PLANAR LASER-INDUCED FLUORESCENCE (PLIF)

OF

$H_2 - O_2 \ COMBUSTION$

The members of the Committee approve the master's thesis of Takashi Yokomae

Frank K. Lu Supervising Professor

Donald R. Wilson

Seiichi Nomura

Copyright © by Takashi Yokomae 2003

All Rights Reserved

PLANAR LASER-INDUCED FLUORESCENCE (PLIF)

OF

$H_2 - O_2$ COMBUSTION

by

TAKASHI YOKOMAE

Presented to the Faculty of the Graduate School of

The University of Texas at Arlington in Partial Fulfillment

of the Requirements

for the Degree of

MASTER OF SCIENCE IN AEROSPACE ENGINEERING

THE UNIVERSITY OF TEXAS AT ARLINGTON

August 2003

ACKNOWLEDGEMENTS

I wish to express my appreciation to my supervising professor, Dr. Frank Lu, for giving me an opportunity to work on this research at the Aerodynamics Research Center (ARC), for being patient with my slow progress and for his technical guidance.

I would also like to thank Drs. Wilson and Nomura for being members of my thesis committee.

Special thanks go to Qiong (Joan) Wei, who has shared her knowledge with me.

I really appreciate those individuals who have assisted me at the ARC: Rod Duke, Jason Meyers, Chris Roseberry and Philip Panicker, just to name a few.

I am grateful to my mother, Tsugiko, for financial and emotional support from home back in Japan.

I would also like to acknowledge the teaching assistantship for the first year and the graduate fellowship for the two years of my graduate study granted by the Mechanical and Aerospace Engineering Department at the University of Texas at Arlington.

August 18, 2003

ABSTRACT

PLANAR LASER-INDUCED FLUORESCENCE (PLIF)

OF

 $H_2 - O_2$ COMBUSTION

Publication No.

Takashi Yokomae, M.S.

The University of Texas at Arlington, 2003

Supervising Professor: Frank K. Lu

The fluorescence in the 297-nm wavelength in H_2 -O₂ burning flame is imaged. A laser sheet is generated when the 248 nm KrF excimer laser beam passes through a cylindrical lens. Hydroxyl (OH) radical is excited by the laser sheet across the H_2 -O₂ flame produced by a flat flame burner. The excited radical emits fluorescence, which is captured by the ICCD camera. However, the fluorescence signal depends on temperature, as well as OH concentration. In the present work, only fluorescence images are obtained of the combustion process. Further work is needed to distinguish between concentration and temperature effects in the fluorescence images.

TABLE OF CONTENTS

ACKNOWLEDGEMENTS	iv
ABSTRACT	v
LIST OF ILLUSTRATIONS	viii
LIST OF TABLES	ix
NOMENCLATURE	x
Chapter	
1. INTRODUCTION	1
1.1 Planar Laser-Induced Fluorescence	1
1.1.1 Laser-Induced Fluorescence	1
1.1.2 Planar Laser-Induced Fluorescence	3
1.1.3 Fluorescence Dependence of Temperature and Mole Fraction	4
1.1.4 Disadvantages of PLIF	5
1.2 Objective	5
2. EXPERIMENTAL SET-UP	7
2.1 KrF Excimer Laser	7
2.2 Combustion System	9
2.2.1 Flat-Flame Burner	9
2.2.2 Combustion Gas Handling System	10

2.3 Optical System	13
2.3.1 Spatial Filtering	13
2.3.2 Laser Sheet Generation	14
2.3.3 Image Collection	15
2.4 ICCD (Intensified Charged Coupled Device)	15
2.5 Image Acquisition and Processing	17
2.5.1 Image Collection	17
2.5.2 Image Processing	18
3. RESULTS AND DISCUSSION	20
3.1 Results	20
3.2 Discussion	27
4. CONCLUSIONS AND RECOMMENDATIONS	29
4.1 Conclusions	29
4.2 Recommendations	29
Appendix	
A. EXPERIMENTAL PROCEDURE	31
B. COMPEX 150 EXCIMER LASER OPERATION PROCEDURE	35
C. WINVIEW/32 PROCEDURE	41
REFERENCES	45
BIOGRAPHICAL INFORMATION	47

LIST OF ILLUSTRATIONS

Figure		Page
1.1	State View of Laser Induced Fluorescence	2
1.2	Basic Arrangement of PLIF	4
2.1	PLIF setup	7
2.2	Sectional View of McKenna Flat Flame Burner	9
2.3	Schematic Diagram of Experimental Setup for the Flat Flame Burner	11
2.4	Gas Handling System	12
2.5	Spatial Filters	14
2.6	ICCD Test with 3-D Garfield	17
2.7	Image Acquisition System Configuration	18
3.1	An Example of WinView/32 Display	23
3.2	Averaging of Five Images	24
3.3	PLIF Images of Six Cases without Ceramic Rod	25
3.4	PLIF Images of Six Cases with Ceramic Rod	26
3.5	Position of Ceramic Rod	27

LIST OF TABLES

Table		Page
3.1	Results	21
3.2	Correlation Chart for Flowmeters	22

NOMENCLATURE

d	diameter of a lens
f	focal length of a lens
$f_B(T)$	temperature-dependent Boltzmann fraction of the absorbing state
FWHM	full-width at half-maximum
HBW	half bandwidth
ICCD	intensified charge-coupled device
KrF	krypton fluoride
LED	laser emitting diode
\dot{m}_{O_2}	mass flow rate of O ₂
\dot{m}_{H_2}	mass flow rate of H ₂
S _{pp}	fluorescence signal per pixel
Т	temperature
UV	ultra violet
VUV	vacuum ultra violet
Хон	OH mole fraction

CHAPTER I

INTRODUCTION

As lasers become increasingly available and reliable, laser spectroscopy becomes promising in the diagnostic probing of combustion processes. Laser-based techniques are capable of remote, non-intrusive, in-situ, spatially and temporally precise measurements of chemical parameters, unlike physical probing methods such as thermocouples that have been traditionally utilized to investigate and characterize combustion phenomena [1]. Laser-induced fluorescence is presumably the most wellknown technique for radical species measurements. It has been in use considerably since the early 1980s because of its merits of providing high spatial resolution (typically 0.1 mm), high temporal resolution (typically less than 100 ns), and high sensitivity (typically concentrations in the ppm range) [2].

1.1 Planar Laser-Induced Fluorescence

1.1.1 Laser-Induced Fluorescence

Laser-induced fluorescence (LIF) is an established, selective and sensitive approach for identifying species concentration from reactive-flow systems without perturbing the system under study [3-5]. LIF is a sequence of molecules or atoms being excited to higher electronic energy states via laser absorption followed by spontaneous emission of fluorescence. The spectral absorption regions are discrete, because the energy states of molecules and atoms are quantized. Typically, fluorescence occurs at wavelengths greater than or equal to the laser wavelength. Therefore, LIF offers the possibility to investigate species of interest by selecting the appropriate wavelength.

Shown in Fig. 1.1 below is the state view of LIF. As the species under investigation are usually molecules or radicals, not atoms, the electronic levels are divided into sub-levels according to the molecular vibrational and rotational energy. Due to quantum mechanical effects, electronic, vibrational and rotational energies are quantized (i.e., 0, 1, 2, ...) [6].

Figure 1.1 also shows energy state description of LIF. The molecules or radicals are excited by laser absorption to the metastable state, and then drop down to the stable ground state, emitting fluorescence.

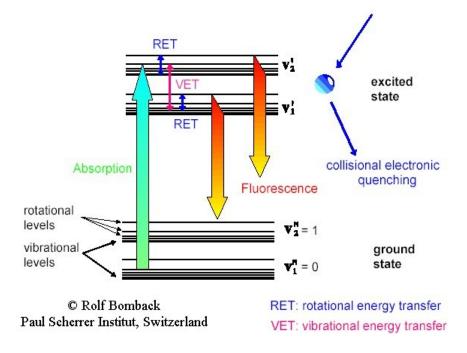


Figure 1.1: State View of Laser Induced Fluorescence

1.1.2 Planar Laser-Induced Fluorescence

The planar laser-induced fluorescence (PLIF) is a derivative of the LIF technique, which necessitates the generation of a laser sheet from the probe laser beam in order to facilitate imaging of the fluorescence. PLIF involves illuminating the flow with a thin sheet of laser light tuned to excite electronic transitions in a chemical species in the flow. The OH radical is accessed in this experiment. A radical is an atom or a group of atoms that is a usually very reactive species, which contains one or more unpaired electrons. The fluorescence emitted by the laser excitation is focused onto an intensified charge-coupled device (ICCD) camera to produce an image of the fluorescence in that region. Besides the species concentration, temperature, pressure and velocity could be measured with prudent choice of transition and subsequent image processing [3].

A basic arrangement of PLIF is illustrated below in Fig. 1.2. A laser beam passes through a cylindrical lens and transforms into a laser sheet. The laser sheet spreads across the H_2 - O_2 flame produced by a flat flame burner, where it excites the OH radical, causing it to emit fluorescence. The fluorescence signals are acquired by an intensified CCD camera. A UV filter and an imaging lens are placed in front of the ICCD to filter and focus the photo signals. The images captured by the ICCD are transmitted to a computer for processing.

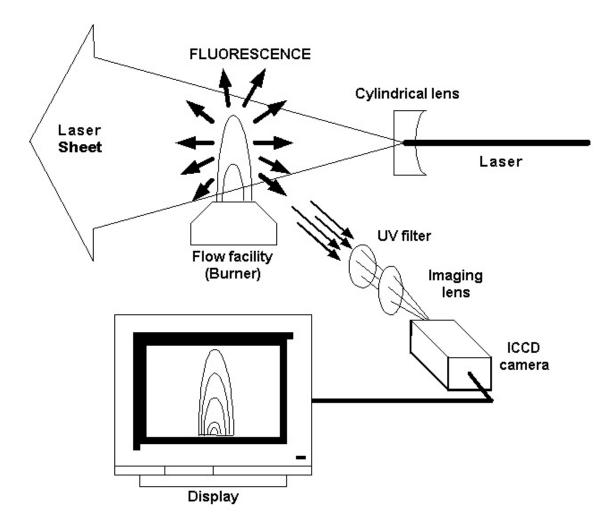


Figure 1.2: Basic Arrangement of PLIF

1.1.3 Fluorescence Dependence of Temperature and Mole Fraction

The measured quantity of the fluorescence signal, in photons per pixel, from a single ICCD camera frame can be written as [2]

$$S_{pp} = const \cdot \chi_{OH} \cdot [f_B(T)/T]$$

The fluorescence signal is directly proportional to the mole fraction of OH and a temperature-dependent function in brackets. Experimental efficiencies, such as

transmission efficiency of the collection lens and filter, the camera's photo-cathode quantum efficiency, and the electronic gain on the camera, are considered constant for a given experimental setup [2].

1.1.4 Disadvantages of PLIF

Just as nothing is perfect, PLIF is not without blemish either. The main disadvantage of PLIF is quenching of fluorescence at higher pressures by numerous collisions of molecules. Such quenching effects can be avoided easily by exciting the molecule to a fast predissociating state from which fluorescence is emitted only during the predissociation lifetime. For sufficiently short predissociation lifetimes, collisions do not occur, which means no quenching within the lifetimes. In some environments where PLIF is unsuitable, other laser techniques, such as coherent anti-Stokes Raman scattering (CARS) or Rayleigh scattering are recommended [7].

1.2 Objective

In long duration facilities, flow parameters could be measured with physical probes that are placed spatially through the flow. However, this method is not suitable for pulsed flow facilities, because of their short flow duration. A desirable alternative to such probe-based methods in pulsed facilities is PLIF imaging, which yields instantaneous measurements at a large number of points in a laser-illuminated plane [8].

The development of pulse detonation engines is underway at the Aerodynamics Research Center. The initiation of detonation is believed to occur at "hot spots" [9] where heat and pressure are sufficient to trigger a rapid chemical reaction. In oxyhydrogen systems that involve chemical activity, a large population of OH is expected. This OH formation can be studied by the use of PLIF, since OH is one of the species that can be detected in PLIF measurements.

The objective of this research is to visualize OH formation within oxy-hydrogen combustion by means of PLIF, as a stepping-stone to the next phase of experiments in detonation waves.

CHAPTER II

EXPERIMENTAL SET-UP

The PLIF set-up without combustion system is shown in Fig. 2.1 below. The combustion system is depicted in section 2.2.

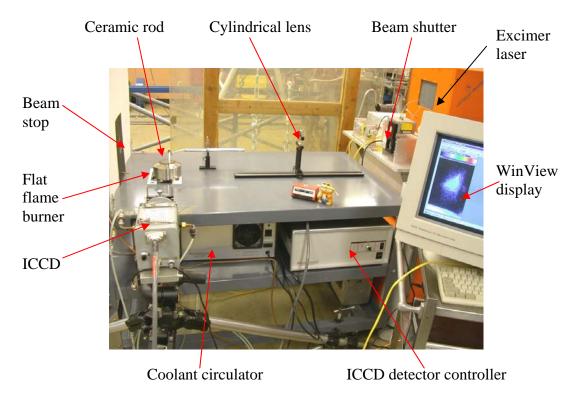


Figure 2.1: PLIF set-up

2.1 KrF Excimer Laser

Excimer stands for excited dimer, a diatomic molecule usually of a rare gas atom and a halide atom that are bound in excited states only. ArF, KrF, XeCl and XeF are the most important excimers, because they dissociate releasing the excitation energy through UV photons [10]. The energy is high-intensity beams of UV light with wavelengths ranging from 193 to 351 nm [11]. When compared with visible light sources, UV lasers are capable of detecting minor species, such as, OH, NO, and CO [12]. The OH radical is naturally occurring [2] and one of the most abundant in flames [6].

In this experiment, a COMPex 150T KrF Excimer Laser (Lambda Physik) was used, because its center wavelength and bandwidth are tunable between 248.0 and 248.8 nm and it coincides with some of the OH A-X (3,0) band system [12]. The absorption line of OH is excited at 248 nm $[A^2\Sigma^+(v'=3) \leftarrow X^2\Pi (v''=0)]$ and detected at 297 nm $[A^2\Sigma^+(v'=3) \rightarrow X^2\Pi (v''=0)]$. Despite the weaker strength of this OH transition than others that could be used, such as 284 and 308 nm, there are advantages of using 248 nm. Excitation of the OH radical at 248 nm not only minimizes the dependence of the resulting fluorescence on quenching effects, but also maximizes the predissociationdominated decay process. "Predissociation occurs when the potential energy curve for a bound molecular state is crossed by the curve of a repulsive, or unbound, state" [2]. Selecting this particular LIF approach enables the specifically-sensitive measurement of OH concentration in the flow. This approach has previously been used in OH measurement of an automobile engine, in a laboratory-scale, supersonic, H₂-air combustion tunnel, etc. [2].

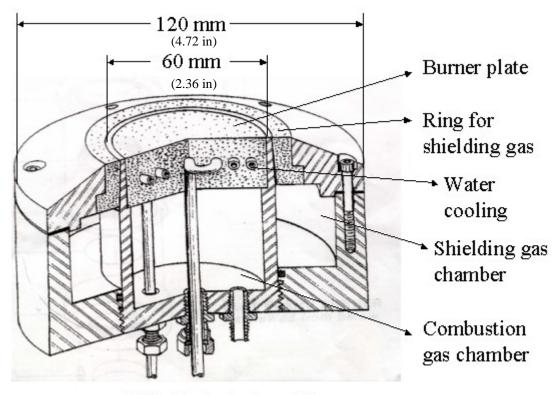
The laser pulse energy is maximum at 380 mJ/pulse at low repetition rates (< 10 Hz, or pulses/sec.). However, the laser was operated at 20 Hz with a constant pulse

energy of 50 mJ/pulse set by a designated computer for the laser. Pulse length is typically 22 nm, which is fast enough to freeze flows [2].

2.2 Combustion System

2.2.1 Flat-Flame Burner

A "laboratory-standard" [2], McKenna flat flame burner is used, as shown below in Fig. 2.2.



© Holthuis & Associates

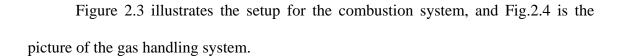
Figure 2.2: Sectional View of McKenna Flat Flame Burner

The burner is made up of a 6-cm-diameter sintered porous stainless-steel plug and a 1.4-cm-width sintered porous bronze shroud ring. The reactant mixture consisting of pre-mixed oxidizer (O_2) and fuel (H_2) is introduced into the bottom of the burner and distributed over the cross section by the sintered plug. Any existing surge in the reactant stream is settled in the cavity located below the sintered plug. Passing a concentric inert gas (N_2) around the flame holder through the bronze plug ring enables shielding of the flames produced from entrainment effects and stabilizes the flame above the sintered plug. This shielding gas is also brought into a cavity in the bottom of the burner in order for any surge present to be damped out. The cooling water is supplied to the burner by a stainless-steel tubing, where it is rolled in a spiral shape with equally spaced rings and embedded in the sintered plug. The plane of the coil is parallel to the surface of the burner to ensure that radial temperature gradients do not occur [13].

2.2.2 Combustion Gas Handling System

List of Equipment

- a. Compressed hydrogen (~200 ft³ cylinder, Airgas)
- b. Compressed oxygen (~80 ft³ cylinder, Airgas)
- c. Compressed nitrogen (~200 ft³ cylinder, Airgas)
- d. Pressure Regulators
- e. Flashback arrestors (Ibeda Inc.)
- f. Correlated flowmeter with standard valve (Cole-Parmer)
- g. Copper, steel and plastic tubing
- h. Standard temperature millboard (McMaster-Carr)



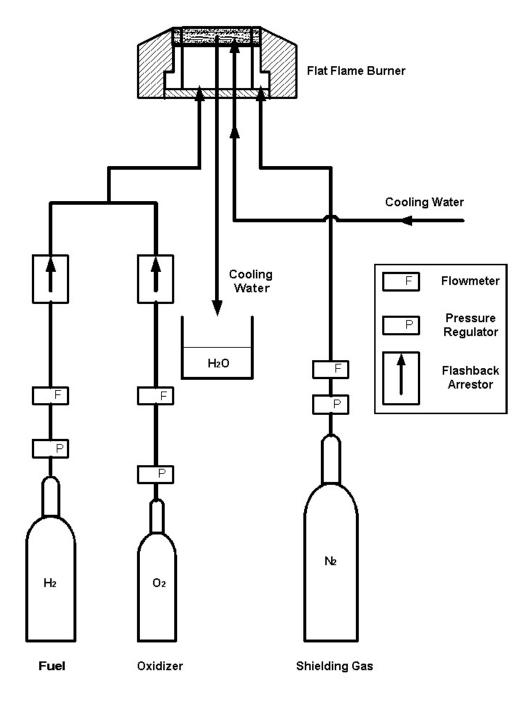


Figure 2.3: Schematic Diagram of Experimental Setup for the Flat Flame Burner

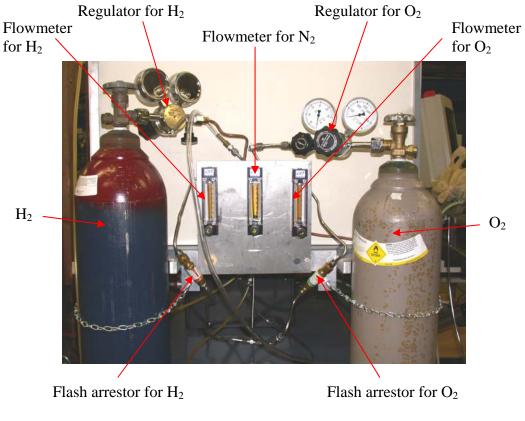


Figure 2.4: Gas Handling System

Oxygen and hydrogen gas cylinders with pressure regulators are fixed on the side of the combustion table. The millboard (40 inch wide by 42 inch high) is erected between the burner and the gas cylinders for safety. The gases from two cylinders are joined at a tube union, and hence pre-mixed before ignited at the burner. To prevent backflash upon ignition, two flash arrestors are installed: one between the O_2 cylinder and the union, the other between the H_2 and the union. In addition, to adjust the flow rates of gases, correlated flowmeters with standard valves (Cole-Parmer catalog nos. A-

03216-22 for O_2 and H_2 , and A-03216-32 for N_2) are installed. A vertical tangential locator line of a flowmeter ensures accuracy in reading the float position.

2.3 Optical System

The optical system consists of spatial filtering, laser sheet generation and image collection.

List of Equipment

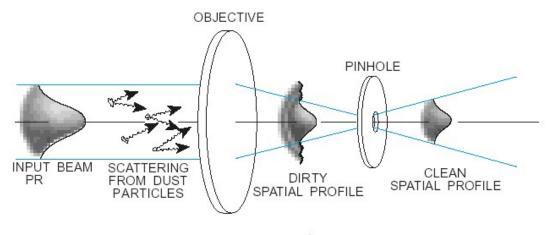
- a. Spherical plano-convex objective lens: $f \sim 350 \text{ mm}$, d = 2 in (CVI Laser)
- b. Pinhole: 40 µm diameter
- c. Cylindrical plano-concave lens: $f \sim -81.4$ mm, d = 1 in (CVI Laser)
- d. Spherical plano-convex imaging lens: $f \sim 50 \text{ mm}, d = 1 \text{ in (CVI Laser)}$
- e. Band-pass filter: center wave length 296.7 ± 10 nm (HBW or FWHM),

d = 1 in (CVI Laser)

f. Optical table rails & carriers (Thermo Oriel)

2.3.1 Spatial Filtering

Spatial filters eliminate random fluctuations from the intensity profile of a laser beam, which increases resolution significantly. Optical defects and particles in the air cause intensity variations of laser beams. Spatial filtering is fairly simple (see the schematic shown in Fig. 2.5). An ideal coherent, collimated laser beam behaves as though produced by a distant point source. Spatial filtering consists of focusing the beam and producing an image of the "source" with all the imperfections in the optical path defocused in an annulus about the axis. A pinhole blocks most of the noise [14].



© Newport Corporation

Figure 2.5: Spatial Filters

In this research, a 2-inch-diameter fused silica singlet lens, PLCX-50.8-180.3-UV from CVI Laser Corporation, is the objective lens, and a 40-µm-diameter pinhole is employed.

2.3.2 Laser Sheet Generation

PLIF requires the generation of a laser sheet. A cylindrical lens is used to do this. In the experiment, a 1-inch-diameter VUV CaF₂ round singlet lens, CLCC-25.4-38.1-CFUV from CVI Laser, is used. Calcium fluoride negative cylindrical lenses are appropriate for anamorphic expansion at vacuum ultra-violet wavelengths. They are ideal for use with excimer lasers [15].

The excimer laser first passes through the objective lens, cleaning the scattering from dust particles [14]. It then goes through the pinhole, creating a "clean" [14] spatial profile. Subsequently, it passes through the cylindrical lens, producing the laser sheet.

2.3.3 Image Collection

A narrow band interference filter with center wavelength of 296.7 ± 10 nm (F10-296.7-3-1.00 from CVI Laser) is selected (see Section 2.1). An ICCD camera with the filter mounted onto, is placed at right angles to the laser sheet. This will let only the selected fluorescence pass and block undesirable ambient light, scattered laser light at 248nm and emission from the flame [2].

A 1-inch-diameter fluorescence-imaging lens (fused silica single lens, PLCX-25.4-25.8-UV from CVI Laser) is placed about 50 mm away from the ICCD intensifier, with the convex side facing the ICCD, and focuses the image into the ICCD. The optical table rail and carrier were used to fix the lens onto the table.

2.4 ICCD (Intensified Charged Coupled Device)

List of Equipment

- a. ICCD detector (Princeton Instruments)
- b. ICCD detector controller (Princeton Instruments)
- c. Coolant circulator (Princeton Instruments)
- d. Ultra high purity, compressed nitrogen (Airgas)

The ICCD detector (Model ICCD 576 E) from Princeton Instruments (PI) has a 384 x 576 pixel format. The front enclosure of the ICCD detector includes the image intensifier and the CCD. The image intensifier is coupled to the CCD using a fiber optic window on both the output of the image intensifier and the front side of the CCD. Due to the1:1 coupling ratio, the fiber optics translates the output of the image intensifier to the input of the CCD at the same size. This type of coupling is the most efficient possible, as well as eliminating lens effects such as vignetting.

"The CCD array in an ICCD camera is cooled by a Peltier-effect thermoelectric cooler, driven by closed-loop proportional-control circuitry. A thermal sensor attached to the cooling block of the detector monitors its temperature. The coolant block is made of Delrin to prevent condensation, but this also prevents heat transfer from the thermoelectric cooler to the atmosphere. Water or other coolant is therefore necessary to carry away heat generated by the thermoelectric cooler [16]." The ICCD is typically operated cooled (\geq -45°C) with chilled water circulation, with the help of a coolant circulator, Model CC-100, also provided by PI.

The ICCD needs continuous flushing of the detector with nitrogen while operating cooled. Even if the detector is not powered, flushing is required whenever the coolant is circulating, in order to prevent moisture condensation. Since PI recommends nitrogen with less than or equal to 10 ppm, compressed nitrogen of grade 'ultra high purity' from Airgas is used.

The ICCD is controlled by the detector controller, Model ST 138-S, from PI which sends information to a computer through a TAXI cable and PCI interface card.

The ICCD was tested in visible light with a Nikon lens with a focal length of 1 meter (3.28 ft), with all the room lights shut off. The UV filter was removed and the laser was not firing. Figure 2.6 shows a 3-D object (Garfield, also shown in Fig. 2.1) acquired.



Figure 2.6: ICCD Test with 3-D Garfield

2.5 Image Acquisition and Processing

The ICCD controller is managed by WinView/32 software from PI. The fluorescence images acquired are stored on the host computer (Windows98), processed and displayed.

2.5.1 Image Acquisition

The WinView/32 software, also from PI, is installed on the host computer (Windows98). The PCI interface card installed in the host computer permits communication between the computer and the ST-138 controller. The controller accepts

input from the computer, or WinView/32, and converts it to the proper control signals for the camera. These signals can be adjusted to specify the readout rate, binning^{*} parameter, regions of interest, gain, and array temperature. Data acquired in the camera is transmitted to the controller where it is processed. It then is passed on to the computer, displayed in the WinView/32 window, and can be stored on disk [17]. The system configuration is presented in Fig. 2.7 below.

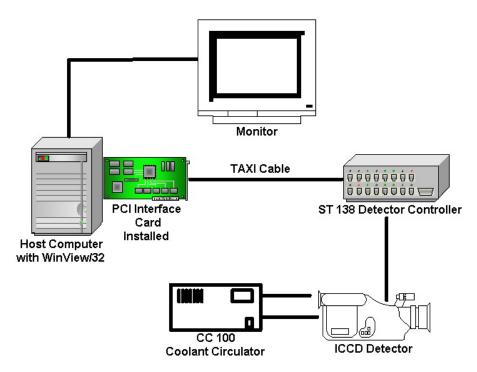


Figure 2.7: Image Acquisition System Configuration

2.5.2 Image Processing

PI systems are adjusted so that data have a small offset. This offset assures that small signals will not be missed, and can be subtracted after the signal is acquired to

^{*} The specified number of pixels in each direction will be combined to form super pixels for greater sensitivity [17].

prevent it from having any influence on the data. The background subtraction feature automatically subtracts any constant background in the signal. This includes not only constant offsets caused by the amplifier system in the controller, but also timedependent buildup of dark charge. The background subtract equation is expressed as follows [17]:

Corrected image data = Raw image data – Background

Another data correction technique, flatfield correction, which divides out nonuniformities in gain from pixel to pixel, was also available in WinView/32, but was not used, because this was found to be unnecessary.

The corrected images are displayed in WinView/32. Gray-scale images can be displayed in false color, which varies with the light intensity or the fluorescence signal. Five frames of images are averaged, since the flame fluctuate even with nitrogen shielding, mostly because of air-conditioning.

CHAPTER III

RESULTS AND DISCUSSIONS

This chapter presents the PLIF results obtained and the interpretation of the results. The experimental procedures are detailed in Appendices A - C.

3.1 Results

The laser repetition rate (20 Hz), laser energy (50 mJ/pulse), camera position and focus and camera filters are kept constant during the experiments. The PLIF technique was employed to capture images of fluorescence emitted by the 248 nm KrF excimer laser excitation, while varying mixture ratio, or controlling the *volumetric* flow rates of hydrogen and oxygen. Mixture ratio (MR) is defined as the ratio of the oxidizer to fuel *mass* flow rates [2]. A ceramic rod was also inserted in the flames, perpendicular to the laser sheet, to stabilize the flame [18].

In the case of the fuel being H_2 and the oxidizer O_2 , the reaction is expressed as:

$$2 H_2 + O_2 \rightarrow 2 H_2O$$

So, the stoichiometric MR for H₂-O₂ combustion is calculated as 8.

$$\frac{m_{O_2}}{\dot{m}_{H_2}} = \frac{1 \,\text{mol} * 32 \,\text{g/mol}}{2 \,\text{mol} * 2 \,\text{g/mol}} = 8$$

A number of PLIF images were taken at various mixture ratios, listed in Table 3.1. The values for volumetric flowrates [ml/min] in Table 3.1 are taken or interpolated

from the correlation charts provided by Cole-Parmer (Table 3.2), and the mixture ratios (MR) are calculated as follows, taking case one for example.

$$\frac{\dot{m}_{O_2}}{\dot{m}_{H_2}} = \frac{\frac{459 \,\text{ml}/\text{min}}{22400 \,\text{ml}/\text{mol}} * 32 \,\text{g}/\text{mol}}{\frac{2025 \,\text{ml}/\text{min}}{22400 \,\text{ml}/\text{mol}} * 2 \,\text{g}/\text{mol}} = 3.627 \,,$$

where 22400 ml is the volume occupied by 1 mole of a gas, assuming an ideal gas, and molecular weight of O_2 is 32 g/mol and H_2 2 g/mol.

Case	H ₂ [mm]	H ₂ [ml/min]	O ₂ [mm]	O ₂ [ml/min]	N ₂ [mm]	N ₂ [ml/min]	Ceramic	MR
1	20	2025	15	459	60	12724	No	3.627
2	20	2025	15	459	60	12724	Yes	3.627
3	25	3000	15	459	60	12724	No	2.448
4	25	3000	15	459	60	12724	Yes	2.448
5	30	4162	15	459	60	12724	No	1.765
6	30	4162	15	459	60	12724	Yes	1.765
7	18	1639	15	459	40	9608	No	4.481
8	18	1639	15	459	40	9608	Yes	4.481
9	20	2025	17	575	40	9608	No	4.543
10	20	2025	17	575	40	9608	Yes	4.543
11	20	2025	20	749	40	9608	No	5.918
12	20	2025	20	749	40	9608	Yes	5.918

Table3.1: Results

The maximum flow of O_2 was 20 mm or 749 ml/min for this setup. When the O_2 flow rate was increased over that value, the burner started making a whistling noise and the flame became unstable, which eventually might have been put out. On the other hand, the H₂ flow rate could not be lowered less than 18 mm or 1639 ml/min, while it could go up to the flowmeter limit, 65 mm or 13.6 l/min. The shielding gas, N₂, was

kept constant at 60 mm or 13.4 l/min for the first six cases, where the flame was relatively larger, and lowered to 40 mm or 96.1 l/min for relatively small flame.

Scale Reading	H ₂ [ml/min]	O ₂ [ml/min]	N ₂ [ml/min]
[mm]			
65	13600	3761	13412
60	11875	3395	12724
55	10360	3051	12049
50	9302	2727	11252
45	7600	2400	10450
40	6450	2098	9608
35	5176	1787	8753
30	4162	1451	7777
25	3000	1108	6748
20	2025	749	5626
15	1060	459	4502
10	430	251	3313
5	178	120	2056

Table 3.2: Correlation Chart for Flowmeters

An example of WinView/32 display is shown in Fig. 3.1 below. The image is rotated 90 degrees counter-clockwise. Colors are assigned according to intensity of fluorescence captured by the ICCD, low intensity being black and high being red. The range of the intensity level was kept at 300 to 1200, after examining all the images. Vertical and horizontal cross-sections profile the intensity at the strips where the cursor (shown as a big white cross) is placed. High intensities shown at the bottom of the image can be assumed to be the fluorescence emitted at the beam stop (see Fig. 2.1) that reflected back and is removed during further processing.

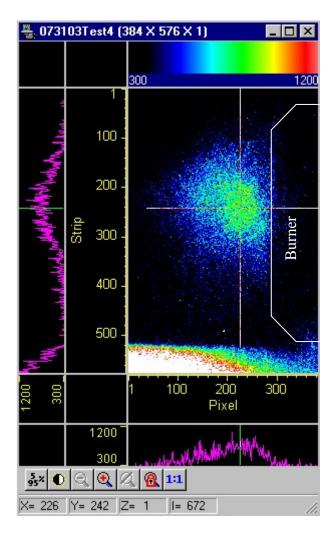
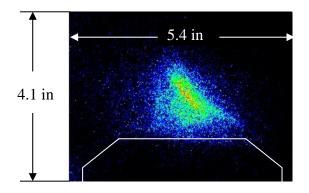
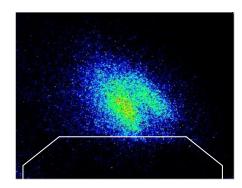


Fig. 3.1 An Example of WinView/32 Display

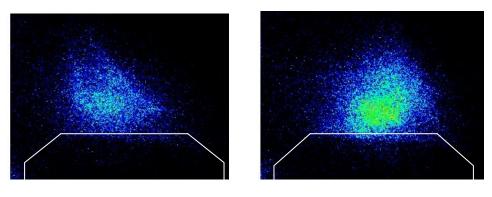
Figure 3.2 demonstrates averaging of five images for Case 1. The reason why the averaging was done is because the flame flickers somewhat. Figures 3.3 and 3.4 show the averaged images for the twelve cases. At each MR, a ceramic rod was inserted to stabilize the flame [18] (Fig. 3.4). The position of the rod is 0.44 in. above the burner surface and 0.31 in. to the right from the vertical centerline of the burner, as described in Fig. 3.5. The burner is outlined with white lines in each image. For Cases 7 and later, the ICCD was accidentally moved and the image frame also moved slightly.



(a) Test 1

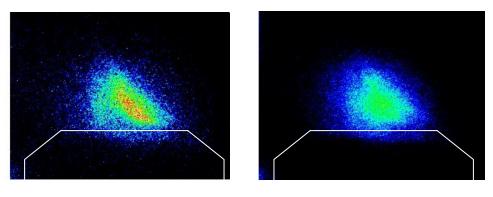






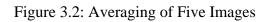
(c) Test 3

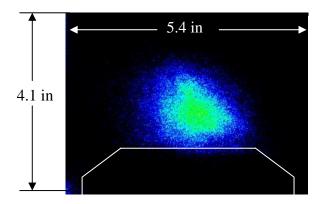




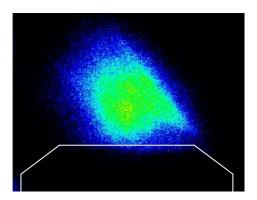


(f) Average

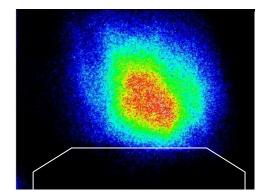




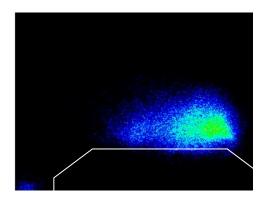
(a) Case 1: MR = 3.627



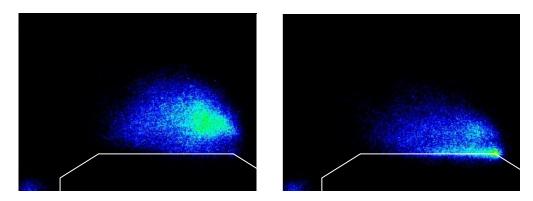
(b) Case 3: MR = 2.448



(c) Case 5: MR = 1.765



(d) Case 7: MR = 4.481



(e) Case 9: MR = 4.543

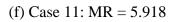
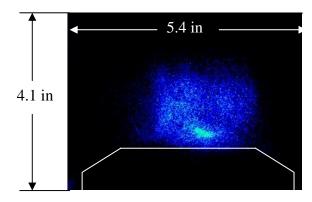
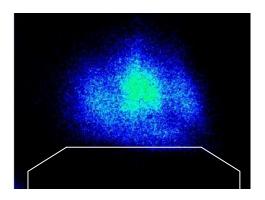


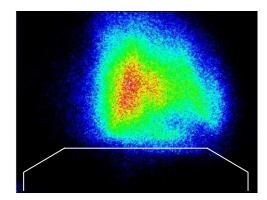
Figure 3.3: PLIF Images of Six Cases without Ceramic Rod



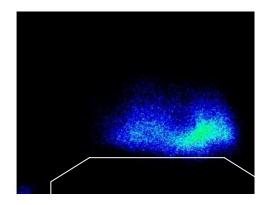
(a) Case 2: MR = 3.627



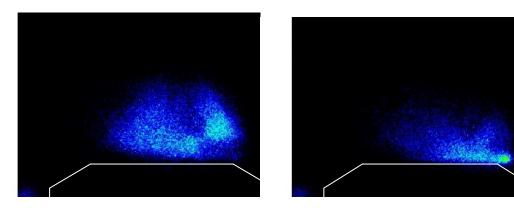
(b) Case 4: MR = 2.448



(c) Case 6: MR = 1.765



(d) Case 8: MR = 4.481



(e) Case 10: MR = 4.543

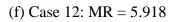


Figure 3.4: PLIF Images of Six Cases with Ceramic Rod

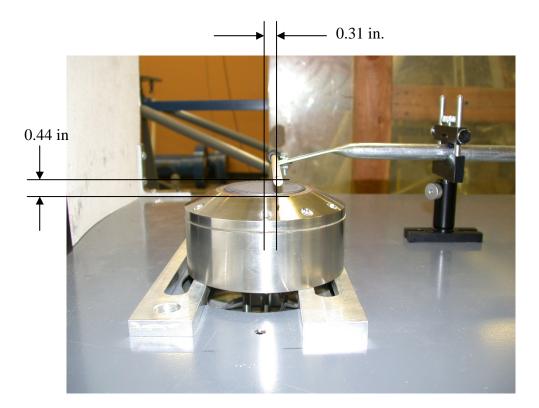


Figure 3.5: Position of Ceramic Rod

3.2 Discussion

The flame was asymmetric for the cases without the ceramic rod, but this was because the flame was non-uniform: the flame was always generated on the right side of the burner, seen from the ICCD. However, after inserting the ceramic rod, the flame became more uniform. The flame in Fig. 3.4 is more uniform than that in Fig. 3.3.

The stoichiometric mixture ratio of 8 was not achieved, due to restriction of the gas handling system, or the burner.

In both Figs. 3.3 and 3.4, it can be seen that lower the mixture ratio (MR), the larger is the flame image. As mentioned in section 1.1.3, the measured quantity of the

fluorescence signal is written as, $S_{pp} = \text{const} \cdot \chi_{OH} \cdot [f_B(T)/T]$. The fluorescence signal is directly proportional to the mole fraction of OH and a temperature-dependent function in bracket.

As the MR increases, both adiabatic flame temperature (T) and OH mole fraction (χ_{OH}) increase[¥]. In this situation, it is difficult to separate the χ_{OH} term from the [$f_B(T)/T$] term, since when χ_{OH} increases, 1/T decreases while $f_B(T)$ varies with temperature, too.

In the experiments, it appears that the temperature effect was dominant. In other words, as MR increased, χ_{OH} increased, but the [f_B(T)/T] term decreased much more, causing the fluorescent to decrease. It is problematic to say that the OH concentration is directly proportional to the fluorescence signal.

The spatial filters discussed in Section 2.3.1 were not used, because the strong laser burned the pinhole. After removing the pinhole and the objective lens, decent laser sheet was still generated and the experiment was continued.

[¥] From personal communication with Dr. D. R. Wilson, who used hypersonic air-breathing propulsion code.

CHAPTER IV

CONCLUSIONS AND RECOMMENDATIONS

4.1 Conclusions

The objective of the study was to visualize the hydroxyl (OH) formation within the hydrogen-oxygen combustion by means of PLIF, which is suitable for use in pulsed facilities. The conclusions drawn from the investigation are summarized as follows:

- 1. Stoichiometric mixture ratio of 8 was not attained.
- 2. As the MR increased, the fluorescence signal decreased.
- 3. The planar laser-induced fluorescence of the OH radical was acquired successfully, however, if it can be directly reflected in OH concentration is in question.

4.2 Recommendations for Future Work

The burner needs repairing in order to get a uniform flat flame, as well as to reach stoichiometric mixture ratio.

In addition, a UG-11 Schott glass filter may be useful to block the 445-nm second transmission band of the 297-nm bandpass filter, as employed in experiments in [2], where they have the same experimental setup: the same KrF excimer laser and the same 297-nm filter.

Although the images were taken by single shot Free Run mode, the whole image acquisition setup can be integrated by introducing a PG-200 pulse generator. The pulser can synchronize the laser, ICCD detector and ST-138 detector controller, when used as a master trigger. This will eliminate the possibility of missing the fluorescence signal by the ICCD.

Further study should be performed for the future applications of PLIF. With more sophisticated equipment, temperature and velocity measurements can be achieved.

APPENDIX A

EXPERIMENTAL PROCEDURE

- 1. Ensure the following connections:
 - a. Connect a 25-pin ICCD cable from the ICCD to "Detector" of the ST-138 controller.
 - b. Connect a TAXI cable from "J7" of the ST-138 controller to the PCI interface card of the computer with WinView/32 software installed.
 - c. Ensure the coolant hoses and N_2 hose are connected to the ICCD.
- 2. Place the cylindrical lens about 23.5 inches from the burner, with the concave side facing the laser.
- 3. Place the ICCD camera as close to the burner as possible (about 20 inches) at the right angle to the laser sheet. Also place the imaging lens about 50 mm from the image intensifier of the ICCD, with the convex side facing the ICCD. Since the bandpass filter is mounted on the ICCD, the imaging lens is almost touching the ICCD.
- 4. Power the laser assembly, designated computer and monitor for the laser. Open the water valve. See Appendix B for the details of the laser operation.
- 5. Ensure all the valves of the four flowmeters are closed.
- Open the nitrogen cylinder valve. Adjust the pressure regulator to about 15 psi (1 bar). Then open the gas line.
- 7. Power the ST-138 detector controller, switch on the cooling of the ST-138, and power coolant circulator CC-100. The ST-138 will be ready (temperature is thermostated to within ±0.050 °C of preset value; in this case, -30 °C) in about 10 minutes when the status LED is turned from orange to green.

- 8. Immediately after procedure 8, open the valve of the flowmeter of the nitrogen for the ICCD and set it to 1~2 liters/min for at least 10 minutes, and then lower to 0.75~1 liters/min. The flowmeter is for air (molecular weight of about 28.8 g/mol), but it can be used in the same way since it is close to that of nitrogen (28 g/mol) and accurate flow rate is not required. One division equals 1 liters/min.
- Power host computer for WinView/32 and monitor. Start the WinView/32 software. If you start WinView/32 before you turn on ST-138, you will see "Controller is not responding!" See Appendix C for WinView/32 operation procedure.
- 10. Open the valves of O₂ and H₂ cylinders. Set the pressure regulators to about 15 psi (100 kPa). Open the gas lines.
- 11. Open the flowmeter valve for H₂ and ignite the burner with a long match. The flowmeter reading should be about 20 mm. Turn off the room lights in order to see the flame better. Open the flowmeter valve for O₂, too, and adjust it to about 15 mm. *Always ignite fuel (H₂) first!*
- 12. Open the flowmeter valve for N_2 and adjust it to the maximum value, 60 mm, or until you do not see orange flame any more. Orange flame means it is reacting with ambient air and the N_2 shielding is not sufficient.
- 13. Fire the laser. See Appendix B for the laser operation procedure.
- 14. "Acquire" the image with WinView/32. See Appendix C for the WinView/32 procedure.

- 15. Change mixture ratio of the H₂ and O₂ and acquire more images and repeat. Also, insert ceramic rod to stabilize the flame [18].
- 16. After the experiments, turn off everything. *Do not forget to close the water valve, as well.*
- 17. After turning off the ST-138 controller, keep flushing the ICCD with nitrogen at a flowrate of at least 2 liters/min for at least 30 minutes. This keeps moisture condensation from forming on the ICCD until it reaches room temperature.

APPENDIX B

COMPEX 150 EXCIMER LASER OPERATION PROCEDURE

I. Starting the Laser Assembly

Keep in mind that the laser assembly consists of Amplifier and Oscillator.

- 1. Check the beam path.
- 2. Turn on main switch of the laser assembly at the utility supply.
- 3. Turn on main switches and key switches of both Amplifier and Oscillator.
- 4. Switch on the designated computer (PC) and the monitor. *Always turn on the laser assembly before turning on the computer to prevent data ring diagnosis failure* (see section II. 1. below).
- 5. Wait until DOS-prompt C:\> and a blinking cursor appears.
- 6. Type "cd lambda" to change directory.
- 7. Type "lc150" to start the COMPex150 software.

II. Booting the Local Software

After the software started, it passes three tests:

1. Data ring diagnosis (also called dynamic fiber loop check)

The data ring of the COMPex 150 is a local area network (LAN), which links the modules of the laser assembly with fiber optics lightwave guides. To test it, the PC sends a light signal and checks if it passes the LAN. If this test fails, a window "DYNAMIC FIBER LOOP CHECK" appears.

When this happens, turn off both the PC and the laser assembly and wait for about 30 minutes and turn them back on.

If everything is all right, the program enters the module availability check.

2. Module Availability Check

This tests what modules of the laser assembly are recognized by the PC. If any of them does not respond, NOT INSTALLED is displayed and a beep tone is displayed. *However, because the last one, F2-SOURCE, is not installed, just hit G to continue if everything else is YES.* The F_2 -source is an extra part that could generate the fluorine (F_2) gas when the laser is turned on, but in our configuration, the F_2 gas in the gas cabinet is used.

Upon pressing G, the program continues with the self test.

3. Self Test

The self test shows the results of the data ring diagnosis and checks the COMPex watchdogs. Afterwards, it sets the stepper motor of the grating to its last position.

Motor reference position once failed. The grating is controlled via a small stepper motor. When the laser is powered on this motor will drive in one direction until it sees an optical "flag" and then stop. This is the motor's way of finding "home". If it doesn't find "home" then the software will give this error. This problem was solved by turning the drive motor pulley with fingers in such a direction to drive the grating in the opposite direction about half an inch with the main power off. Then turn on the main power and watch the motor move the grating.

If the test results are OK, the program begins to wam-up the thyratron and enters the Main Menu. It takes about 8 minutes to warm up.

III. Main Menu – The Central Program Menu

- 1. Check current settings on Main Menu.
- 2. Press M to open the Data Menu.

Notice that both laser devices (Ampl. for Amplifier and Osc. for Oscillator) are

represented with their own page. The Oscillator page is the master page;

parameters in Amplifier are ignored.

Check Parameter Setting of the Oscillator as follows:

- a. Excimer gas and its wavelength (KrF, 248 nm)
- b. Buffer gas (Ne for KrF)
- c. Repetition rate (20 Hz)
- d. Trigger mode (internal)
- e. Running mode and its preset parameter (Energy Constant mode at 50mJ)

It is possible to change the gas values, but it is not recommended since the preset gas values are carefully optimized.

3. Laser Firing

a. Open the water valve over the stairs.

The laser has to be cooled with water at repetition rates higher than 10 Hz.

- b. Open the beam shutter.
- c. Type S, and then Y to confirm.

The laser now needs a few seconds to initialize the power supply. The menu changes to the Running Menu. Then is begins emitting laser light with the given repetition rate. It can be recognized by clicking noise. The higher the repetition rate is, the faster the clicking will be.

All the gas valves remain closed while firing the laser.

d. If the laser beam seems weak, New Gas Fill has to be performed.

IV. Conditions Before Starting New Fills

- 1. The laser must not be firing.
- 2. Make sure that all the four required gases are available.

Do not use gas cylinders with less than 74 psi (5 bar), since the impurities from gas cylinder walls can contaminate the tube.

- 3. Turn on the vacuum pump of the gas cabinet.
- 4. Open all the four gas cylinder valves in the gas cabinet.
- 5. Close all the four pressure regulators in the gas cabinet.
- 6. Ensure all the four pressure regulators are set to about 60 psi (4 bar).
- 7. Open all the gas lines, marked with post-it notes.

V. New Gas Fill (automatic)

- 1. Go to the Main Menu and press M to enter the Data Menu.
- 2. Press N and select the page with KrF Osc. or KrF Ampl. that needs a new fill.

-----VERY IMPORTANT-----

The New Fill procedure fills the laser device that is actually the upper most page on the Data Menu. You have to fill Oscillator and Amplifier, separately. To change the page on the Data Menu, hit N for next page.

-----VERY IMPORTANT-----

- 3. Push Q to leave the Data Menu. Now you are on the Main Menu.
- 4. Push G to open the Gashandling Menu.
- 5. Push N to start a new fill.

The laser tube is evacuated to 30 mbar. If the evacuation exceeds 12 minutes, there may be a gas leak. The gases will be filled one by one for the amount specified. The new fill will stop automatically when it is filled to 3330 mbar.

APPENDIX C

WINVIEW/32 OPERATION PROCEDURE

1. Start WinView/32. Unless the ST-138 is powered, "Controller Is Not Responding!" will appear. If the ST-138 is on and the message still shows up, the pins of the TAXI cable is most likely torn. Check the pins of the cable with continuity test of a multimeter. The TAXI cable has only four wires in it:

pin # 2 connects to pin #4 on the opposite side
pin # 7 connects to pin #9 on the opposite side
pin # 4 connects to pin #2 on the opposite side
pin # 9 connects to pin #7 on the opposite side

Once you find discontinuities, dismantle the both ends of the cable and solder the pins.

- 2. In Setup | Hardware | Interface tab, select "High Speed PCI".
 - In Setup | Hardware Controller/Camera tab, select "ST 138" for Controller Type, "EEV 576x 384 [3ph]" for Camera Type, "Electronic" for Shutter Type and check "Full Frame" for Readout Mode.
- 3. Go to Acquisition | Experimental Setup.
 - a. In Main tab, select "0.1 sec" for Exposure Time, "1" for Number of Images, and"1" for Accumulations. Also check "Use Full Chip" for CCD Readout.
 - In Timing tab, select "Free Run" for Timinig Mode and "Normal" for Shutter Control.
 - c. In Data File tab, type in Data File Name a file name you want to create, such as
 073103Test. Click on "Enable" for Auto Increment File Name and set Current

Value to "1". This will facilitate the file handling by adding the suffix with incrementing numbers from 1, as 073103Test1, 073103Test2 and so on.

- d. In ROI tab, select "1" for New Pattern, "1", "380" and "5" for X, and "1", "575", "5" for Y.
- 4. The proper position of the ICCD can be checked with Flat Field images, although flat field is not used for the right purpose. With the room lights on, take flat field by "Acquisition | Acquire Flat Field." *You will see high intensities caused by fluorescent lights on the sides of the burner on the image*. Using these clues, move the ICCD to the right position.
- 5. Acquire background for background correction, by selecting Acquisition | Acquire Background. Then in Acquisition | Experimental Setup | Data Correction tab, select the background file. *Make sure the Flat Field box is unchecked*. Flat Field feature is not used, because it does not improve the images in this particular experiment.
- 6. Turn off the room lights, ignite the burner, adjust the flowrates to desired mixture ratio and fire the laser. Acquire images by selecting Acquisition | Acquire. A black-and-white picture will be shown on display.
- Change flowrates of the H₂ and O₂ gases and repeat 5. Also, insert a ceramic rod to stabilize the flame [18] and repeat 5.
- 8. Gray-scale images can be converted to color images by color icon.
- 9. In Display | Display Layout | General tab, Axes, Cross Sections and Info Bar boxes are checked to make analysis easier.
- 10. In Display | Display Layout | Range tab, Intensity level can be manually selected.

11. Since WinView/32 can be run only on Windows 98 machine (we have license only on Windows 98), press "Print Screen" on the keyboard, paste onto wordpad and save as Word 6 file in order to make color copy. This can be seen and edited in Microsoft Photo Editor on other machines

REFERENCES

- Eckbreth, A. C, "Laser Diagnostics for Combustion Temperature and Species", Abacus Press, Kent UK and Massachusetts USA, 1988.
- Cohen, L.M., Jassowski, D.M., and Ito, J.I., "Performance of a Titan rocket engine using laser-induced fluorescence of OH", *AIAA Journal*, Vol. 39, No. 10, pp. 1926-1935, October 2001.
- Houwing, A.F.P, Smith, D.R., Fox, J.S., Danehy, P.M., & Mudford, N.R., "Laminar boundary layer separation at a fin-body junction in a hypersonic flow", *Shock Waves*, Vol. 11, pp. 31 - 42, 2001.
- Bessler, W.G., Hildenbrand, F., & Schulz, C., "Two-line laser-induced fluorescence imaging of vibrational temperatures in a NO-seeded flame", *Applied Optics*, Vol. 40, No. 6, pp. 748 - 755, 2001.
- Rothe, E.W., and Andresen, P., "Application of tunable excimer lasers to combustion diagnostics: a review", *Applied Optics*, Vol. 36, No.18, pp. 3971 – 4033, 1997.
- Bombach, R., "Laser Spectroscopy in Combustion Research", Ch.2, Paul Scherrer Institut, Switzerland, Internet URL: http://cdg.web.psi.ch/Summerschool/chap2.pdf.
- Anderson, P., Bath, A., Gröger, W., et al, "Laser-induced fluorescence with tunable excimer lasers as a possible method for instantaneous temperature field measurements at high pressures: checks with an atmospheric flame", *Applied Optics*, Vol. 27, No. 2, pp. 365 – 378, 1988.
- Houwing, A.F.P., Palma, P.C., et al, "Fluorescence-imaging-based flow diagnostics: applications to free-piston shock tunnel flows", 22nd International Symposium of Shock Waves, Imperial College, London, UK, July 18 – 23, Paper 0890, 1999.

- Oran, E.S., and Khokhlov, A.M., "Deflagrations, Hot Spots, and the Transition to Detonation", 17th International Colloquium on the Dynamics of Explosions and Reactive Systems, Heidelberg, Germany, July 25 - 30, 1999, Internet URL: http://www.iwr.uni-heidelberg.de/~icders99/program/papers/001-050/031.pdf.
- Databank Oxydex, Internet URL http://www.databank.oxydex.com/science_mathematics_references/excimer_lasers. html.
- 11. Lambda Physik Inc., "User Manual COMPex 150", Ft. Lauderdale, Nov., 1997.
- Chen, Y.-C., Mansour, M.S., "Topology of turbulent premixed flame fronts resolved by simultaneous planar imaging of LIPF of OH radical and rayleigh scattering", *Experiments in Fluids*, Vol. 26, pp. 277 - 287, 1999.
- Prucker, S., Meier, W., & Stricker, W., "A flat flame burner as calibration source for combustion research: Temperatures and species concentrations of premixed H₂/air flames", *Review of Scientific Instruments*, Vol. 65, No. 9, pp. 2908 – 2911, Sept., 1994.
- 14. Newport Corporation, "Spatial Filter", Internet URL: http://www.newport.com/file_store/PDFs/tempPDFs/Spatial_Filters_e3873.pdf.
- 15. Lambda Research Optics, Inc., Internet URL: http://www.lambda.cc/PAGE51.htm.
- Roper Scientific, "ICCD Detector Operation Manual", 1999, Internet URL: ftp://ftp.roperscientific.com/manuals/files%5C0009v3f.pdf.
- Roper Scientific, "WinView Princeton Instruments Imaging Software", 2001, Internet URL: ftp://ftp.roperscientific.com/manuals/files%5C0046v25b.pdf.
- Hanson, R.K., Seitzman, J.M., & Paul, P.H., "Planar Laser-Fluorescence Imaging of Combustion Gases", *Applied Physics B*, Vol. 50, pp. 441 - 454, 1990.

BIOGRAPHICAL INFORMATION

Takashi Yokomae received his Master of Science in Aerospace Engineering from the University of Texas at Arlington (UTA) in December 2003. He was born in Japan and upon completing high school there, he came to the United States to study in the University. He received his Bachelor of Science in Aerospace Engineering from UTA in May 2003. Culturally diverse UTA community allowed him to be more broadminded during his 6-year stay at UTA. He likes to play various kinds of sports: for example, basketball, tennis, baseball, softball, swimming and rollerblading.