



# MSc/4th Year Aerospace Advanced CFD Laboratory Laminar Flow in tube bundles using Code\_Saturne

03/03/10

## **1** Introduction

Flows through tubes bundles are encountered in many heat exchanger applications. One example is in nuclear power plants where the nuclear fuel rods are cooled by passing cold water over them, but similar arrangements are also employed in more conventional exchangers. The objective of this type of heat exchanger is, of course, to extract as much heat as possible from the tubes. In a computational study, in order to calculate the heat transfer reliably, it is necessary to resolve the flow around the tubes correctly.

A typical staggered tube bundle configuration, which is to be studied in this exercise, can be seen in Figure 1. It is assumed to consist of a large number of tubes, so the flow around any one is considered to be identical to that around the others. The computational domain employed is highlighted, which takes advantage of some of the symmetries present in the flow. Symmetry conditions are applied on the top and bottom boundaries, whilst periodic conditions (with a prescribed pressure drop) are applied between the east and west boundaries.



Figure 1: Flow geometry

The flow is to be computed as laminar, and three meshes are provided. The first one consists of block structured quadrilateral cells with a resolution of 540 cells in total with 72 points around one cylinder. The second mesh is an unstructured mesh made of quadrilateral cells (PAVE). The total number of cells is 433 with 64 points around one cylinder. The third mesh (TETRA) is mader of triagular elements, with a total number of 490 cells with 64 points around one cylinder. The meshes can be seen in Figure 2.



Figure 2: Three types of meshes

The use of a structured mesh means that there are rather highly skewed cells at the interface between the curved edge of the cylinder and the straight symmetry edge. The unstructured approach deals with this by removing the constraint of the cell edges having to lie along the same lines as the block boundaries.

By using these relatively coarse meshes, the influence of the skewness of the grid can be seen. Although the error introduced can be reduced by refining the mesh, the number of cells required to reduce the error significantly is different for each meshing technique. This means that before doing a mesh refinement study, it is often worth spending some effort in achieving a grid structure with the least distortion possible.

A converged solution on a very fine grid is also available, allowing comparisons to be drawn between these grid-independent results and those obtained on the three coarser grids described above.

#### 2 Objectives

One of the objectives of this exercise is to compare the three different types of meshing techniques described above for a complex curved geometry, and assess their relative strengths and weaknesses in terms of solution accuracy.

A second objective is to gain more experience in using a wider range of CFD and other software tools for simulations and post-processing; in particular here using the *Code\_Saturne* program to perform the flow simulations.

### **3** Investigations

First, follow the instructions given below to run the case using the block-structured grid (BLOCK), and post-process the results. Then, follow the same steps to compute and examine the solution for the other two grids.

Comparisons should be drawn between the computed solutions on the different grids, and the grid-independent one from the provided fine grid solution. Qualitative comparisons can be made by examining velocity vector plots, contour plots of velocity or pressure, and streamlines. For quantitative output, useful comparisons include

- U-velocity component profiles along the periodic boundary (x = 0) and symmetry line (y = 0).
- The flow recirculation length, deduced from the velocity profile along y = 0.
- The computed mass flow rate, conveniently expressed non-dimensionally as the Reynolds number based on tube diameter and bulk velocity ( $Re = U_b d/\nu$ ).
- Compute the pressure drop coefficient  $K_p$  as in:

$$\Delta P = K_p \frac{1}{2} \rho U_b^2 \tag{1}$$

The pressure drop  $\Delta P/L = 10^{-5}$  has been set in the user subroutine *ustsns.f90* as a forcing term  $F = \Delta P/L$  where L is the distance between the periodic faces.

#### **OPTIONAL:**

You could also investigate how the pressure drop coefficient  $K_p$  changes with Re number. By systematically increasing the viscosity (multiply by 2, 4, 8 ...) you can decrease Re while keeping the same  $\Delta P$ . Calculate the quantity  $K_pRe$  and see if it remains constant. How does the recirculation region change with Re?

#### **4** Reporting Requirements

Students should submit a short report within 2 weeks of the laboratory. The report should contain a brief introductory section, outlining the nature of the investigations conducted, and should then describe and discuss (with the aid of a *selection* of figures) the results obtained and comparisons between the various solutions. Excluding tables and figures this is unlikely to require more than 4–5 pages of text.

#### 5 Code\_Saturne

*Code\_Saturne* is a system designed to solve the Navier-Stokes equations in the cases of 2D, 2D axisymmetric or 3D flows. Its main module is designed for the simulation of flows which may be steady or unsteady, laminar or turbulent, incompressible or potentially dilatable, isothermal or not. The code employs a finite volume discretisation and allows the use of various mesh types, including hybrid arrangements (where the mesh contains several kinds of elements) and structural non-conformities (hanging nodes).

The code includes specific modules, referred to as "specific physics", for the treatment of Lagrangian particle tracking, semi-transparent radiative transfer, gas, pulverised coal and heavy fuel oil combustion, electricity effects (Joule effect and electric arcs), compressible flows, but these are not required for the present exercise.

*Code\_Saturne* is free software; it can be downloaded, modified and/or redistributed under the terms of the GNU General Public License as published by the Free Software Foundation.

The notes below guide you through the steps required to prepare and run the present cases in *Code\_Saturne* version 2.0, and subsequently post-process the results using the open-source post-processing program *ParaView*.

The meshes and data necessary for the exercise are available from the Manchester CFD and Turbulence Modelling website at:

http://cfd.mace.manchester.ac.uk/Main/TubeBundleTutorial.

Further information on *Code\_Saturne* can be found in the user manual (accessed by typing code\_saturne info -g user at the prompt). More details of *ParaView* can be obtained from the website http://www.paraview.org.

#### 6 Guide to Running Code\_Saturne

Before starting the Graphical Interface on *Code\_Saturne*, it is necessary to create the appropriate file structure for the code to work.

In order to allow all the scripts of *Code\_Saturne* to work, you will need to add the path to the directory where the *Code\_Saturne* installation is located. Open a console (Applications - Accessories - Terminal) and type:<sup>1</sup>

[bash:\$] export PATH=\$PATH:/opt/cs-2.0-rc2/bin

(Note: [bash:\$] in this document represents the prompt. It might be different, depending on your GNU-Linux installation and your profile file. There is no need to type this.)

The above command will load all the variables required (note the dot and space (.) at the beginning of the command line).

<sup>&</sup>lt;sup>1</sup>In a command line interpreter, such as bash, autocomplete of command names and file names may be accomplished by keeping track of all the possible names of things the user may access. Here autocomplete is usually done by pressing the Tab key after typing the first several letters of the word. For example, if the only file in the current directory that starts with x is xLongFileName, the user may prefer to type x and autocomplete to the complete name. If there were another file name or command starting with x in the same scope, usually the user would have to type some more letters or press the Tab key repeatedly to disambiguate what he or she means to the computer.

If everything went OK, you should get the following information when typing code\_saturne:

```
[bash:$] code_saturne
Usage: /opt/cs-2.0-rc2/bin/code_saturne <topic>
Topics:
    help
    check_consistency
    check_mesh
    compile
    config
    create
    gui
    info
    plot_probes
Options:
    -h, --help show this help message and exit
```

Now that you have your PATH updated, you can use the case creator script. You will need about 150MB of free space, if your home directory has not enough space, use the temporary directory on the machines, but be aware that you will need to copy your data to other media as everything under /tmp will be deleted after reboot. Go to the desired location (cd /tmp/ for example) and type:

[bash:\$] code\_saturne create --study TUBE\_BUNDLE --case BLOCK

This command will create a study called TUBE\_BUNDLE and a case called BLOCK. The file structure can be seen in Figure 3

The details of the structure are:

- BLOCK: This is the case file. A study can have several cases, for example each using different turbulence models, different boundary conditions, etc.
  - DATA: This is where the restarting files need to be copied, it also contains the script to launch the Graphical Interface SaturneGUI.
  - SRC: This is where the user subroutines should be saved. During the compilation stage the code will re-compile all the subroutines that are in this directory. In the REFERENCE directory all the user-defined subroutines are stored. For a specific case you might only need a few of these, depending on what is needed to implement the case. In this case we will make use of one such routine, in order to prescribe a streamwise pressure drop between the periodic boundary planes. The base directory contains the basic flow solver subroutines, the other directories contain the subroutines needed by other modules.
  - RESU: Directory where the results will be exported once the calculation has finished.
  - SCRIPTS: This is where the launching scripts are copied. You will need to run the executable file *runcase* to start the calculation.



Figure 3: File structure created by Code\_Saturne

- MESH: The program will read the mesh from this directory. The mesh formats that can be read are I-DEAS, CGNS, Gambit Neutral files (neu), EnSight, pro-star/STAR4 (ngeom), Gmsh, NUMECA Hex, MED, Simail and Meta-mesh files.
- POST: This is an empty directory designed to contain post-processing macros that may be relevant for particular cases.

You will need to copy the mesh into the MESH directory. The mesh can be found on the website: http://cfd.mace.manchester.ac.uk/Main/TubeBundleTutorial (This web address is case sensitive)

Download the three meshes into the MESH directory. If your browser does not ask you for a place to save the files try right clicking on the lick and select 'Save link as'.

Because the case is periodic in the streamwise direction, it is necessary to impose a pressure gradient to make the flow move. To do this, download the file called *ustsns.f90* from the above webpage and save it into the SRC directory. This file is a user subroutine that adds source terms into the momentum equation. The file has already been edited for this tutorial it is not necessary to change it.

Now, you can use the graphical interface of *Code\_Saturne* to complete the case set up. To launch the graphical interface go to the DATA directory and launch the SaturneGUI script by typing:

```
[bash:$] cd TUBE_BUNDLE/BLOCK/DATA
[bash:$] ./SaturneGUI &
```

Laminar flow in tube bundles

The graphical interface offers different folders that can be expanded by double clicking on them (see Figure 4). On the pages you can set all the values needed to run a simulation. Once the Graphical Interface is up and running, create a new case (File -New) and the values of *Identity and paths* should be filled in automatically. Open on the *Calculation environment* folder and click on the *Mesh selection* page. The meshes that you have downloaded into the MESH directory should appear in the box. Delete the meshes that you are not going to use (i.e. leave only ROD\_BUNDLE\_BLOCK.neu)

🛞 💿 🙆 New parameters set - Code_Saturne Gl	Л		0 🔀
<u>File Tools Window H</u> elp			
i 🗖 🔚 🖄 😵 📭			<u>a</u>
			• •
Study: TUBE_BUNDLE			
Case: BLOCK			
XML file:			
● ⊗ N	eshes Periodic Boundaries	948). 	
- El Calculation environment	List of meshes		
Meshes selection	Meshes	Format	
Thermophysical models	ROD_BUNDLE_BLOCK.neu	GAMBIT Neutral ".neu"	
+ Physical properties			
+ P Volume conditions			
+ P Numerical parameters			
Calculation control		Dalata	
+- e Calculation management	Aug Delete		
	Join meshes		
	Subdivide warped faces		
l l	Correct cell and face orientations		
			)

Figure 4: Mesh Selection

Click on the periodic boundary conditions tab and set them up as a translation set. Click on the Add button. A new panel will appear and in the List of boundary faces for the selected periodicity click on New. Double click on the Groups box and type the groups INLET and OUTLET Separated by a space. The translation vector is TX = 2, TY = 0, TZ = 0. See Figure 5.



Figure 5: Periodic boundary conditions

Next, set the *Thermophysical models*. Click on the *Calculation features* page and select *Steady Flow*. Click on the *Turbulence model* page and select no model, i.e. laminar flow.

In the *Physical properties* folder, make sure the default properties of air are checked, i.e.  $\rho = 1.17862 \text{ kg/m}^3$ ,  $\mu = 1.83 \times 10^{-5} \text{ Ns/m}^2$ .

Next, expand the *Boundary Conditions* folder and click on the *Definition of boundary regions*. Create the regions as shown in figure 6. Make sure there are no typos on the localisation of the faces.

Then set the faces WALLR1, WALLR2 and WALLR3 as walls and the other as symmetry by clicking on each region and changing the *Nature* box, then click *Modify*.

Next, expand the *Numerical parameters* folder and in the *Steady management* page set the number of iterations to 2000.

Finally, expand the *Calculation management* folder and click on the *Prepare batch* calculation page. Click on the *Select the batch script file* button to select the runcase file.

The above step will open new dialogues and give you the button to run the case, labelled 'Code\_Saturne batch running'.

Important: Save the file as block.xml before continuing.

Click on the button and watch the output on the console screen. It tells you where the temporary directory is located. The temporary directory is useful to see the progress of the calculation. The calculation should take about two minutes on a Intel 2.1 GHz processor. If everything is correct, you should see the following message:



Figure 6: Boundary Conditions

Once the calculation is finished, close the Graphical Interface and continue with the post processing.

#### 7 Post-Processing

Once the calculation of *Code\_Saturne* is finished, the results are copied into the RESU directory with the date appended. The listpre file contains the output of the pre-processor which reads the mesh and passes the information to the solver kernel. The listing file has information about all the parameters of the calculation and useful data about the convergence at each time-step. The HIST directory contains ASCII files with the history of monitored values for each variable. The CHR.ENSIGHT directory has the results files in EnSight format. These files can be viewed with *ParaView* directly. Additionally, a copy of the FORT directory and the runcase script used for the calculation are saved.

In order to visualise the results, go into the results directory by typing:

[bash:\$] cd ../RESU/CHR.ENSIGHT.XXX'

where XXX is the date. Once inside the directory type:

[bash:\$] /opt/paraview-3.6.1/bin/paraview --data=CHR.case &

This will bring up the *ParaView* interface with the case data ready to be loaded. click on apply to load the data.



Figure 7: ParaView initial window

By default the domain will be coloured by pressure contours. The variable used can be changed via the top left corner box. The colouring scheme can also be set to a more standard 'blue-to-red' scheme by clicking on the *Edit color map* button.

A quantitative way of comparing the results is to plot velocity profiles along particular lines. For this exercise you can use the bottom symmetry line at y = 0 and the periodic line at x = 0.

To plot a curve along the symmetry line, go to *Filters - Alphabetical - PlotOverLine*. Select the points (0.5, 0, 0.2) and (1.5, 0, 0.2) and the resolution to 20 points, then click 'Apply'. A new window should appear with an XY plot of all variables. To select only the streamwise velocity component, click on the 'Display' tab and tick only 'VelocityU:X' in the list of variables.

To create another XY plot on the x = 0 plane, select the CHR.case box on the panel on the left and repeat the procedure, but now using the points (0, 0.5, 0.2) and (0, 1, 0.2) and a resolution of 10 points. Before clicking 'Apply' make sure that you have the window with the full domain selected (a red square around the window highlights the selected one). If you have the window with the first XY plot selected, both lines will appear in the same window. In the end your solution should be similar to what is shown in Figure 8 If you want to export the values of a given line, selected from the pipeline browser panel on the left and click on *File - Save Data*. This will open a new window where you can select the name and the format. To save as text, use the .csv format (Comma Separated Values). This values can be imported into OpenOffice Calc (Similar to Microsoft Excel) or viewed with a text editor such as gedit or emacs.



Figure 8: ParaView results

Many other postprocessing tools are available from ParaView. Some of these (such as vectors or streamlines) require that the data be interpolated from the cell centres to the nodes. To do this go to *Filters - Alphabetical -CellToDataPoint*. This will produce a new data set with all the variables interpolated to the cell nodes, and other filters will then be available.

**NOTE:** If you want to save a screenshot, you will need to change the default settings to avoid paraview crashing. Go to *Edit - Settings*. Click on the "Render View" in the left pane and then uncheck "Use offscreen for screenshots". After you have changed the setting you can save a screen shot by going to *File - Save screen shot*.

# 8 Fine Grid Solutions

After post processing the results for the BLOCK case, the same procedure can be applied to the other two meshes, i.e. PAVE and TETRA. Create a new case for each mesh and follow the same instructions. To move back a directory, use the cd . . command. To list the contents of a directory type ls and to get the current working directory use pwd. Once you are inside the TUBE\_BUNDLE directory you can use the command

[bash:\$] code\_saturne create --case PAVE --case TETRA

to create two new cases inside the same study called PAVE and TETRA. After the cases are created, you can proceed to open the graphical interface and fill in the values of each case.

Laminar flow in tube bundles

After the results for the three meshes have been obtained, they can be compared with the reference data available on the website. These results were obtained with a mesh of 30180 cells, and can be loaded into *ParaView* for comparison with the coarse mesh results obtained during the exercise.