
ENGINE INDICATIONS / SYSTEM PAGECHECK

DESCENT-APPROACH

When the altimeter setting for the destination airport is set:

- LANDING LIGHT ON
- ALTIMETERS SET (3X)

Set local altimeter setting for the destination airport. Complete answer to this checklist item is: “____(altimeter setting) set”.

- ANNUNCIATIONS / ENGINE / SYSTEM PAGE CHECKED
- GEAR WARNING HORN CHECKED
- FUEL SELECTORS ON
- DE-ICING SYSTEM CONSIDER

“DESCENT-APPROACH CHECKLIST”

LANDING

Turning base:

- LANDING GEAR DOWN-3 GREENS
- BRAKES CHECKED

After checking the brakes, make sure to slide your feet back to the normal position (heels on floor, toes on lower part of the pedals).

- PARKING BRAKE OFF
- FLAPS AS REQUIRED

“LANDING CHECKLIST”

SHORT FINAL

- LANDING GEAR DOWN-3 GREENS
- TOUCHDOWN

Maintain desired approach flight path with BA. Maintain desired airspeed with power. Reduce power to idle during the flare before the main wheels touch the runway. This gives the gear warning horn a chance to blow if the gear is not locked down. After ground contact, hold the nose wheel off as long as possible. Braking—if needed—is most effective when back pressure is applied to the control wheel, putting most of the aircraft weight on the main wheels. When the headwind component exceeds 15 knots, or when the crosswind component exceeds 10 knots, as well as in gusty wind conditions, the approach will be flown at a slightly higher than normal speed (+5 KIAS) with APPROACH FLAPS.

AFTER LANDING

Runway vacated, aircraft stopped behind hold short line. On request of PF “**After landing items**”, the starred * items may be performed by the PM.

- *LANDING LIGHT OFF
- *TAXI LIGHT ON
- *PITOT HEAT OFF
- TIMER START-ENGINE IDLE

Engines have to run at idle power for 2 minutes before shut down.
Taxi with idle power if possible.

- TRANSPONDER CODE & GROUND
- *FLAPS UP

“AFTER LANDING CHECKLIST “

If the starred items were performed by the PM, he/she will read the AFTER LANDING checklist silently and announce “**AFTER LANDING CHECKLIST COMPLETED**” to the PF.

SHUTDOWN

- TAXI LIGHT OFF

TOTAL TIME IN SERVICE NOTE

Write down the G1000 total time in service from the FUEL page.

- TIMECHECK 2 MINUTES IN IDLE
- AVIONICS MASTER SWITCH.....OFF
- ENGINE MASTER SWITCHES.....OFF
- STROBE LIGHT.....OFF
- ELECTRIC MASTER SWITCH.....OFF
- IGNITION KEY..... OUT
- PARKING BRAKE AS REQUIRED

Never taxi the aircraft facing fuel pumps with engines running, park parallel to pumps. Stop with plenty of space, then pull the aircraft forward into the fueling position. Do not set parking brake unless on a slope.

PILOT OVERHEAD READING LIGHT.....OFF

“SHUTDOWN CHECKLIST”

MOORING

PARKING BRAKE RELEASE

Push aircraft backward or pull aircraft forward into parking space using the tow bar for steering. Do not turn the nose gear beyond its steering radius in either direction as this will result in damage to the nose gear and steering mechanism.

TOW BAR STOW PROPERLY
SUNSCREENS, STALL WARNING COVER, PITOT COVER, TIE-DOWN,
CHOCKS..... INSTALL

Tie-down ropes are required when the aircraft is left unattended longer than for a normal crew change. If no tie-down ropes are available on ramp, use the aircraft tie-down ropes (in flight gear bag). Secure tie-down ropes to the wing tie-down rings and to the tail skid at approximately 45° angles to the ground. Use bowline

knots, square knots or locked slip knots. Do not use plain slip knots.

Parking brake is required when aircraft is not tied down, but with the crew around. Do not set parking brake when the brakes are overheated!

Wheel chocks are required when the aircraft is not tied down and left unattended. If no wheel chocks available at the FBO, use the aircraft's own wheel chocks (in flight gear bag).

SEAT BELTS.....LOCK

Seat belts of all seats will be fastened.

PERSONAL BELONGINGS AND TRASH..... REMOVE
CANOPY AND DOORSCLOSE & SECURE

Lock with key.

If required: fill in strip in aircraft squawk book and report problem to dispatch and maintenance.

If required: fill in ASR.

Fill out aircraft logbook.

Return aircraft key to dispatch.

If required: close flight plan.

FLOWS ARE INTENDED TO PROVIDE A LOGICAL SEQUENCE IN WHICH THE ITEMS CAN BE DONE AND SHOULD BE FOLLOWED BY A CHECK-LIST AT ALL TIMES, EXCEPT FOR THE WALKAROUND.

PRE—FLIGHT PREPARATION FLOW

- ALL PERTINENT INFORMATION CONCERNING FLIGHT CHECK
- FLIGHT RELEASE FORM COMPLETE
- AIRPLANE SQUAWK BOOK CHECK
- EQUIPMENT REQUIRED FOR FLIGHT CHECK
- AIRCRAFT KEYS AND POUCH CHECK
- I—M—S—A—F—E CHECKLIST CHECK
- MISSION BRIEFING CHECK

WALKAROUND FLOW

APPROACHING THE AIRCRAFT

- NEARBY OBSTACLES CHECK
- FLAP POSITION NOTE
- SURFACES CLEAR
- ANTENNAS CHECK

PRELIMINARY INSPECTION

- IGNITION KEY CHECK OFF & KEY REMOVED
- FRONT CANOPY & REAR DOOR CHECK
- FIRE EXTINGUISHER CHECK SECURE / CONDITION
- UPPER SWITCH PANEL ALL SWITCHES OFF
- ECU SWITCHES AUTO
- ALTERNATOR SWITCHES ON
- LOWER SWITCH PANEL ALL SWITCHES OFF
- GEAR SELECTOR DOWN
- PARKING BRAKE SET
- ELECTRIC MASTER SWITCH ON
- FUEL GAUGES CHECK QUANTITY
- FLAPS LDG
- ELECTRIC MASTER SWITCH OFF
- TRIM NEUTRAL
- SUNSCREENS, PITOT COVER, STALL WARNING COVER & TIE-DOWN REMOVED
- CHOCKS REMOVED
- TOWBAR PROPERLY STOWED
- REQUIRED EQUIPMENT ON BOARD
- FRONT BAGGAGE DOORS SECURE

LEFT WING

- STEPCHECK
- NACELLE UNDERSIDE FUEL COOLER AIR IN & OUTLETCHECK / CLEAR
- WING FLAPSCHECK
- FLAP AND LINKAGE.....CHECK
- FLAP HINGES & SAFETY PINCHECK
- AILERONCHECK
- AILERON AND LINKAGECHECK
- FOREIGN OBJECTS IN AILERON PADDLECHECK
- STATIC DISCHARGERS.....CHECK
- WING TIPCHECK
- POSITION LIGHT / STROBE LIGHTCHECK
- TIE DOWNCHECK / CLEAR
- MAIN FUEL TANK FILLERCHECK / CLOSE
- PITOT PROBECHECK
- STALL WARNING DEVICECHECK
- FUEL TANK AIR OUTLETCHECK
- WING SURFACECHECK
- TANK DRAIN / TANK AIR INLETCHECK
- OPENINGS ON LOWER SURFACECHECK

LEFT ENGINE NACELLE

- 3 AIR INLETS/2 AIR OUTLETS.....CLEAR
- ENGINE OIL LEVEL.....CHECK
- GEARBOX OIL LEVEL.....CHECK
- COWLING.....CHECK
- PROPELLER & SPINNER.....CHECK
- EXHAUST.....CHECK
- VENTING PIPE..... CHECK FOR BLOCKAGE
- GASCOLATOR/AIR INLET.....DRAIN/CLEAR
- NACELLE UNDERSIDE.....CHECK
- ▶ AUXILIARY TANK FILLER.....CHECK/CLOSE
- ▶ AUXILIARY TANK VENT OUTLET.....CHECK
- ▶ AUXILIARY TANK DRAIN.....CHECK/DRAIN

LEFT MAIN LANDING GEAR

- LANDING GEAR STRUT & LOCKCHECK
- DOWN & UNLOCK SWITCHES (3X)CHECK
- WEAR, TREAD & DEPTH OF TIRECHECK
- WHEEL BRAKES / HYDRAULIC BRAKE LINE.....CHECK
- SLIP MARKS.....CHECK
- LANDING GEAR DOORS.....CHECK

CHOCKS REMOVE

FRONT FUSELAGE / NOSE LANDING GEAR

FRONT CANOPY LEFT & RIGHT SIDE CHECK
 LEFT & RIGHT FRONT BAGGAGE DOORS CHECK / CLOSE & LOCK
 NOSE LDG GEAR STRUT, LOCK & CENTERING DEVICE CHECK
 DOWN & UPLOCK SWITCHES CHECK
 WEAR, TREAD & DEPTH OF TIRE CHECK
 SLIP MARKS CHECK
 CHOCKS REMOVE
 LANDING GEAR & LINKAGE CHECK
 OAT SENSOR CHECK
 EPU CONNECTOR CHECK
 TOWBAR CHECK REMOVED

RIGHT MAIN LANDING GEAR

LANDING GEAR STRUT & LOCK CHECK
 DOWN & UPLOCK SWITCHES (3X) CHECK
 WEAR, TREAD & DEPTH OF TIRE CHECK
 WHEEL BRAKES / HYDRAULIC BRAKE LINE CHECK
 SLIP MARKS CHECK
 LANDING GEAR DOORS CHECK
 CHOCKS REMOVE

RIGHT ENGINE NACELLE

3 AIR INLETS/2 AIR OUTLETS CLEAR
 ENGINE OIL LEVEL CHECK
 GEARBOX OIL LEVEL CHECK
 COWLING CHECK
 PROPELLER & SPINNER CHECK
 EXHAUST CHECK
 VENTING PIPE CHECK
 GASCOLATOR/AIR INLET DRAIN/CLEAR
 NACELLE UNDERSIDE CHECK
 ► AUXILIARY TANK FILLER CHECK/CLOSE
 ► AUXILIARY TANK VENT OUTLET CHECK
 ► AUXILIARY TANK DRAIN CHECK/DRAIN

RIGHT WING

CABIN VENT AIR INLET CHECK/CLEAR
 TANK DRAIN/TANK AIR INLET CHECK/DRAIN

TANK AIR OUTLET.....CHECK
 OPENINGS ON LOWER SURFACE.....CHECK
 ENTIRE WING SURFACE.....CHECK
 MAIN FUEL TANK FILLER.....CHECK/CLOSE
 WING TIP.....CHECK
 STATIC DISCHARGERS.....CHECK
 POSITION LIGHT, STROBE LIGHT.....CHECK
 TIE-DOWN.....CHECK/CLEAR
 AILERON & LINKAGE.....CHECK
 FOREIGN OBJECTS IN AILERON PADDLE.....CHECK
 FLAP AND LINKAGE.....CHECK
 FLAP HINGES & SAFETY PIN.....CHECK
 NACELLE UNDERSIDE FUEL COOLER AIR IN- & OUTLET.....CHECK/CLEAR
 STEP.....CHECK

FUSELAGE RIGHT SIDE / UNDERSIDE

AFT CANOPY.....CHECK
 FUSELAGE SKIN.....CHECK
 ANTENNAS UPPER SIDE.....CHECK
 STATIC PORT.....CHECK

EMPANNAGE

STABILIZERS & CONTROL SURFACES/ELEVATOR TIPS.....CHECK
 HINGES.....CHECK
 ELEVATOR TRIM TAB.....CHECK
 RUDDER TRIM TAB.....CHECK
 TIE DOWN.....CHECK/CLEAR
 TAIL SKID & LOWER FIN.....CHECK
 STATIC DISCHARGERS.....CHECK

FUSELAGE LEFT SIDE / UNDERSIDE

STATIC PORT.....CHECK
 FUSELAGE SKIN.....CHECK
 FUSELAGE UNDERSIDE.....CHECK FOR CONTAMINATION
 ANTENNAS UNDERSIDE.....CHECK
 AFT CANOPY.....CHECK

MISCELLANEOUS

FOR NIGHT FLIGHT ONLY:

INTERIOR LIGHTING.....ON & CHECK
 EXTERIOR LIGHTING.....ON & CHECK
 ALL LIGHTING SWITCHES.....OFF

FOR FLIGHT IN IMC ONLY:

PITOT HEATON
 PITOT HEAT ANNUNCIATOR.....EXTINGUISHED
 PITOT TUBE.....CHECK WARM
 PITOT HEATOFF

FOR FLIGHT IN KNOWN ICING CONDITIONS ONLY:

DE-ICING FLUIDCHECK QTY
 ICE LIGHT ON & CHECK
 ICE LIGHT OFF

COCKPIT PREPARATION FLOW

RUDDER PEDALS.....ADJUSTED & LOCKED
 FLIGHT CONTROLS.....PROPER OPERATION
 SEATBELTS.....LOCKED
 REAR DOORCLOSED & SECURED
 FRONT CANOPYPOSITION 1 OR 2
 FRONT BAGGAGE DOORS.....CLOSED & LOCKED
 AVIONICS MASTER SWITCH.....OFF
 ELECTRIC MASTER SWITCH.....ON
 ENGINE MASTER SWITCHES.....OFF
 IGNITION KEY.....READY & OFF
 PITOT HEAT.....OFF
 ALTERNATE STATIC.....CLOSED
 ALTERNATORS.....CHECK ON
 ECU SWAP.....CHECK AUTOMATIC
 DIMMER CONTROL.....SET
 LIGHT SWITCHES.....OFF
 EMERGENCY SWITCH.....CHECK OFF & GUARDED
 STANDBY INSTRUMENTS.....CHECK
 MAGNETIC COMPASS.....CHECK
 DEVIATION CARD.....CHECK
 ELT.....ARMED
 ►DE-ICING SYSTEM.....CHECK
 CIRCUIT BREAKERS.....IN
 G1000.....ACKNOWLEDGE
 FUEL TEMPERATURE.....CHECK
 FUEL QUANTITY.....CHECK
 FUEL CALCULATOR.....SET
 TOTAL TIME IN SERVICE (HOBBS).....NOTE
 G1000.....BACKUP OR REV MODE
 GEAR WARNING/FIRE DETECTOR.....CHECK

MANUAL GEAR EXTENSION HANDLE.....CHECK PUSHED IN
 ALTERNATE AIR.....CHECK CLOSED
 FLAPS.....CHECK LDG
 RUDDER TRIM.....CHECK/SET
 PARKING BRAKE.....SET
 DEFROSTER SWITCH.....OFF
 HEATER SWITCH.....OFF
 VARIABLE ELEVATOR BACKSTOP.....CHECK
 FLAPS.....UP
 FUEL SELECTORS.....ON
 ELEVATOR TRIM.....SET FOR T/O
 THROTTLE FRICTIONADJUST
 AVIONICS MASTER SWITCH.....ON
 ATIS/VFR OR IFR CLEARANCE.....COPY
 AVIONICS MASTER SWITCH.....OFF
 BRIEFING.....PERFORM

"BEFORE START CHECKLIST"

ENGINE START FLOW

STROBE LIGHTS (DAY) OR POSITION LIGHTS (NIGHT).....ON
 ENGINE MASTER SWITCH (L).....ON
 ENGINE GLOW.....ON
 ENGINE INDICATIONS.....CHECK
 ENGINE GLOW.....CHECK OFF
 PROPELLER AREA.....CLEAR
 BRAKES.....HOLD
 IGNITION KEY.....START
 STARTER ANNUNCIATION.....OFF
 RPM.....CHECK
 OIL PRESSURECHECK
 ANNUNCIATIONS/ ENGINE/ SYSTEM PAGE.....CHECK

REPEAT FOR OTHER ENGINE

AVIONICS MASTER SWITCH.....ON

"AFTER START CHECKLIST"

BEFORE TAXI FLOW

PITOT HEAT.....CHECK
 G1 000 SET UP (I-F-R-B).....COMPLETE
 AUTO PILOT.....TEST

ELECTRICAL TRIM.....TEST
ECUTEST
ECU SWAP.....CHECKED/AUTO

"BEFORE TAXI CHECKLIST"

TAXI FLOW

TAXI LIGHT.....ON
FUEL SELECTORS.....CROSSFEED
AREA.....FREE
PARKING BRAKE.....RELEASE
BRAKES.....CHECK
FLIGHT INSTRUMENTS.....CHECK
FUEL SELECTORS.....ON

BEFORE TAKEOFF FLOW

CONTROLS.....FREE
CIRCUIT BREAKERS.....IN
FLAPS.....AS REQUIRED
G1000.....REV. MODE
TRIMS.....SET
FUEL SELECTORS.....ON
SHORT BRIEFING.....PERFORM
FRONT CANOPY.....CLOSED & LOCKED
DOOR WARNING.....CHECK OFF

"BEFORE TAKEOFF CHECKLIST"

LINE UP FLOW

**MAYBE PERFORMED BY PM ON REQUEST OF PF*

APPROACH.....FREE
*LANDING LIGHT..... ON
*TAXI LIGHT OFF
*STROBE LIGHTS ON
ALTIMETER.....CHECK (3X)
COMPASS SYSTEM..... CHECK
*PITOT HEAT AS REQUIRED

TAKEOFF FLOW

BRAKES.....HOLD
POWER LEVERSFULLY FORWARD
PARAMETERS..... CHECK

AFTER TAKEOFF FLOW

GEAR.....UP
 FLAPS.....UP
 ENGINE INSTRUMENTS.....CHECK
 LANDING LIGHT.....AS REQUIRED
 TRANSPONDER.....CODE & ALT
 MAP PAGE.....SWITCH

“AFTER TAKEOFF CHECKLIST”

CRUISE FLOW

FUEL STATUS.....CHECK
 ENGINE INDICATIONS / SYSTEM PAGECHECK

DESCENT / APPROACH FLOW

LANDING LIGHT.....ON
 ALTIMETERS.....SET(3X)
 ANNUNCIATIONS/ ENGINE/ SYSTEM PAGE.....CHECKED
 GEAR WARNING HORN.....CHECKED
 FUEL SELECTORS.....ON
 DE-ICING SYSTEM.....CONSIDER

“DESCENT / APPROACH CHECKLIST”

LANDING FLOW

LANDING GEAR.....DOWN-3 GREENS
 BRAKES.....CHECKED
 PARKING BRAKE.....OFF
 FLAPS.....AS REQUIRED

“LANDING CHECKLIST”

SHORT FINAL FLOW

LANDING GEAR.....DOWN-3 GREENS

AFTER LANDING FLOW

**MAYBE PERFORMED BY PM ON REQUEST OF PF*

*LANDING LIGHT.....OFF

- *TAXI LIGHT.....ON
- *PITOT HEAT.....OFF
- TIMER.....START-ENGINE IDLE
- TRANSPONDER.....CODE & GROUND
- *FLAPS.....UP

"AFTER LANDING CHECKLIST"

ENGINE SHUTDOWN FLOW

- TAXI LIGHT.....OFF
- TOTAL TIME IN SERVICE (FUEL PAGE).....NOTE
- TIME.....CHECK 2 MINUTES IDLE
- AVIONICS MASTER SWITCH.....OFF
- ENGINE MASTER SWITCHESOFF
- STROBE LIGHTS.....OFF
- ELECTRIC MASTER SWITCH.....OFF
- IGNITION KEY.....OUT
- PARKING BRAKE.....AS REQUIRED
- PILOT OVERHEAD READING LIGHT OFF

"SHUTDOWN CHECKLIST"

MOORING FLOW

- PARKING BRAKE.....RELEASE
- TOW BAR.....STOW PROPERLY
- SUNSCREENS, STALL WARNING COVER, TIE DOWNS, PITOT COVER.....INSTALL
- SEAT BELTS.....LOCK
- PERSONAL ITEMS.....REMOVE
- CANOPY.....CLOSE AND LOCK

INTENTIONALLY LEFT BLANK

BEFORE START CHECKLIST

WALK AROUND COMPLETED
 RUDDER PEDALS ADJUSTED AND LOCKED
 SEAT BELTS LOCKED
 PARKING BRAKE SET
 GEAR WARNING/FIRE DETECTOR CHECKED
 AVIONICS MASTER SWITCH OFF
 ELECTRIC MASTER SWITCH ON
 ALTERNATORS ON
 ECU SWAP AUTO
 CIRCUIT BREAKERS IN
 FLAPS AS REQUIRED
 G1000 ACKNOWLEDGED
 FUEL QUANTITY CHECKED
 FUEL TEMPERATURE CHECKED
 ALTERNATE AIR CLOSED
 THROTTLE IDLE
 VARIABLE ELEVATOR BACKSTOP CHECKED
 FUEL SELECTORS ON
 TRIMS SET

AFTER START CHECKLIST

OIL PRESSURE (2X) CHECKED
 ANNUNCIATORS/ENGINE/SYSTEM PAGE CHECKED

BEFORE TAXI CHECKLIST

PITOT HEAT CHECKED
 G1000 SET UP COMPLETED
 AUTO PILOT TEST COMPLETED
 ELECTRICAL TRIM TEST COMPLETED
 ECU TESTED
 ECU SWAP CHECKED / AUTO

BEFORE TAKEOFF CHECKLIST

FLIGHT INSTRUMENTS CHECKED
 ENGINE INSTRUMENTS CHECKED
 ALTERNATORS ON
 ECU SWAP AUTO
 FLAPS AS REQUIRED
 FUEL SELECTORS ON
 TRIMS SET
 FLIGHT CONTROLS FREE
 CANOPY & DOORS CLOSED & SECURED

AFTER TAKEOFF CHECKLIST

GEAR UP-NO LIGHTS
 FLAPS UP
 ENGINE INSTRUMENTS CHECKED
 LANDING LIGHT AS REQUIRED

DESCENT APPROACH CHECKLIST

- These items only when remaining in the pattern

LANDING LIGHT ON
 ALTIMETERS SET(3X)
 • ANNUNCIATIONS/ENGINE/SYSTEM PAGE CHECKED
 • GEAR WARNING HORN CHECKED
 FUEL SELECTORS ON
 ► DE-ICING SYSTEM CONSIDER

LANDING CHECKLIST

LANDING GEAR DOWN-3 GREENS
 BRAKES CHECKED
 PARKING BRAKE OFF
 FLAPS AS REQUIRED

AFTER LANDING CHECKLIST

LANDING LIGHT OFF
 TAXI LIGHT ON
 PITOT HEAT OFF
 TIMER STARTED
 TRANSPONDER CODE & GROUND
 FLAPS UP

SHUTDOWN CHECKLIST

PILOT OVERHEAD READING LIGHT CHECK OFF
 LIGHTS OFF
 AVIONIC MASTER SWITCH OFF
 ENGINE MASTER SWITCHES OFF
 ELECTRIC MASTER SWITCH OFF
 IGNITION KEY OUT
 PARKING BRAKE AS REQUIRED

INTENTIONALLY LEFT BLANK

EMER

PROC

PERF

W&B

LIM

PAT

EMER

NORM

MAN

PPT

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NON NORMAL WORKMETHOD

1. **Announce** the failure or read the G1000 annunciation.
E.g. "Engine failure" or "Hi oil temp".
2. Perform **Recall items** (by memory) if applicable. When completed: "**Recall items completed**".
3. Brief the **Short term plan** to PM. Aircraft vertical & lateral flight path (altitude & heading) during the checklist reading (1–2min).
4. Perform the **Emergency checklist** (confirm recall items completed correctly) Announce: "**____V-list completed**".
5. Perform the **Normal checklist** (if applicable).
6. **Longterm plan**: Collect all necessary info to make a decision (Wx, Notams, Fuel remaining, ...).
7. Announce **Decision** to: PM, ATC, Passengers.

CHECKLIST TITLE

The title of the checklist refers to the type of failure. **G1000** annunciated failures are between filled boxes. **Unannunciated** failures are not.

Warnings and Emergencies are RED.

Cautions and Abnormals are AMBER.



G1000 WARNINGS



OTHER EMER PROC



G1000 CAUTIONS



OTHER ABNORMAL PROC

RECALL ITEMS

Some emergency checklist require an immediate response/ action from the pilot. The recall items must be performed quickly without hesitation but without rushing either.



Recall items are identified in by a dashed line box around the critical items.

Other checklists that are not boxed must be performed in read and do.

G1000 WARNINGS

Warning	Page	Description
L/R OIL PRES	4.7	Oil pressure low (< 1 bar)
L/R OIL TEMP	4.7	Oil temp high (> 140°C)
L/R GBOX TEMP	4.8	Gearbox temp high (> 120°C)
L/R FUEL TEMP	4.8	Fuel temp high (>75°C)
L/R STARTER	4.8	Starter not disengaging
L/R ENG TEMP	4.9	Coolant temp high (> 105°C)
L/ENG FIRE	4.10	Engine fail/fire during takeoff
	4.11	Engine fail/fire in flight
		Engine fire on ground
L/R ALTN AMPS	4.15	High current (> 60A)
DOOR OPEN	4.22	Unlocked door(s) Front/Rear Canopy/Bag doors not closed and locked

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PERF

OTHER EMER PROC

Situation	Page
ENGINE TROUBLESHOOTING	4.12
OSCILLATING RPM	4.13
RPM OVERSPEED	4.14
ENGINE RESTART	4.14
COMPLETE ELECTRICAL FAILURE	4.15
CABIN SMOKE > 10.000 FT	4.17
CABIN FIRE > 10.000 FT	4.17
ELECTRICAL FIRE ON GROUND	4.18
ELECTRICAL FIRE IN FLIGHT	4.18
UNINTENTIONAL FLIGHT IN ICING	4.19
ICE PROTECTION FAILURE	4.19
LDG WITH DEFECTIVE MAIN TIRE	4.20
LDG WITH DEFECTIVE BRAKES	4.20
LDG GEAR UNSAFE WARNING	4.20
MANUAL GEAR EXTENSION	4.21
GEAR UP LANDING	4.21
EMERGENCY DESCENT	4.22
SUSPICION CARBON MONOXIDE	4.22

G1000 CAUTIONS

Warning	Page	Description
L/R ECU A FAIL	4.23	ECU A failed
L/R ECU B FAIL	4.23	ECU B failed
L/R COOL LVL	4.23	Engine coolant level low
L/R ALTN FAIL	4.24	Alternator failed
L/R VOLTS LOW	4.24	Bus voltage too low
PITOT FAIL	4.25	Pitot heating system failed
PITOT HT OFF	4.25	Pitot heating system OFF
STALL HT FAIL	4.25	Stall warning heating failed
STALL HT OFF	4.25	Stall warning heating OFF
DEICE LVL LO	4.26	De-icing fluid level low
DEIC PRES LO	4.26	De-icing pressure low
DEIC PRES HI	4.26	De-icing pressure high
L/R FUEL LOW	4.27	Main tank fuel quantity low
STICK LIMIT	4.27	Stick limiting system failed

OTHER ABNORMAL PROC

Situation	Page
BOTH ALTN FAIL	4.24
HYD PUMP FAIL OR CONT OPS	4.29

ENGINE IND OUTSIDE GRN RANGE

Situation	Description/Action
COOLANT TEMP HI	Refer to EMER 'L/R ENG TEMP'
COOLANT TEMP LO (during low power descent from altitude, consider increasing power)	<ul style="list-style-type: none"> • Check G1 000 for LOW COOLANT LVL caution light • If LOW COOLANT LVS caution light ON: <ul style="list-style-type: none"> • Reduce power on affected engine • Expect loss of coolant fluid • Expect ENG FAIL, LAND ASAP
OIL TEMP HI	Refer to EMER 'L/R OIL TEMP'
OIL TEMP LO	<ul style="list-style-type: none"> • Increase power • Reduce airspeed
OIL PRES HI	<ul style="list-style-type: none"> • Check oil temp and coolant temp • If within green range: <ul style="list-style-type: none"> • Oil pressure indication may be faulty, monitor temp • If outside green range: <ul style="list-style-type: none"> • Reduce power on affected engine • Expect ENG FAIL, LAND ASAP
OIL PRES LO	Refer to EMER 'L/R OIL PRES'
FUEL TEMP HI	Refer to EMER ' L/R FUEL TEMP'
FUEL TEMP LO	<ul style="list-style-type: none"> • Increase power • Reduce airspeed <ul style="list-style-type: none"> • If not recovered: expect ENG FAIL, LAND ASAP
VOLTS LO	<ul style="list-style-type: none"> • On ground: <ul style="list-style-type: none"> • Check circuit breakers & increase RPM • If not solved: terminate flight • In flight: <ul style="list-style-type: none"> • Check circuit breakers • Reduce ELEC load • If not solved: Refer to ABN 'L/R ALTN FAIL
RPM HI	<ul style="list-style-type: none"> • Reduce power • Keep RPM in green range <ul style="list-style-type: none"> • If not solved: Refer to EMER 'RPM OVERSPEED'

TOC

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MAN

NORM

EMER

PAT

LIM

W&B

PERF

L/R OIL PRESS

POWER AFFECTED ENGINE..... REDUCE

- Be prepared for a loss of oil and engine failure.

LAND ASAP

L/R OIL TEMP

OIL PRESSURE.....CHECK

- If oil pressure is too low

POWER AFFECTED ENGINE REDUCE

- Be prepared for a loss of oil and engine failure.

LAND ASAP

- If oil pressure is in green range

POWER AFFECTED ENGINE..... REDUCE

AIRSPEED INCREASE

- If oil temperature not returning to green range
 - Be prepared for a loss of oil and engine failure.

LAND ASAP

- If oil temperature returning to green range
 - Continue normal operations.

L/R GBOX TEMP

POWER AFFECTED ENGINE REDUCE
AIRSPEED INCREASE

- If not returning to green range
 - Be prepared for a loss of oil and engine failure.

LAND ASAP

- If returning to green range
 - Continue normal operations.

L/R FUEL TEMP

POWER AFFECTED ENGINE REDUCE
AIRSPEED INCREASE

- If not returning to green range

LAND ASAP

L/R STARTER

POWER AFFECTED ENGINE IDLE
ENGINE MASTER AFFECTED ENGINE OFF
ELECTRIC MASTER OFF

L/R ENGINE TEMP

G1000 LOW COOL LVL CAUTION.....CHECK

- If LOW COOL LVL caution light OFF
 - During climb

POWER AFFECTED ENGINE.....REDUCE BY ~ 10%
AIRSPEED INCREASE BY ~ 10 KIAS
TIMER..... START

- If coolant temp not returning to green range within 60 seconds

POWER AFFECTED ENGINE.....REDUCE FURTHER
AIRSPEED INCREASE

- Consider level off or descent according to MEA/MSA
- During cruise

POWER AFFECTED ENGINE..... REDUCE (as required)
AIRSPEEDINCREASE (as required)

This requires to increase power on the other engine (check temperature!) or start a descent according to MEA/MSA.

- If coolant temperature not returning to green range
 - Be prepared for a loss of oil and engine failure

LAND ASAP

- If LOW COOL LVL caution light ON

POWER AFFECTED ENGINE..... REDUCE

- Expect loss of coolant fluid. Be prepared for an engine failure.

L/R ENGINE FIRE/FAILURE AND SHUTDOWN

DURING TAKEOFF

PERFORM A REJECTED TAKEOFF OR EMERGENCY RE-LANDING

POWER LEVERS IDLE
 BRAKESAPPLY
 ATC..... INFORM

•If necessary:

ENGINE MASTER (BOTH)..... OFF
 FUEL SELECTORS (BOTH)..... OFF
 ELECTRICAL MASTER OFF

•In case of fire:

CABIN HEAT AND DEFROST OFF

CONTINUED ON NEXT PAGE

L/R ENGINE FIRE/FAILURE AND SHUTDOWN

IN FLIGHT (AIRSPEED ABOVE 68 KIAS)

POWER LEVERS MAX
AIRPLANE CONTROL..... MAINTAIN

3-5° bank into good engine
Ball ½ deflected into good engine
Speed Vyse 82KIAS

GEAR UP
FLAPS..... UP

IDENTIFY..... DEAD FOOT = DEAD ENGINE
Announce: “Engine 1/2 failed”

THROTTLE ENGINE # RETARD
Verify No yaw; sound change

• IF CONDITIONS PERMIT (A/C under control, V > 82KIAS, above MSA)

ENGINE TROUBLESHOOTING..... PERFORM
WARNING: Never in case FIRE
Refer to page 4.

• IF CONDITIONS DO NOT PERMIT.

ENGINE MASTER SWITCH # OFF
After confirmation from PF

ALTERNATOR # OFF

FUEL SELECTOR # OFF

X-FEED CONSIDER
Max imbalance 5USG

• In case of fire:

CABIN HEAT AND DEFROST OFF

LAND ASAP

ENGINE TROUBLESHOOTING

POWER LEVER (AFFECTED ENGINE) IDLE

- If in icing conditions:

ALTERNATE AIR OPEN

FUEL QUANTITY CHECK

FUEL SELECTOR (AFFECTED ENGINE) ON OR X-FEED (AS REQUIRED)

ECU SWAP (AFFECTED ENGINE)..... ECU B

- If unsuccessful:

ECU SWAP (AFFECTED ENGINE) AUTO

CIRCUIT BREAKERS..... CHECK/RESET

- If unsuccessful:

AIRSPEED MIN 82 KIAS-VYSE

LANDING GEAR UP

FLAPS UP

ENGINE MASTER (AFFECTED ENGINE) OFF

ALTERNATOR (AFFECTED ENGINE) OFF

FULE SELECTOR (AFFECTED ENGINE) OFF

LAND ASAP

OSCILLATING RPM

POWER LEVER (AFFECTED ENGINE)..... CHANGE

- If unsuccessful:

ECU SWAP (AFFECTED ENGINE)..... ECU B

- If unsuccessful:

ECU SWAP (AFFECTED ENGINE)..... AUTO

LAND ASAP

RPM OVERSPEED

POWER LEVER (AFFECTED ENGINE) REDUCE

- If unsuccessful:

ECU SWAP (AFFECTED ENGINE) ECU B

- If unsuccessful:

ECU SWAP (AFFECTED ENGINE) AUTO

LAND ASAP

- Be prepared for a loss of oil and engine failure

ENGINE RESTART

AIRSPEED 80–120 KIAS

PRESSURE ALTITUDE MAX 6000FT

POWER LEVER (AFFECTED ENGINE) IDLE

FUEL SELECTOR (AFFECTED ENGINE) ON

ALTERNATE AIR AS REQUIRED

ENGINE MASTER (AFFECTED ENGINE) ON

STARTER ENGAGE IF NECESSARY

- If engine starts:

POWER (AFFECTED ENGINE) MODERATE

ENGINE INSTRUMENTS CHECK GREEN RANGE

ALTERNATOR (AFFECTED ENGINE) ON

L/R ALTN AMPS

CIRCUIT BREAKERSCHECK
ELECTRICAL LOAD..... REDUCE

LAND ASAP

COMPLETE ELECTRICAL FAILURE

CIRCUIT BREAKERSCHECK

- If unsuccessful:

EMERGENCY SWITCHON
FLOOD LIGHTON
POWER SET ACC. TO LEVER POSITION OR NOISE
FLAPS VERIFY POSITION

LAND ASAP

Landing gear may slowly extend.

- For landing:

MANUAL LANDING GEAR EXTENSIONAPPLY

ENGINE FIRE ON GROUND

POWER LEVERS (BOTH)..... IDLE
 ENGINE MASTERS (BOTH)..... OFF
 FUEL SELECTORS (BOTH)..... OFF
 ATC INFORM—CONSIDER MAYDAY CALL
 ELECTRIC MASTER OFF

- When engine and aircraft stopped:

CANOPY OPEN
 AIRCRAFT EVACUATE

CABIN SMOKE ABOVE 10.000FT

► OXYGEN SYSTEM INSTALLED (OPTIONAL)

OXYGEN..... CHECK ON
EMERGENCY DESCENT..... INITIATE

- When passing 10.000ft:

OXYGEN OFF

LAND ASAP

CABIN FIRE ABOVE 10.000FT

► OXYGEN SYSTEM INSTALLED (OPTIONAL)

OXYGEN.....OFF
EMERGENCY DESCENT..... INITIATE

LAND ASAP

ELECTRICAL FIRE IN FLIGHT

EMERGENCY SWITCH ON
 ATC INFORM—CONSIDER MAYDAY CALL
 AVIONIC MASTER OFF
 ELECTRIC MASTER OFF
 CABIN HEAT & DEFROST OFF
 EMERGENCY WINDOWS OPEN (AS NECESSARY)
 CANOPY (BELOW 120 KIAS) UNLATCH (IF NECESSARY)

LAND ASAP

ELECTRICAL FIRE ON GROUND

ATC INFORM—CONSIDER MAYDAY CALL
 ELECTRIC MASTER OFF
 POWER LEVERS (BOTH) IDLE
 ENGINE MASTERS (BOTH) OFF
 FUEL SELECTORS (BOTH) OFF

- When engine and aircraft stopped:

CANOPY OPEN
 AIRCRAFT EVACUATE

UNINTENTIONAL FLIGHT INTO ICING

PITOT HEAT ON
 CABIN HEAT & DEFROST ON
 POWER INCREASE
 *DE-ICE SYSTEMS USE AS APPROPRIATE
 ALTERNATE AIR OPEN (AS REQUIRED)
 EMERGENCY WINDOWS OPEN (AS REQUIRED)

- If pitot heat fails:

ALTERNATE STATIC VALVE OPEN
 EMERGENCY WINDOWS CLOSE

ICE PROTECTION FAILURE

AIRSPEED MIM 121 KIAS
 FLAPS MAX APP

- For approach with residual ice:

FLAPS APP
 APPROACH SPEED MIM 82 KIAS
 LANDING DISTANCE (ACTUAL) X 1.4

LDG WITH DEFECTIVE MAIN TIRE

ATC INFORM

- For landing:

TOUCHDOWN L OR R OF CENTERLINE (SIDE OF GOOD TIRE)

WING KEEP LOW (SIDE OF GOOD TIRE)

DIRECTIONAL CONTROL MAINTAIN WITH BRAKES

LDG WITH DEFECTIVE BRAKES

- After touchdown (if necessary):

ENGINE MASTERS (BOTH) OFF

FUEL SELECTORS (BOTH) OFF

ELECTRIC MASTER OFF

LDG GEAR UNSAFE WARNING

- If on for more than 20 seconds:

AIRSPEED MAX 156 KIAS

- In cold temperature:

AIRSPEED MAX 110 KIAS

LANDING GEAR RECYCLE

- If landing gear EXTENSION unsuccessful: perform MANUAL GEAR EXTENSION
- If landing gear RETRACTION unsuccessful: consider flight with landing gear down

MANUAL GEAR EXTENSION

AIRSPEED MAX 156 KIAS
GEAR INDICATOR LIGHTS TEST
ELECTRIC MASTER..... CHECK ON
BUS VOLTAGE CHECK NORMAL
GEAR SELECTOR DOWN
MANUAL EXTENSION HANDLE PULL

- If necessary:

AIRSPEED MAS 110 KIAS
YAW APPLY MODERATELY

GEAR INDICATOR LIGHTS CHECK 3 GREENS

GEAR UP LANDING

- LANDING GEAR COMPLETELY RETRACTED

ATC INFORM
APPROACH..... NORMAL

- Just before touchdown:

POWER LEVERS (BOTH) IDLE

- After touchdown:

ENGINE MASTERS (BOTH) OFF
FUEL SELECTORS (BOTH) OFF
ELECTRIC MASTER..... OFF

EMERGENCY DESCENT

FLAPS..... UP
 LANDING GEAR DOWN
 POWER LEVERS (BOTH)..... IDLE
 AIRSPEEDAS REQUIRED
 LIGHTS..... ON
 ATCINFORM
 Level off FL100 (10.000ft) or MEA/MSA, whichever is higher.

SUSPICION CARBON MONOXIDE

CABIN HEAT & DEFROST OFF
 VENTILATION (6 OUTLETS)OPEN
 EMERGENCY WINDOWS.....OPEN
 CANOPY (BELOW 120 KIAS)UNLATCH (IF NECESSARY)
 Push up and lock in the cooling gap position.

DOOR OPEN

AIRSPEEDREDUCE
 CANOPY, REAR DOOR & FRONT BAGGAGE DOORS.....CHECK

- If unable to latch canopy or rear door, or if front baggage doors verified open (visually):

LAND ASAP

Never unlatch the rear door in flight.

L/R ECU A (OR B) FAIL

ON GROUND
TERMINATE FLIGHT PREPARATION

IN FLIGHT

In case of ECU A fail the system automatically switches to ECU B.

ECU TEST BUTTON..... PRESS FOR MORE THAN 2 SECONDS

- If ECU A (B) CAUTION MESSAGE re-appears and cannot be reset:

LAND ASAP

- If ECU A (B) CAUTION MESSAGE can be reset:

FLIGHT..... CONTINUE

Maintenance action required after landing.

L/R COOL LVL

ENGINE INSTRUMENTS/ANNUNCIATIONS MONITOR
COOLANT TEMPERATURE..... CHECK

Continue with 'ENGINE INSTRUMENT INDICATIONS OUTSIDE GREEN RANGE-COOLANT TEMPERATURE'.

L/R ALTN FAIL

ALTN (affected side) OFF
 BUS VOLTAGE MONITOR
 ELECTRICAL LOAD REDUCE

- If both ALTN failed:

BOTH ALTN FAIL CHECKLIST PERFORM

BOTH ALTN FAIL

AVIONICS MASTER OFF
 LH/RH ALTN OFF
 TRANSPONDER STANDBY
 LANDING GEAR DOWN

- When landing gear down & locked:

MANUAL GEAR EXTENSION HANDLE PULL

STALL/PITOT HEAT OFF
 ALL LIGHTS OFF

Expect battery power to last for 30 minutes

Expect engine stoppage after this time.

LAND ASAP

L/R VOLTS LOW

Possible reasons are:

- Fault in ELEC PWR supply
- RPM too low.

Continue with 'ENGINE INSTRUMENT INDICATIONS OUTSIDE GREEN RANGE-VOLTS LOW'.

PITOT FAIL/PITOT HT OFF

PITOT HEAT CHECK ON

- If PITOT HEAT cannot be recovered:

Avoid icing conditions.

- If icing conditions cannot be avoided:

Expect failure of the pitot-static system.

ALTN STATIC VALVE OPEN

Leave area with icing conditions (see EMER UNINTENTIONAL FLIGHT INTO ICING.)

STALL HT FAIL/STALL HT OFF

PITOT HEAT CHECK ON

- If STALL HEAT cannot be recovered:

Avoid icing conditions.

- If icing conditions cannot be avoided:

Expect loss of aural stall warning.

Leave area with icing conditions (see EMER UNINTENTIONAL FLIGHT INTO ICING.)

DE-ICE LVL LO

Maximum duration of ice protection:

- NORMAL mode: 45 min.
- HIGH mode: 22 min.

Leave area with icing conditions accordingly.

DE-ICE PRES LO

DE-ICE HIGH

- If DE-ICE PRES LO light still ON:

PUMP1 /PUMP 2SELECT OTHER PUMP
If necessary prime pump by activating windshield pump.

- If DE-ICE PRES LO light still ON:

ALTERNATE SWITCH.....ACTIVATE

- If DE-ICE PRES LO light still ON:

EMER ICE PROTECTION FAILURE CHKLST APPLY

- If DE-ICE PRES LO OFF:

FLIGHTCONTINUE
De-icing fluid flow: 30 liters/hour

ICE PROTECTION SYSTEM OPERATION MONITOR
DE-ICING FLUID LEVEL CHECK PERIODICALLY

DE-ICE PRES HI

Possible reduced system performance.

Filter cartridge to be replaced at next scheduled maintenance.

L/R FUEL LOW

FUEL QUANTITY CHECK

- If LH and RH quantities show a remarkable difference:

Expect fuel leak on side with lower indication.

FUEL SELECTOR (SIDE WITH LOW FUEL INDIC) X-FEED

STICK LIMIT

1 OR 2 POWER LEVERS SET FOR MORE THAN 20% LOAD

Elevator variable backstop is INOPERATIVE.
Aviod STALL in any condition.

2 POWER LEVERS SET FOR LESS THAN 20% LOAD

Elevator variable backstop is ALWAYS ACTIVE.
Reduced ELEV capacity (travel).

MIM VREF 76/78 KIAS

HYD PUMP FAIL OR CONTINUOUS OPERATION

GEAR INDICATOR LIGHTS..... CHECK

Prepare for manual landing gear extension.

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PATTERNS

TOC

PPT

MAN

NORM

EMER

PAT

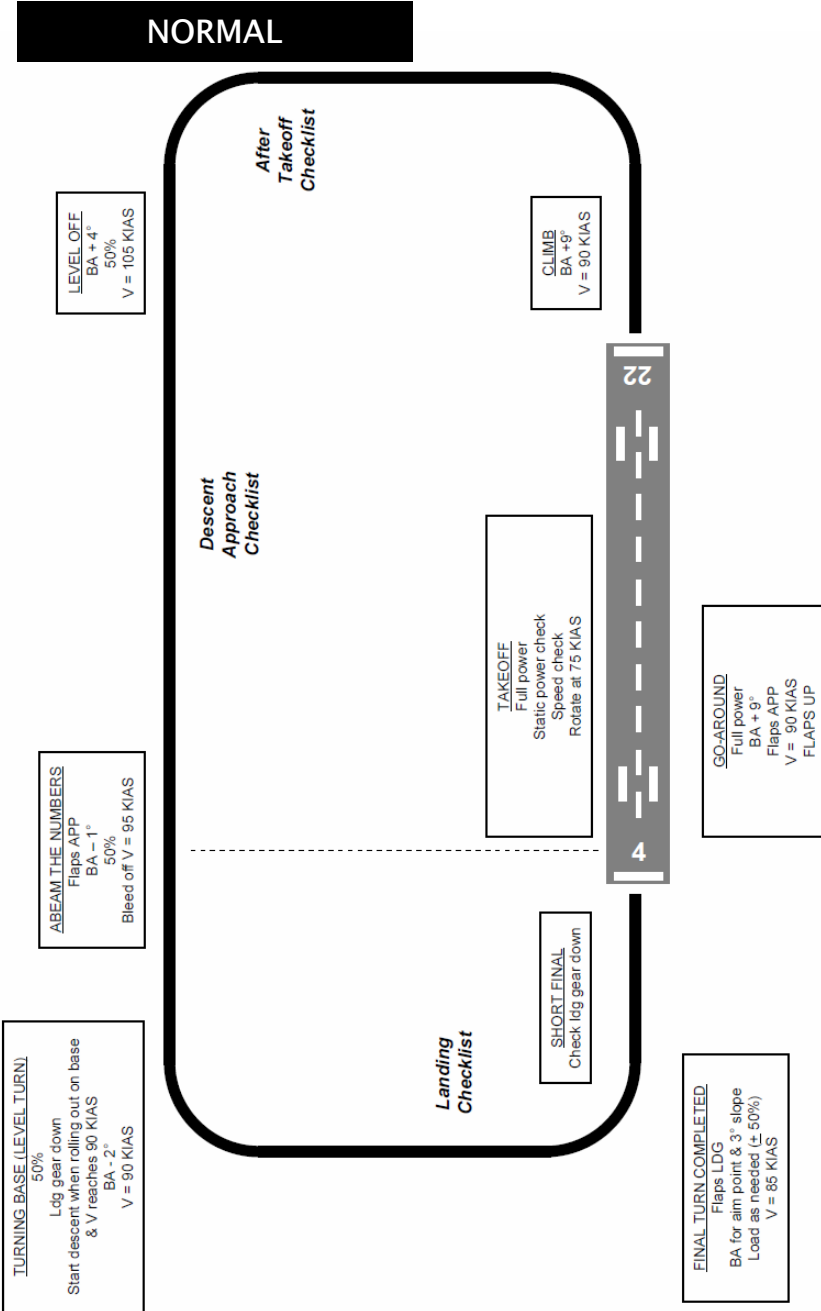
LIM

W&B

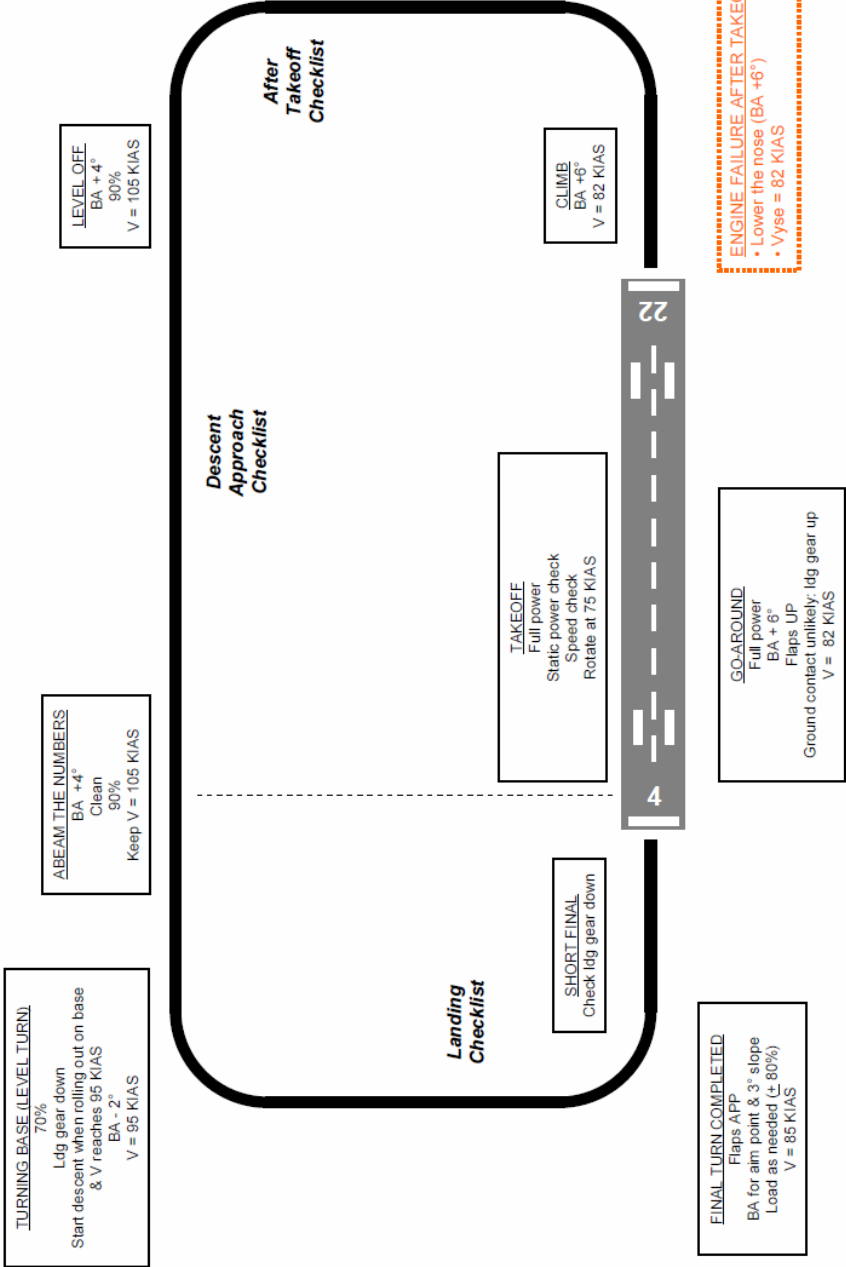
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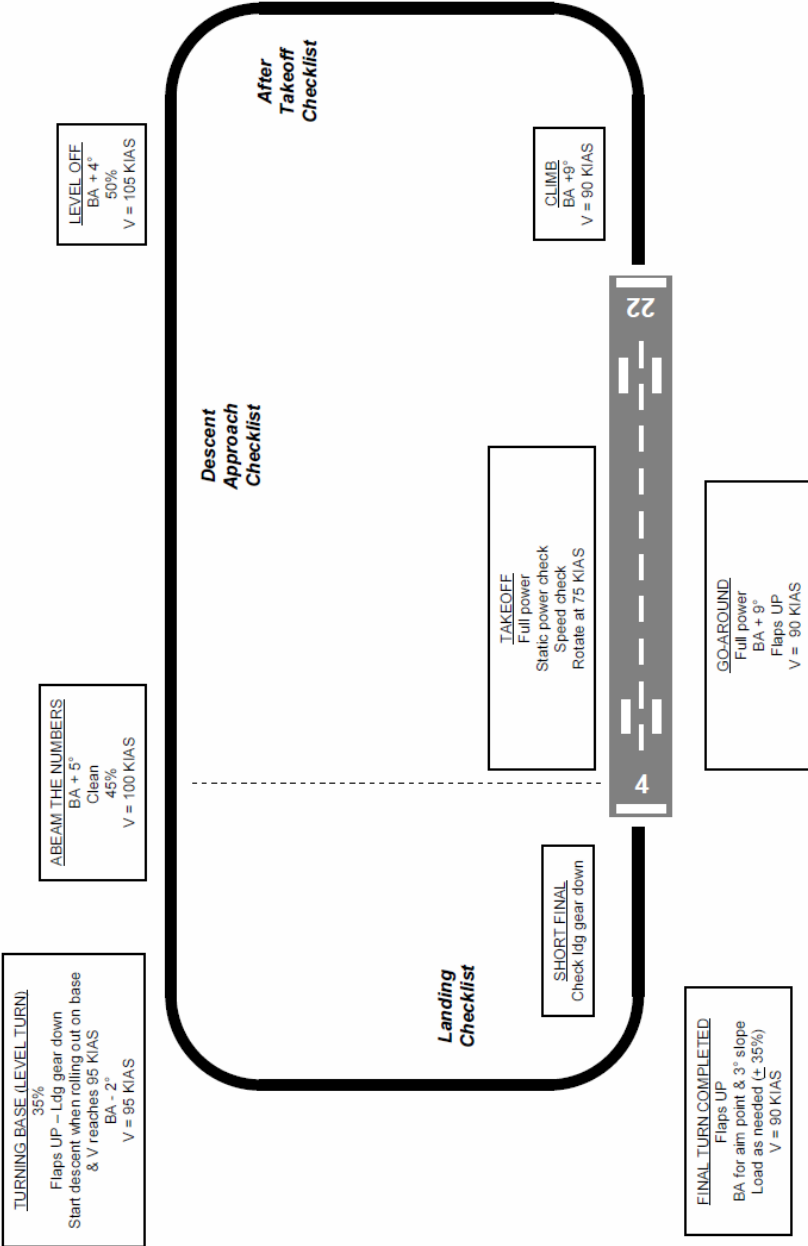
NORMAL



1 ENGINE



FLAPLESS



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EMER

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LIM

W&B

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STABILIZED APPROACH CONCEPT

- Configuration changes on short final (<500ft AGL) are not recommended to respect the stabilized approach concept. Exception is made if landing distance is a factor. If visual contact with the runway is established prior 500ft AGL LDG flaps may be selected, otherwise a flaps APP landing is recommended if sufficient rwy length is available.
- When becoming visual on an ILS approach, you must remain established on the Glide Slope (don't dive below the profile) until crossing the THR. For a Non-precision approach you must remain on the PAPI or if no PAPI installed fly a constant flight path to the aiming marks (blocks). Diving below the profile is only allowed if landing distance becomes a factor. This must be clearly briefed in advance.

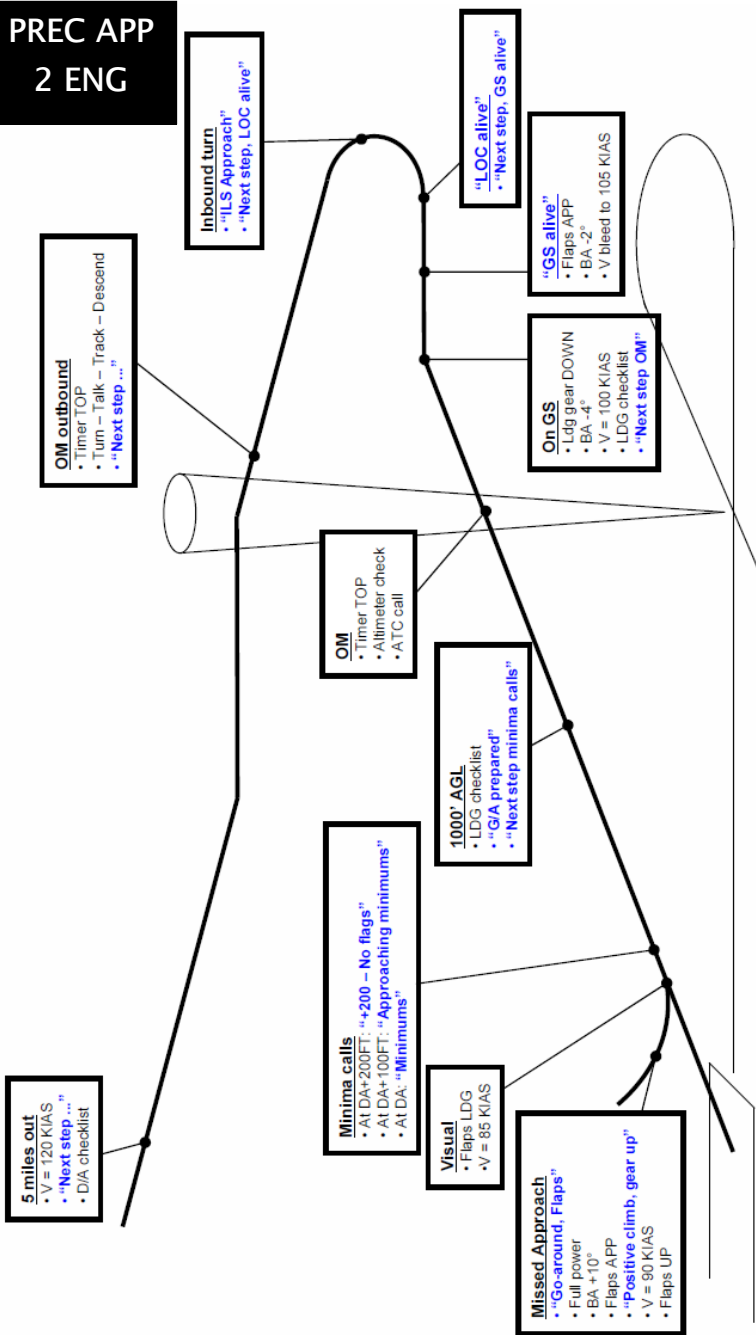
ALTITUDE BUG AND ALTITUDE WARNING

Altitude bug: gives a visual indication on the G1000 PFD.

Altitude warning: gives an aural warning when approaching (1000ft) the selected altitude on the AP.

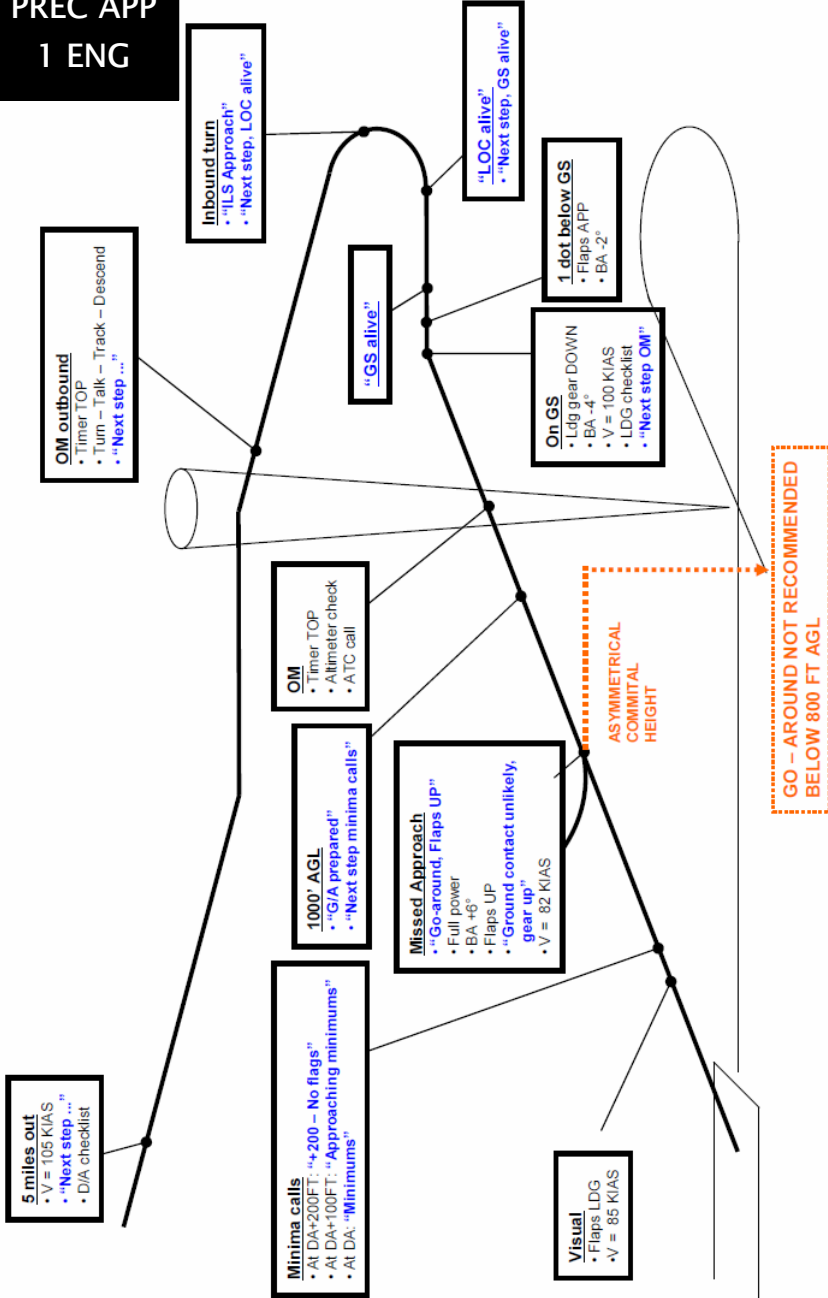
When established on GS or leaving the FAF altitude. Set Altitude bug on DA/MDA (rounded up 100ft) and the AP altitude on the go around altitude. This will give a visual warning when reaching the minima and an aural warning when approaching the level off altitude in the go around.

**PREC APP
2 ENG**

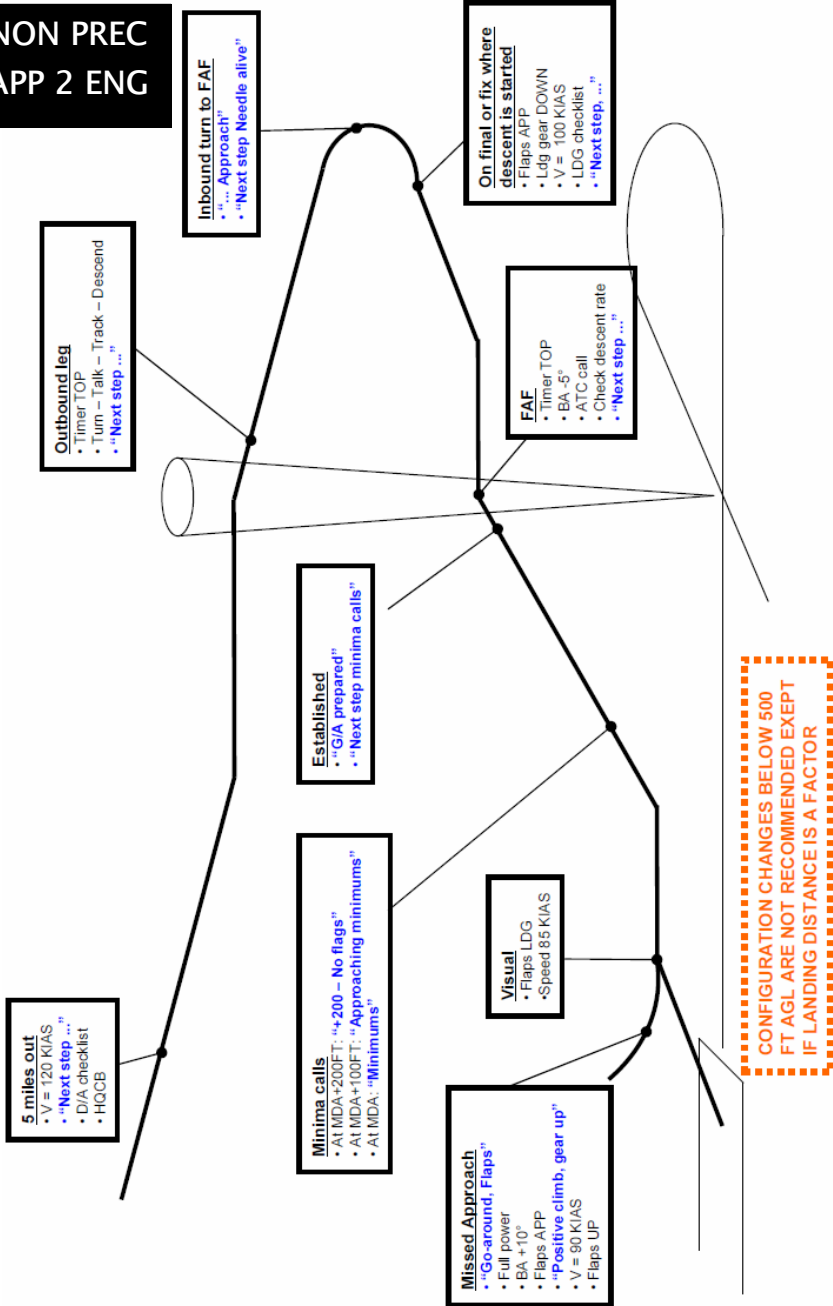


CONFIGURATION CHANGES BELOW 500 FT AGL ARE NOT RECOMMENDED EXCEPT IF LANDING DISTANCE IS A FACTOR

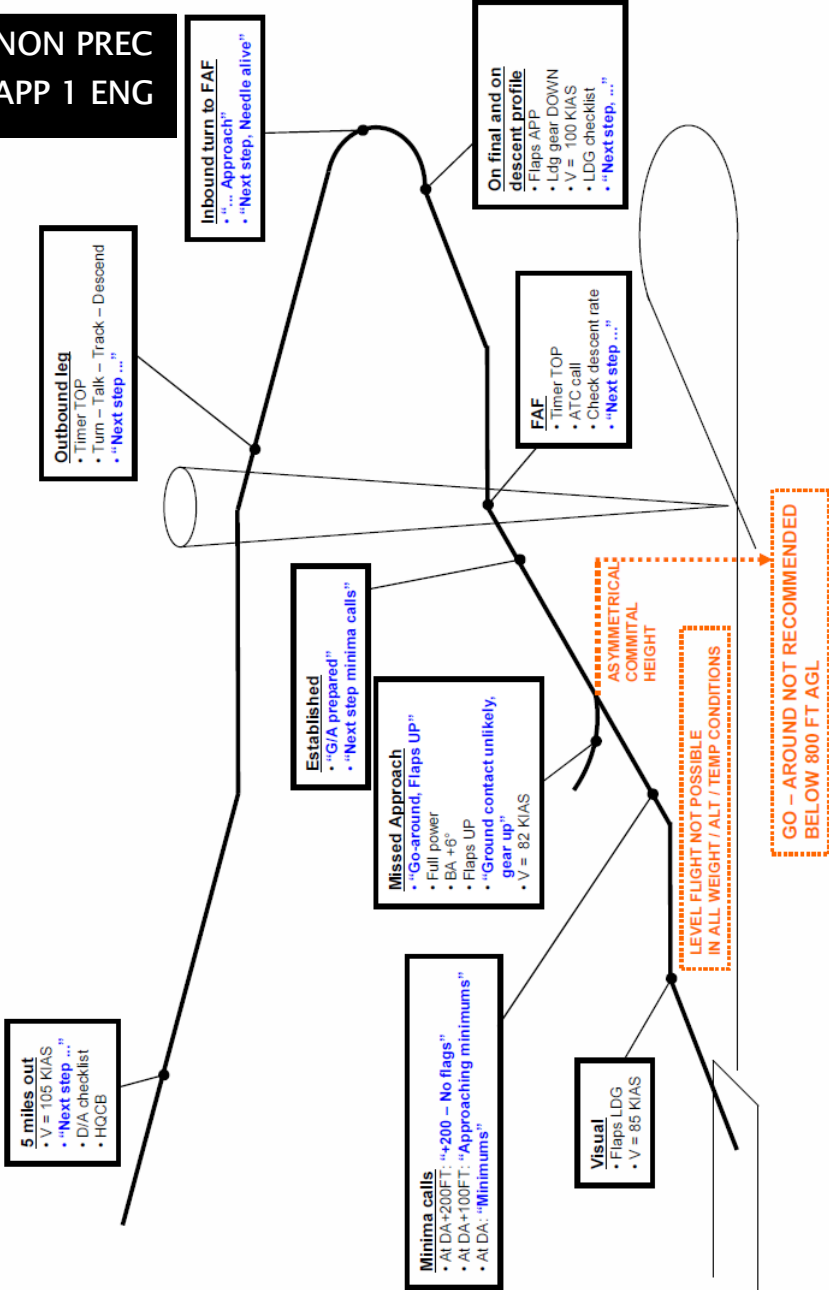
**PREC APP
1 ENG**



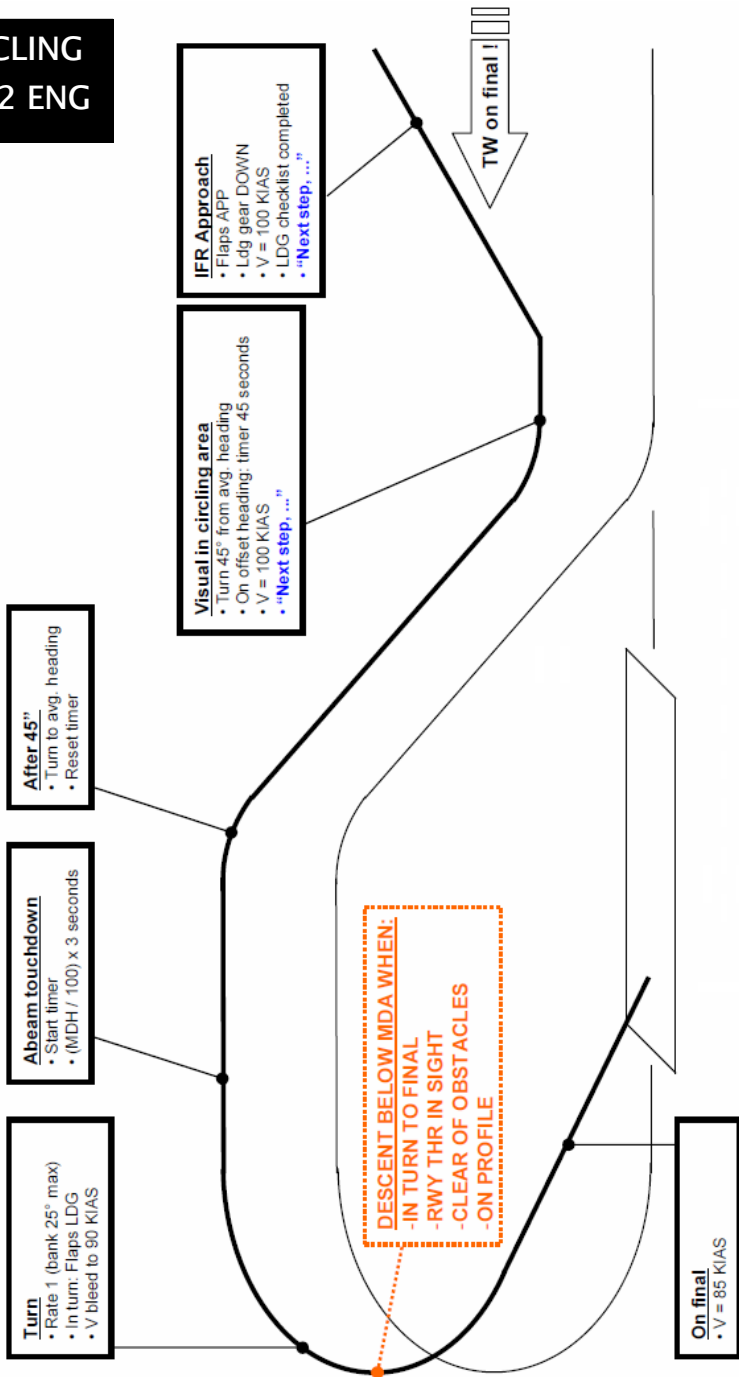
**NON PREC
APP 2 ENG**



**NON PREC
APP 1 ENG**



**CIRCLING
APP 2 ENG**



LIMITATIONS

PERF	W & B	LIM	PAT	EMER	NORM	MAN	PPT	TOC
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AIRSPEED LIMITATIONS

Speed Definition	Limitation		Indicator Marking
Never Exceed Speed (Vne)	194 KIAS	194 KCAS	Red radial line
Maximum Structural Cruising Speed (Vno)	155 KIAS	155 KCAS	End of green arc
Design Maneuvering			
Above 3.400 lbs.	126 KIAS	126 KCAS	
Below 3.400 lbs.	120 KIAS	120 KCAS	
Maximum Gear Operating Speed			
Extension	194 KIAS		
Retraction	156 KIAS		
Maximum Flaps Extended Speed	111 KIAS	111 KCAS	End of white arc
1 Engine Inoperative Best Rate of Climb Speed	82 KIAS	82 KCAS	Blue radial line
Air Minimum Control Speed	68 KIAS	68 KCAS	Red radial line

POWERPLANT LIMITATIONS		
Definition	Limitation	Indicator Marking
Rotational Speed	MAX 2300/2500 RPM	Red line
Fuel Temp	MAX 75°C MIN -30°C	
Oil Temperature	MAX 140°C	Red line
Oil Pressure	MIN 1.0 BAR MAX 6.5 BAR	Red line
Fuel Grade	JET A-1 NO ADDITIVES	

WEIGHT LIMITATIONS	
Max Ramp Weight	3.935 lbs
Max Takeoff Weight	3.935 lbs
Max Landing Weight	3.748 lbs

WEIGHT LIMITATIONS	
Max Weights in Baggage Compartments	
Forward (nose)	66 lbs
Aft (cockpit baggage compartment)	100 lbs
Max Zero Fuel Weight	3.638 lbs

MANEUVER LIMITS
All intentional acrobatic maneuvers (including spinning) are prohibited. Avoid abrupt maneuvers.

FLIGHT MANEUVERING LOAD FACTORS	
Positive Load Factor	
Flaps Up	Max 3.8G
Flaps Down	Max 2.0G
Negative Load Factor	No inverted maneuvers approved

FUEL LIMITATIONS	
Total Fuel Capacity	52 U.S. Gallons
Unusable Fuel	2 U.S. Gallons
Usable Fuel	50 U.S. Gallons

X-WIND LIMITATION
Maximum demonstrated crosswind component = 20 kts

WEIGHT & BALANCE

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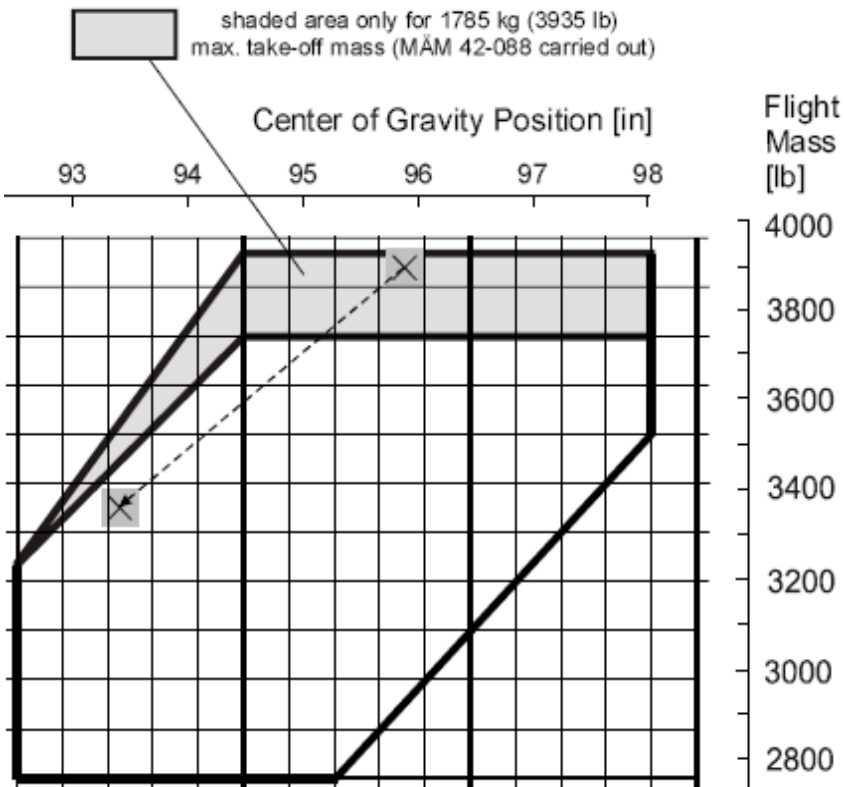


WEIGHT & BALANCE

QRH DA42
REV 1.2 — 7.1

ITEM	ARM (IN)	WEIGHT (LBS)	MOMENT (IN-LBS)
BEW			
PILOT & FRONT PASSENGER	90.6		
AFT PASSENGERS	128.0		
NOSE BAG COMP (MAX 66 LBS)	23.6		
COCKPIT BAG COMP (MAX 100 LBS)	153.1		
BAG EXTENSION (MAX 40 LBS)	178.7		
DE-ICING FLUID (1.1 KG/LTR OR 9.2 LBS/USG)	39.4		
ZFW (MAX 3638 LBS)			
FUEL IN MAIN TKS (50 USG MAX—7.01 LBS/USG)	103.5		
RW			
FUEL ALLOWANCE (START-TAXI)	103.5	8	828
TOW (MAX 3935 LBS)			
TRIP FUEL	103.5		
LW (MAX 3748LBS)			

Always check aircraft POH for latest weight & moments.



AIRCRAFT	WEIGHT (LBS)	MOMENT (IN-LBS)
N4129M	2844.80	270648.70
N4130M	2836.00	269154.80
N4197D	2833.10	270104.81
N449TS (AUX TANKS + DE-ICE)	2867.30	271666.30
OO—SFI (DE-ICE)	2837.00	269175.00
OO—SFA	2870.00	275347.80

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PERFO

PERF

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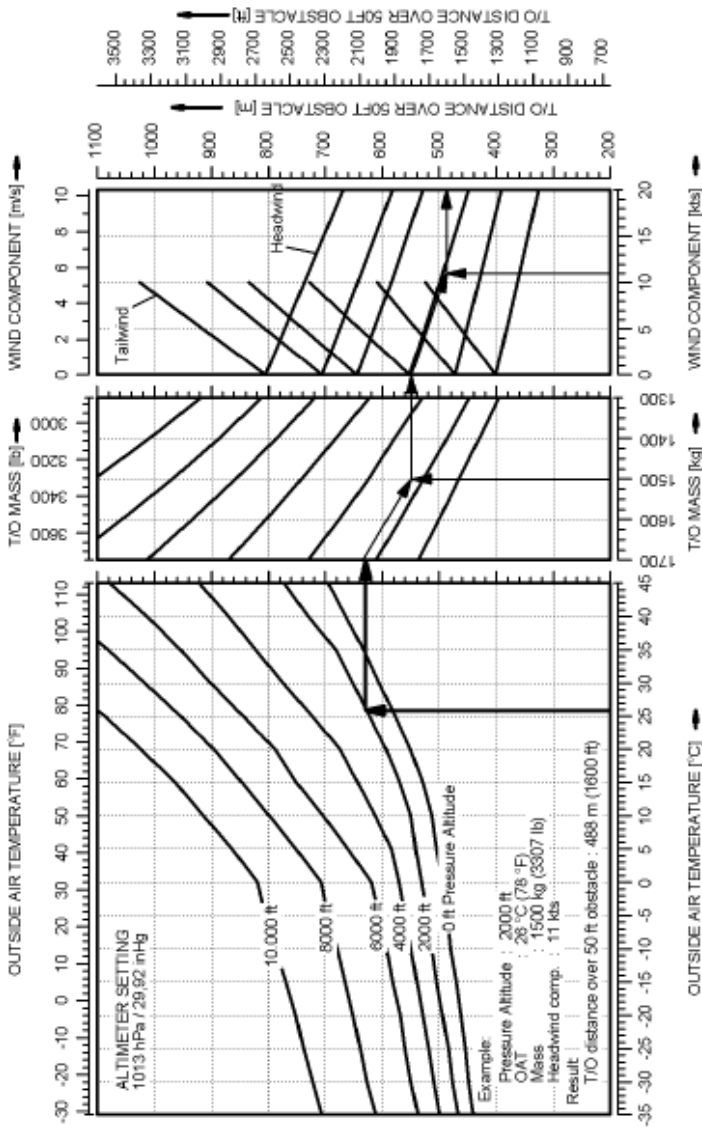
TOC

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TAKEOFF DISTANCE—50FT UP TO 1.700 KG	8.2
TAKEOFF DISTANCE—50FT ABOVE 1.700 KG.....	8.3
TAKEOFF CLIMB UP TO 1.700 KG	8.4
TAKEOFF CLIMB ABOVE 1.700 KG	8.5
1 ENG INOP CLIMB UP TO 1.700 KG	8.6
1 ENG INOP CLIMB ABOVE 1.700 KG	8.7
LANDING DISTANCE—50FT UP TO 1.700 KG	8.8
LANDING DISTANCE—50FT ABOVE 1.700 KG	8.9
CRUISING TAS.....	8.10

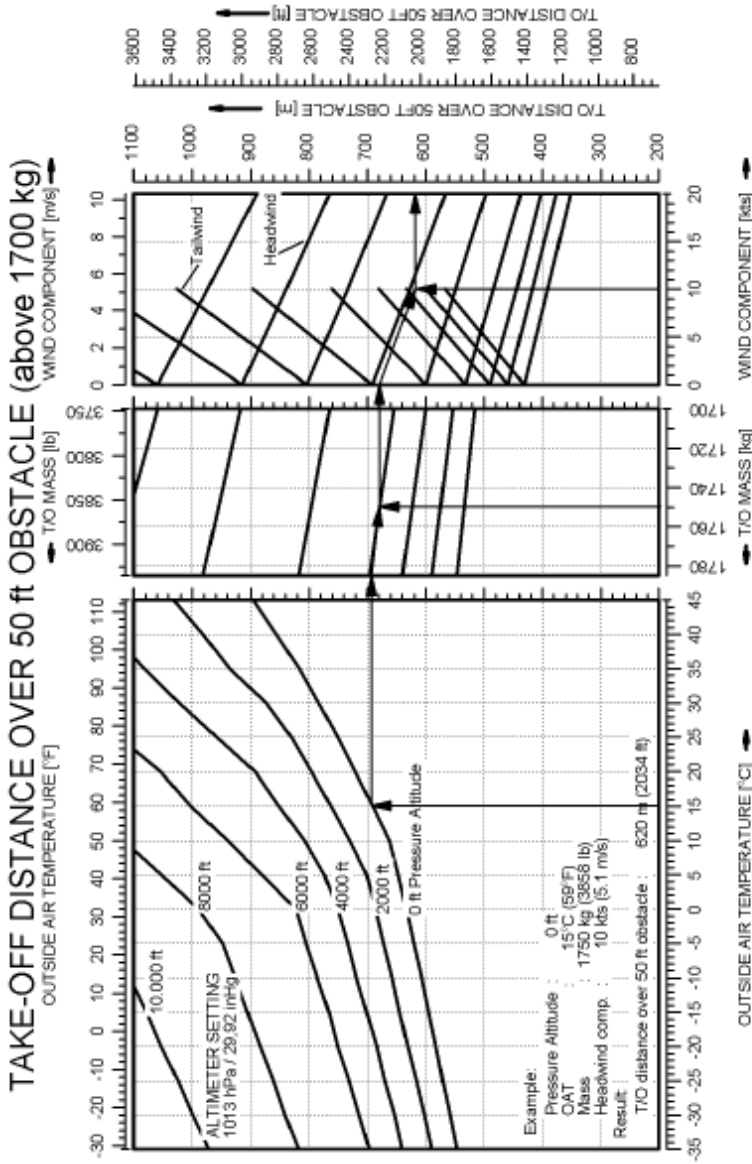
TAKEOFF DISTANCE
50FT—UP TO 1.700KG

TAKE-OFF DISTANCE OVER 50 ft OBSTACLE



Conditions: Flaps: UP, Power: both MAX @ 2300 RPM, v_r : 70 KIAS, Airspeed: 77 KIAS, Runway: hard, paved

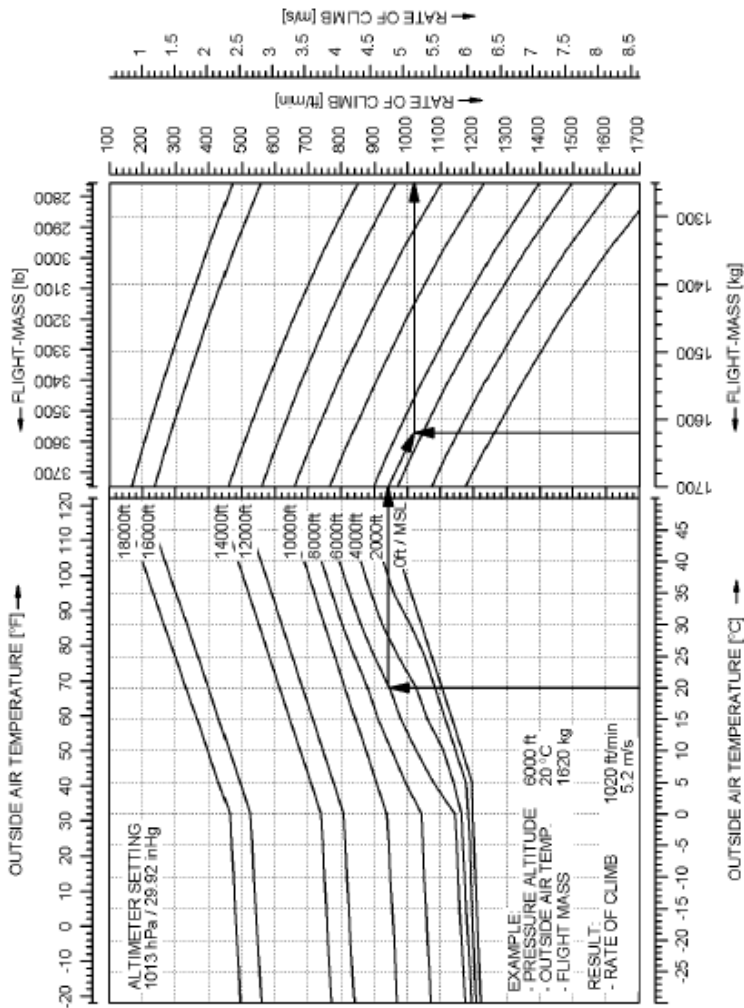
TAKEOFF DISTANCE
50FT—ABOVE 1.700KG



Conditions: Flaps: UP, Power: both MAX @ 2300 RPM, vr: 72 KIAS, Airspeed: 79 KIAS, Runway: hard, paved

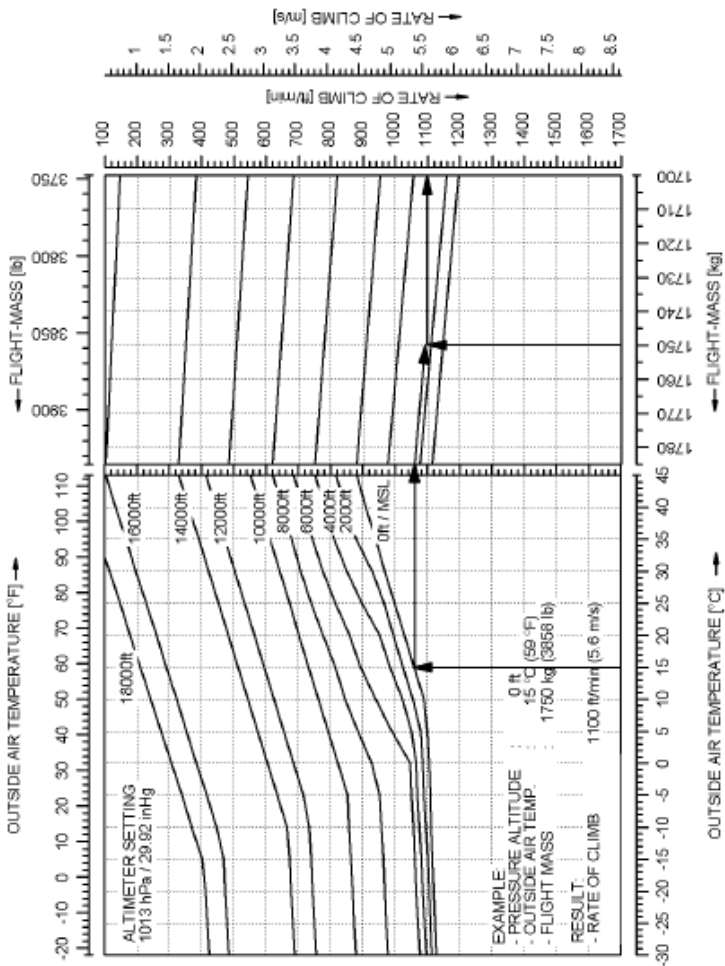
**TAKEOFF CLIMB
UP TO 1.700KG**

CLIMB PERFORMANCE - TAKE-OFF CLIMB



**TAKEOFF CLIMB
ABOVE 1.700KG**

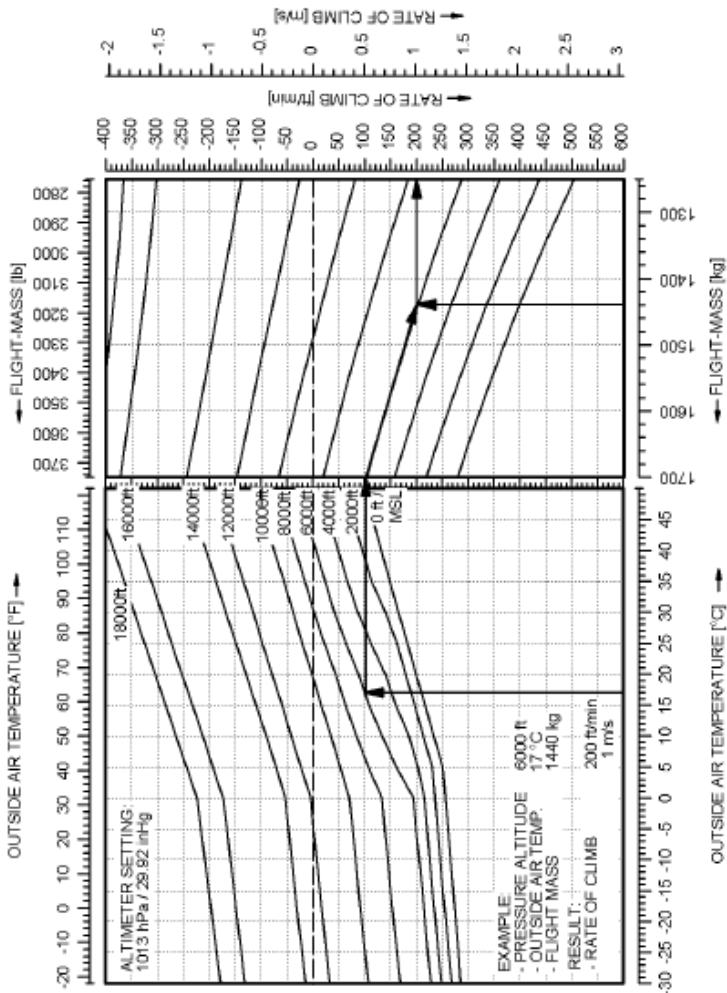
CLIMB PERFORMANCE - TAKE-OFF CLIMB (above 1700 kg)



Conditions: Flaps: UP, Power: both MAX @ 2300 RPM, Landing gear: retracted, Airspeed: 79 KIAS

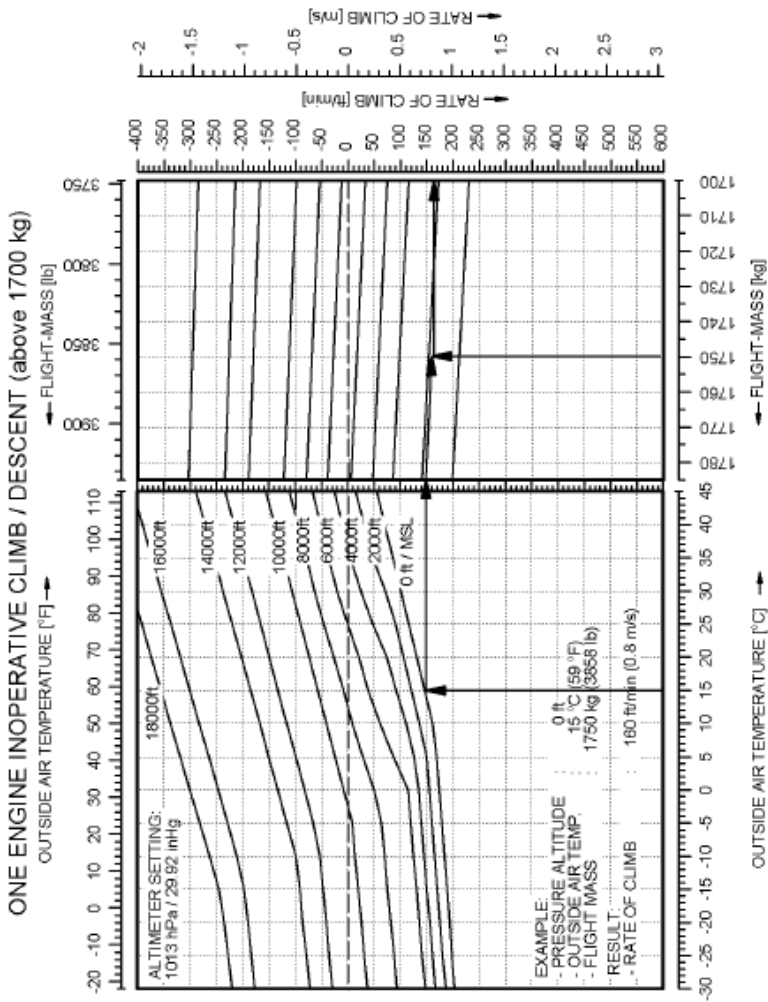
**1 ENG INOP CLIMB
UP TO 1.700KG**

ONE ENGINE INOPERATIVE CLIMB / DESCENT



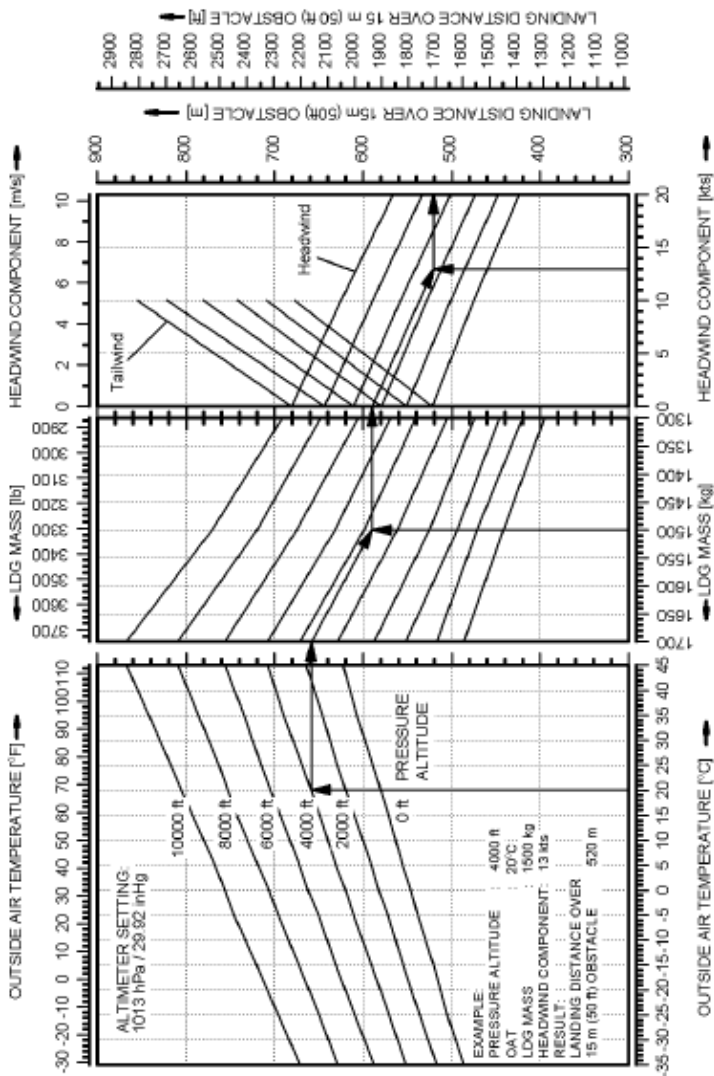
Conditions: Flaps: UP, Power: remaining engine MAX @ 2300 RPM / Dead engine: feathered+secured, Airspeed: 82 KIAS

**1 ENG INOP CLIMB
ABOVE 1.700KG**



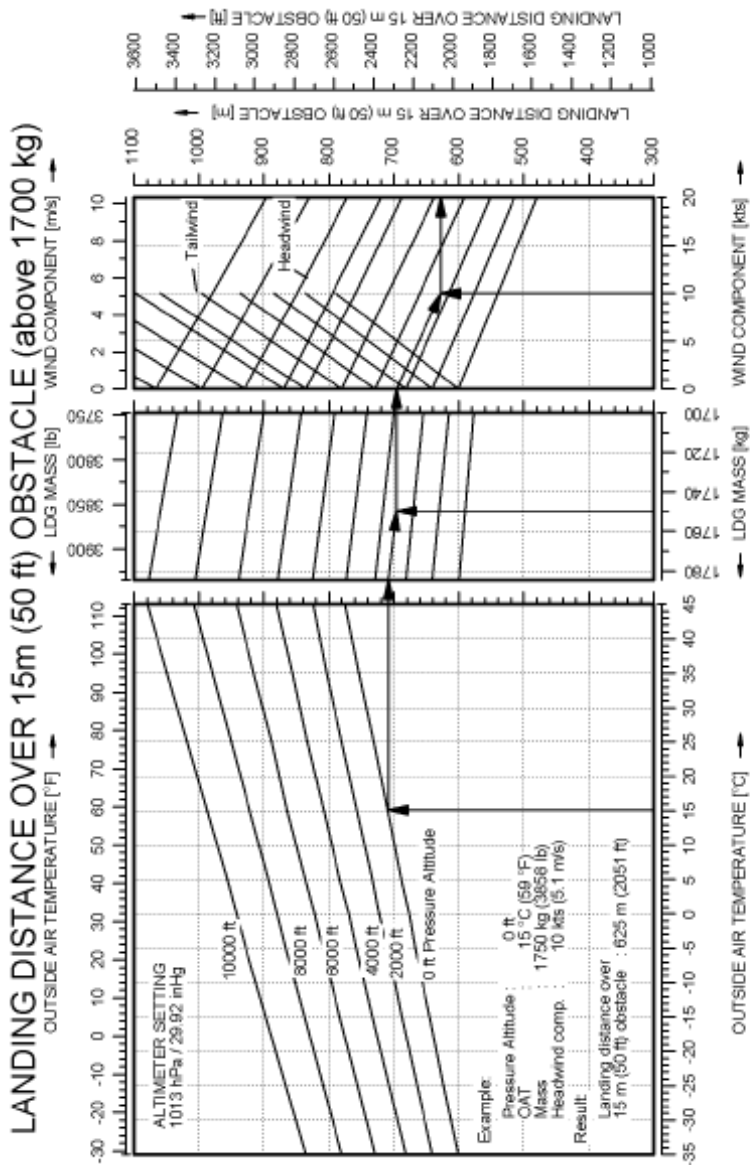
**LANDING DISTANCE
50FT—UP TO 1.700KG**

LANDING DISTANCE OVER 15 m (50 ft) OBSTACLE



Conditions: Flaps: LDG, Power: both IDLE, Runway: level and hard, dry surface, Approach Speed: 76 KIAS

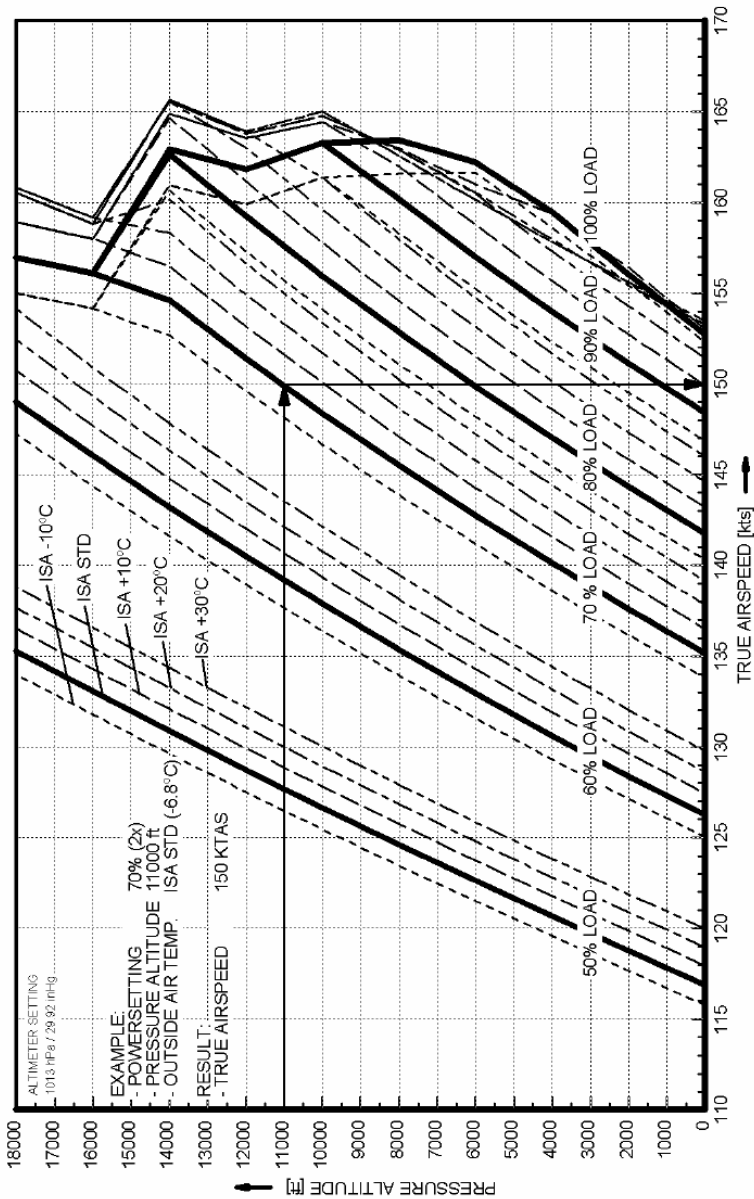
LANDING DISTANCE
50FT—ABOVE 1.700KG



Conditions: Flaps: LDG, Power : both IDLE, Runway : level and hard, dry surface, Approach Speed: 78 KIAS

CRUISING TAS

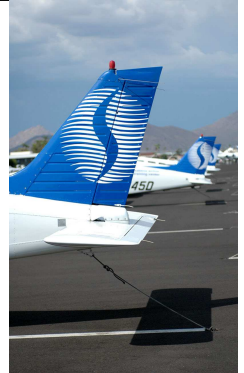
CRUISING (TRUE AIRSPEED / TAS)



Conditions: Engines: all operating, Power: as required to adjust selected LOAD [%], Flaps: UP, Landing Gear: retracted

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Full Throttle Flight Training



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